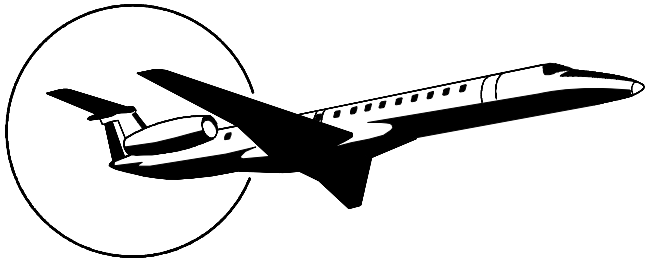
EMB145

EMBRAER



UNITED STATES

QUICK REFERENCE HANDBOOK

THIS PUBLICATION CANCELS AND SUPERSEDES THE QRH–145/1115 DATED APRIL 16, 2001; REVISION 8 DATED APRIL 04, 2005.

QRH–145/1115-06

SEPTEMBER 29, 2006

REVISION 16 – APRIL 26, 2019

In connection with the use of this document, Embraer does not provide any express or implied warranties and expressly disclaims any warranty of merchantability or fitness for a particular purpose. This document contains trade secrets, confidential, proprietary information of Embraer and technical data subject to U.S. Export Administration Regulation ("EAR") and other countries export control laws and regulations. Diversion contrary to the EAR and other laws and regulations is strictly forbidden. The above restrictions may apply to data on all pages of this document.

Copyright © 2019 by Embraer S.A. All rights reserved.

LIST OF EFFECTIVE PAGES

THE PORTION OF THE TEXT AFFECTED BY THE CHANGE IS INDICATED BY A VERTICAL LINE IN THE OUTER MARGIN OF THE PAGE.

ORIGINAL	**0**	**SEP 29, 2006**
REVISION	**1**	**JUL 20, 2007**
REVISION	**2**	**NOV 30, 2007**
REVISION	**3**	**JUN 30, 2008**
REVISION	**4**	**DEC 19, 2008**
REVISION	**5**	**OCT 30, 2009**
REVISION	**6**	**JAN 29, 2010**
REVISION	**7**	**AUG 15, 2011**
REVISION	**8**	**JAN 10, 2012**
REVISION	**9**	**OCT 22, 2012**
REVISION	**10**	**OCT 29, 2012**
REVISION	**11**	**NOV 14, 2013**
REVISION	**12**	**MAY 20, 2014**
REVISION	**13**	**MAR 06, 2015**
REVISION	**14**	**APR 05, 2017**
REVISION	**15**	**MAR 26, 2018**
REVISION	**16**	**APR 26, 2019**

TOTAL NUMBER OF PAGES IN THIS PUBLICATION IS 242 CONSISTING OF THE FOLLOWING:

PAGE N°	CHANGE N°	PAGE N°	CHANGE N°
* Title	16	NAP-2A	14
		NAP-2B	14
* A	16	NAP-3	14
* B	16	NAP-4	14
* C	16	NAP-5	14
		NAP-6	15
I	0	NAP-7	14
		NAP-8	14
NP INTRO-1	14	NAP-8A	14
NP INTRO-2	14	NAP-8B	14
		NAP-9	14
NP-1	14	NAP-10	14
NP-2	14	* NAP-11	16
NP-3	14	NAP-12	14
NP-4	14	NAP-13	14
* NP-5	16	NAP-14	15
NP-6	14	NAP-15	14
		NAP-16	14
* EAP INTRO-1	16	NAP-17	14
* EAP INTRO-2	16	NAP-18	14
* EAP INTRO-3	16	* NAP-19	16
* EAP INTRO-4	16	NAP-20	14
		NAP-21	14
* S-1	16	NAP-22	14
* S-2	16	NAP-23	14
* S-3	16	NAP-24	14
* S-4	16	NAP-25	14
* S-5	16	NAP-26	14
* S-6	16	NAP-27	14
* S-7	16	* NAP-28	16
* S-8	16	* NAP-29	16
* S-9	16	NAP-30	15
* S-10	16	NAP-31	14
* S-11	16	NAP-32	14
* S-12	16	NAP-33	14
* S-13	16	NAP-34	14
* S-14	16	NAP-35	14
		NAP-36	15
NAP-1	15	NAP-37	14
NAP-2	15	NAP-38	14

* Asterisk indicates pages changed, added, or deleted by the current revision.

LIST OF EFFECTIVE PAGES

THE PORTION OF THE TEXT AFFECTED BY THE CHANGE IS INDICATED BY A VERTICAL LINE IN THE OUTER MARGIN OF THE PAGE.

PAGE N°	CHANGE N°	PAGE N°	CHANGE N°
EAP0-1	15	* EAP5-9	16
* EAP0-2	16	* EAP5-10	16
* EAP0-3	16	* EAP5-11	16
EAP0-4	14	* EAP5-12	16
EAP0-5	14	* EAP5-13	16
EAP0-6	14	* EAP5-14	16
* EAP0-7	16		
EAP0-8	14	EAP6-1	14
		EAP6-2	14
EAP1-1	15	EAP6-3	14
EAP1-2	15	EAP6-4	14
EAP1-3	15	EAP6-5	14
EAP1-4	15	EAP6-6	14
EAP1-5	15	EAP6-6A	14
EAP1-6	15	EAP6-6B	14
EAP1-7	15	EAP6-7	14
EAP1-8	15	EAP6-8	14
EAP1-9	15	EAP6-9	14
EAP1-10	15	EAP6-10	14
EAP1-11	15	EAP6-11	14
EAP1-12	15	EAP6-12	14
* EAP1-13	16		
* EAP1-14	16	* EAP7-1	16
		EAP7-2	15
EAP2-1	15	EAP7-3	15
EAP2-2	15	EAP7-4	15
EAP2-3	15	EAP7-5	15
EAP2-4	15	EAP7-6	15
EAP2-5	15		
EAP2-6	15	EAP8-1	14
EAP2-7	15	EAP8-2	14
EAP2-8	15	EAP8-3	14
EAP2-9	15	EAP8-4	14
EAP2-10	15	EAP8-5	14
EAP2-11	15	EAP8-6	14
EAP2-12	15	EAP8-7	14
EAP2-13	15	EAP8-8	14
EAP2-14	15	EAP8-9	14
EAP2-15	15	* EAP8-10	16
EAP2-16	15	* EAP8-11	16
		EAP8-12	14
EAP3-1	0		
EAP3-2	0	EAP9-1	5
EAP3-3	11	EAP9-2	5
EAP3-4	0	EAP9-3	13
EAP3-5	13	EAP9-4	0
EAP3-6	0	EAP9-5	1
		EAP9-6	5
EAP4-1	0	EAP9-7	5
EAP4-2	0	EAP9-8	0
EAP4-3	11	EAP9-9	0
EAP4-4	0	EAP9-10	0
EAP4-5	0		
EAP4-6	0	EAP10-1	14
		EAP10-2	14
* EAP5-1	16	EAP10-3	14
* EAP5-2	16	EAP10-4	14
* EAP5-3	16	EAP10-5	14
* EAP5-4	16	EAP10-6	14
* EAP5-5	16	EAP10-7	14
* EAP5-6	16	EAP10-8	14
* EAP5-7	16		
* EAP5-8	16	EAP11-1	0

* Asterisk indicates pages changed, added, or deleted by the current revision.

LIST OF EFFECTIVE PAGES

THE PORTION OF THE TEXT AFFECTED BY THE CHANGE IS INDICATED BY A VERTICAL LINE IN THE OUTER MARGIN OF THE PAGE.

PAGE N°	CHANGE N°	PAGE N°	CHANGE N°
EAP11-2	0		
EAP11-3	0		
EAP11-4	0		
EAP11-5	1		
EAP11-6	0		
EAP11-7	11		
EAP11-8	11		
EAP12-1	14		
EAP12-2	14		
EAP12-3	14		
EAP12-4	14		
EAP12-5	14		
EAP12-6	14		
EAP12-7	14		
EAP12-8	14		
EAP13-1	1		
EAP13-2	1		
EAP13-3	1		
EAP13-4	0		
EAP14-1	0		
EAP14-2	0		
EAP14-3	11		
EAP14-4	11		
EAP14-5	11		
EAP14-6	0		
* PD-1	16		
PD-2	5		
PD-3	8		
PD-4	5		
PD-4A	8		
PD-4B	7		
PD-5	5		
PD-6	5		
PD-7	5		
PD-8	5		
PD-9	5		
PD-10	5		
PD-11	5		
PD-12	5		
PD-13	5		
PD-14	13		
PD-15	13		
PD-16	13		
PD-17	13		
PD-18	13		
PD-19	13		
PD-20	13		
PD-21	13		
PD-22	13		
APP-1	0		
APP-2	0		

* Asterisk indicates pages changed, added, or deleted by the current revision.

APPLICABILITY

This handbook is applicable to the EMB-135, ERJ-140 and the EMB-145 models.

The procedures contained in this handbook have been developed by the manufacturer for use during the operation of the EMB-135, ERJ-140 and EMB-145 models. These procedures are provided as guidance and should not be construed as prohibiting the development of equivalent procedures.

The use of the on board checklist is based on the assumption that both pilots have been properly trained on the type of airplane and, therefore, have a through knowledge of the airplane's systems and procedures.

It further assumes that they know the consequences of not performing the right actions at the right time.

In case of conflicting information between this handbook and the AFM-145/1153 or AFM-140/1330, the AFM must prevail.

INTRODUCTION

The normal checklist is just a memory aid to assist the pilots so they do not forget actions which, if not carried out, can result in some type of risk to the airplane, to the operational environment, to any of its systems, to its occupants or to the passengers comfort. Specific regulations also ask for items to be included in the checklist.

The normal checklist assumes that the pilots previously accomplished all normal procedures.

The normal checklist is named and divided according to each specific phase of flight.

When a disagreement between the response and the checklist answer is found, the checklist should be interrupted until the item is resolved.

Upon completion of the checklist the pilot reading it should state: "_____Checklist Complete".

- ∗ **Items marked with an asterisk are to be performed at least once a day, by flight crew or maintenance personnel, at operator's discretion.**
- ♦ **Items marked with a diamond are to be on through flights.**

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

INTERNAL SAFETY INSPECTION

CHALLENGE	ACTION
Maintenance Status	**CKD**
Cockpit Emergency Equip	**CKD**
Reinforced Cockpit Door Vent Louver (if applicable)	**OPN**
Circuit Breakers	**CKD**
ELECTRIC Panel	**SET**
Emergency Lights	**OFF**
Fire Extinguishing Handle 1	**PUSHED IN**
FUEL Panel	**SET**
APU	**SET**
Ignitions	**AUTO**
START/STOP Selectors	**STOP**
Fire Extinguishing Handle 2	**PUSHED IN**
Ailerons & Rudders Shutoff	**PUSHED IN**
HYDRAULIC Panel	**SET**
Windshield Heating	**PUSHED OUT**
Sensors	**PUSHED IN**
Ice Detection Override	**AUTO**
AIR COND/PNEUM Panel	**SET**
Windshield Wipers (if installed)	**OFF**
Lights	**OFF**
ELT	**ARMED**
Weather Radar	**OFF**
Landing Gear Lever	**DOWN**
Crew Oxygen	**ON**
PASSENGER OXYGEN Panel	**SET**
Gust Lock	**LOCKED**
Speed Brake	**CLSD**
Emergency/Parking Brake	**AS RQRD**
FLAPS Selector Lever	**VERIFY POS**
Alternate Gear Extension Compartment	**CKD**

Copyright © by Embraer. Refer to cover page for details.

POWER UP

	CHALLENGE	ACTION/RESPONSE
	Batteries 1 & 2	AUTO
	Avionics Master 1 or 2	PUSHED IN
*	Batteries Voltage	CKD
*	Backup Battery (EMB XR only)	CKD
	GPU Voltage (if available)	CKD
	Avionics Master 1 & 2	PUSHED OUT
	GPU (if available)	PUSHED IN
	Fuel Pump Power Tank 2	AS RQRD
	Navigation Light	ON
*	FIRE EXTINGUISHER Panel	CKD
	APU	AS RQRD
	Avionics Master 1 & 2	PUSHED IN
	With APU Power	
	GPU (if available)	PUSHED OUT
	3 minutes After APU Start	
	APU Bleed	PUSHED IN
	Air Conditioning	AS RQRD

BEFORE START

	CHALLENGE	RESPONSE
	Manuals & Documents	ON BOARD
	CVR	CKD
	ELECTRICAL Panel	SET
	Emergency Lights	ARM
	Push Button Lights Test (if installed)	CKD
♦	FUEL Panel	SET
*	Fire Detection	CKD
	FIRE EXTINGUISHER Panel	CKD
	POWERPLANT Panel	SET
*	Elec Pump Sys 1 & 2	CKD
	Elec Pump Sys 1 & 2	OFF
♦	PAX SIGNS Panel	SET
	ICE PROTECTION Panel	SET
	AIR COND/PNEUM Panel	SET
	Oxy Masks & Regulators	CKD
	Weather Radar	TST/STBY
	Pitch Trim Cutout Buttons	CKD
	DISPLAY CONTROL Panel	SET
♦	Flight Number & Clocks	SET

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

CHALLENGE	RESPONSE
AHRS (if applicable)	SET
IRS (if applicable)	NAV
Autopilot	CKD
Gust Lock	LOCKED
REVERSIONARY Panel	SET
♦ Flight Instruments	SET/X-CKD
♦ RMU	SET
Thrust Levers	IDLE
♦ Stall Protection System	CKD
TRIM Panel	CKD
♦ PRESSURIZATION Panel	SET
♦ FMS	SET
♦ Fuel QTY	CKD
♦ FMS	SET
♦ Speed Bugs	SET
♦ TRIM Panel	___SET/ZERO/ZERO
♦ Doors & Windows	CLSD
♦ Takeoff Briefing	COMPLETED
♦ Fuel Pump Power	ON
♦ Red Beacon	ON
♦ Emergency/Parking Brake	AS RQRD
♦ Steering	AS RQRD
Safety Pins	ON BOARD

AFTER START

CHALLENGE	RESPONSE
Ground Equipment	REMOVED
ELECTRICAL Panel	SET
APU	AS RQRD
FADEC	RST/ALTN
Elec Hyd Pumps	AUTO
Windshield Heating	AS RQRD
AIR COND/PNEUM Panel	SET
FLAPS	___SET
Flight Controls	CKD
Taxi Lights	ON

Copyright © by Embraer. Refer to cover page for details.

BEFORE TAKEOFF

CHALLENGE	ACTION/RESPONSE
Takeoff Briefing	PERFORM
Ice Protection Test	AS RQRD
Brakes Temperature	CKD
EICAS	CKD
Transponder	TA/RA
Takeoff Configuration	CKD
Gust Lock	UNLOCKED
Elevator	CKD

AFTER TAKEOFF

CHALLENGE	ACTION/RESPONSE
Landing Gear	UP
FLAPS	0
Thrust Rating	CLB
Windshield Heating	AS RQRD
AIR COND/PNEUM Panel	SET
Altimeters	SET/X-CKD
Pressurization	CKD
APU	AS RQRD

DESCENT

CHALLENGE	ACTION
Windshield Heating	PUSHED IN
Approach Briefing	COMPLETED
Speed Bugs	SET
PRESSURIZATION Panel	SET
---	---
External Lights	ON
Pax Signs	SET

APPROACH

CHALLENGE	ACTION/RESPONSE
PASS SIGNS Panel	SET
Altimeters	SET/X-CKD
Approach Aids	SET/X-CKD

BEFORE LANDING

CHALLENGE	ACTION
Landing Gear	DOWN
FLAPS	___SET
Lights	AS RQRD
AP/YD	OFF

Copyright © by Embraer. Refer to cover page for details.

SHUTDOWN

CHALLENGE	ACTION/RESPONSE
Thrust Levers	**IDLE**
Emergency/Parking Brake	**SET**
GPU/APU Generators	**PUSHED IN**
Shed Buses	**AS RQRD**
START/STOP Selectors	**STOP**
Red Beacon	**OFF**
FSTN BELTS	**OFF**
Fuel Pump Pwr	**AS RQRD**
Elec Hyd Pumps	**OFF**
Ice Protection Sys	**OFF**
AIR COND/PNEUM Panel	**SET**

LEAVING THE AIRPLANE

CHALLENGE	ACTION/RESPONSE
IRS (if applicable)	**OFF**
Avionics Master 1 & 2	**PUSHED OUT**
Emergency Lights	**OFF**
External & Internal Lights	**OFF**
PAX SIGNS Panel	**OFF**
Weather Radar	**OFF**
Standby Attitude (if applicable)	**CAGED**
GPU/APU	**OFF**
AIR COND/PNEUM Panel	**SET**
Fuel Pumps	**OFF**
Batteries	**OFF**

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

INTRODUCTION

The Emergency/Abnormal Procedures published in the Quick Reference Handbook (QRH) are provided to pilots as quick guide to minimize the consequences of emergency and abnormal situations that might occur during airplane operation.

In case a discrepancy is found between the QRH and the approved Airplane Flight Manual (AFM), the AFM shall prevail.

Use the QRH requires proper training on the execution of all operational, emergency and abnormal procedures set forth in the AFM and a thorough knowledge of airplane systems.

The procedures set forth herein also require situational awareness for identification of an emergency or abnormal situations and pilot skills to guarantee safety. The Emergency Evacuation procedure accomplishment may be necessary in many situations and its need is at pilot's discretion.

It is EMBRAER recommendation that any unusual situation encountered should be reported as quickly as possible to Flight Operations and Maintenance Personnel.

Three blocks of procedures are contained in this manual:

- **Smoke Procedures: address all annunciated and non annunciated smoke related procedures.**
- **Non Annunciated Procedures: procedures that are not related to an EICAS message but rather to a condition present in the airplane. The Checklists are arranged in alphabetical order with Emergency Checklists first, followed by Abnormal Checklists.**
- **Annunciated Procedures: procedures related to EICAS message. These procedures are grouped by system and the system tabs are in alphabetical order. Each title procedure is followed by the corresponding EICAS message identification. The Checklists for each System Tab are arranged in alphabetical order with the Emergency Checklists first followed by the Abnormal Checklists. The message provided for each procedure represents the root cause of the failure.**

The emergency evacuation procedure is repeated in the last page of QRH, after Performance Data to make it easier to find.

Copyright © by Embraer. Refer to cover page for details.

Some procedures can either be annunciated or non annunciated. In this case, the procedures are presented in the Annunciated block but are referenced in the Non Annunciated index.

In each Annunciated System Tab Index, the related non annunciated procedures are presented with a cross-reference to the Non Annunciated Tab page. The procedures index is classified into Emergency and Abnormal procedures, while EICAS Messages List is classified into Warning, Caution and Advisory messages.

Some EICAS messages do not have an associated QRH procedure. In those cases, "Crew Awareness" identifies the EICAS message as noted in the Index Table. If a Crew Awareness message is displayed on the EICAS, takeoff is prohibited, unless at least one of the following conditions is met:

- **The message is an expected result of an intentional operation;**
- **Flight crew action is taken to clear the message;**
- **Maintenance personnel take action to clear the message;**
- **The airplane is dispatched in accordance with all approved company MEL provisions.**

If one of the following Crew Awareness messages is presented after gate departure, the flight may continue only to the intended destination without further action:

- **AHRS BASIC MODE,**
- **DU 1 (2, 3, 4, 5) FAN FAIL,**
- **E1 (2) OIL IMP BYP,**
- **ENG A/ICEOVERPRES,**
- **IC 1 (2) FAN FAIL or**
- **GEN 1 (2, 3, 4) BRG FAIL.**

Some procedures include a characterization below the title if a relevant emergency/abnormal condition is present, such as aural warnings, lights, EICAS indications, flight instrument flags and the airplane condition itself.

The actions contained in the bold square boxes are recall items. They must be performed expeditiously, by memory.

Flying the airplane is always the priority in any emergency/abnormal situation. Checklists should be called after the flight path is under control, critical phases of flight (such as takeoff and landing) have ended and all recall items have been accomplished.

Copyright © by Embraer. Refer to cover page for details.

Some emergency and abnormal situations require landing at the nearest suitable airport. This statement will be listed at the beginning of a task checklist to give the crew proper time to plan the landing. Also, as an aid for planning the diversion airport, the landing distance correction factor will be presented together with the "Land at nearest suitable airport" statement.

Throughout this manual, a text followed by () means that either condition applies. A text followed by "-" means that both conditions apply simultaneously.

Some procedures require depressurizing the cabin. This will require either dumping the cabin air or the use of manual control to accomplish this task. In this situation, manual control depressurization is the recommended method to be used for passenger comfort and should be made by setting the pressurization mode selector to MAN and smoothly setting the controller to FULL UP. When there is a need to depressurize by a specific method, it will be clearly stated in the procedure.

The procedures contained herein assume that:

- **Airplane systems were operating normally prior to the failure.**
- **All emergency/abnormal actions are performed in the order they are listed in the procedure.**
- **Normal procedures have been properly performed.**
- **Aural warnings are silenced as necessary. Master Warning/Caution lights are reset as soon as the failure is recognized.**
- **All procedures are self-contained. All other messages that may be generated by a single failure do not require that procedures associated to those messages be accomplished in addition to the procedure addressing the root cause.**
- **Circuit breakers must not be pushed in if they pop up.**

All assigned tasks in the procedures have the indication END at the completion of each assigned task. No task is over until END has been reached.

Upon completion of the checklist the pilot reading it should state: "(*Procedure Title)* Checklist Complete".

In the event of multiple failures (excluding cascade failures) with different landing configuration and/or landing distance correction factors, the crew should use good judgment to determine the safest action.

Copyright © by Embraer. Refer to cover page for details.

According to the QRH philosophy, Rejected Takeoff (at or below V_1) procedure is not considered in this manual. Indeed, should the flight crew decide to reject the takeoff; they will do it by memory, not by referring to the QRH. Consequently, Rejected Takeoff (at or below V1) procedure is a matter of flight crew training and is considered within the Standard Operating Procedures Manual, which contains the complete guidance to accomplish it.

Indentation exists when the information is displaced to the right relative to the paragraph that immediately precedes it. The indentation is used to establish a relationship between the indented and the preceding information. An indented information is normally preceded by a condition (e.g. "during landing", "if something is true", "when something happens"). When this is the case, observe the indented information when the preceding condition is satisfied.

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

BAGGAGE SMOKE..... S-3

LAVATORY SMOKE S-3

SMOKE EVACUATION S-4

SMOKE / FIRE / FUMES..... S-6

Copyright © by Embraer. Refer to cover page for details.

LIST OF EICAS MESSAGES

BAGG SMOKE S-3
LAV SMOKE S-3

Copyright © by Embraer. Refer to cover page for details.

BAGGAGE SMOKE

EICAS Warning: BAGG SMOKE

Fire Extg Bagg Button (if installed)..... PUSH IN

LAND AT THE NEAREST SUITABLE AIRPORT.

Shed Buses ..... OFF

Altitude ..... MAINTAIN

Delay the descent as long as possible.

NOTE: Advise Ground Crew of possible Halon vapors approximately 50 minutes after discharging fire extinguishing bottle.

END

LAVATORY SMOKE

EICAS Warning: LAV SMOKE (may be presented)

Lavatory Flush and Lavatory Light CB's (Located in Line E) ..... PULL

Establish contact with the cabin crew.

If necessary:

Diversion ..... CONSIDER

SMOKE EVACUATION Procedure (S-4) ..... ACCOMPLISH

END

Copyright © by Embraer. Refer to cover page for details.

SMOKE EVACUATION

Condition: Smoke or odor inside the cabin and/or cockpit requiring smoke removal.

Crew Oxygen Masks.....DON, 100%
Smoke Goggles.....DON
Crew Communication....ESTABLISH

LAND AT THE NEAREST SUITABLE AIRPORT.

Cockpit Door .....CLOSE

Reinforced Cockpit Door Louver Vent (if applicable) .....CLOSE

Recirculation Fan.....PUSH OUT

Gasper Fan.....PUSH OUT

Pressurization Manual Controller.....1 O'CLOCK POSITION

.....WAIT 15 SECONDS

Pressurization Mode Selector .....PUSH IN (MAN)

Passenger Oxygen .....AS REQUIRED

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

DESIRED EVACUATION RATE?

Fast

Normal

Pressurization Manual Controller ..... AS REQUIRED

Turn the controller clockwise towards UP to adjust desired evacuation rate.

END

Pressurization Manual Controller ..... UP

Packs 1 and 2 ..... PUSH OUT

Bleeds (at least one) .... PUSH IN

Altitude ..... 10’000 FT OR MEA, WHICHEVER IS HIGHER

Recover cabin pressure as soon as smoke has been cleared.

EMERGENCY DESCENT Procedure (NAP-6) .... AS REQUIRED

END

Copyright © by Embraer. Refer to cover page for details.

SMOKE / FIRE / FUMES

Condition: Smoke fire or fumes visually confirmed or identified by odor without an EICAS warning.

Crew Oxygen Masks.....DON, 100%
Smoke Goggles.....DON
Crew Communication....ESTABLISH

LAND AT THE NEAREST SUITABLE AIRPORT.

Recirculation Fan.....PUSH OUT

Gasper Fan.....PUSH OUT

NOTE: Any time smoke becomes dense, perform **SMOKE EVACUATION Procedure(S-4)**.

SMOKE ORIGIN IS OBVIOUS AND CAN BE REMOVED? **No**

Yes

Affected Source.....REMOVE

SMOKE STOPS OR DECREASES? **No**

Yes

SMOKE EVACUATION Procedure (S-4).....AS REQUIRED

END

Cabin Crew.....NOTIFY

FSTN Belts.....ON

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

Thrust Levers ..... IDLE

Speed Brakes ..... OPEN

Airspeed ..... MAX 250 KIAS

Landing Gear ..... DOWN

Altitude ..... 10’000 FT OR MEA, WHICHEVER IS HIGHER

Transponder 7700

ATC ..... NOTIFY

Cockpit Door ..... CLOSE

Reinforced Cockpit Door Louver Vent (if applicable) ..... CLOSE

Pressurization Manual Controller ..... 1 O'CLOCK POSITION

..... WAIT 15 SECONDS

Pressurization Mode Selector ..... PUSH IN (MAN)

Passenger Oxygen ..... AS REQUIRED

Pressurization Manual Controller ..... FULL UP

Packs 1 and 2 ..... PUSH OUT

Shed Buses ..... OFF

Bus Ties ..... OFF

VTRL PUMP SEL (if applicable) ..... SET TO A

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

Fuel Pump 1 .....1A OR 1C

Fuel Pump 2 .....2B

Battery 2 .....OFF

Generators 2 and 4 .....PUSH OUT

Shed Buses, Central DC Bus, DC Bus 2 and Essential Bus 2 deenergized.

SMOKE STOPS OR DECREASES?

No

Yes

Icing Conditions.....EXIT/AVOID

Airspeed .....MAX 250 KIAS

SG On Reversionary Panel 2 .....PUSH IN

NOTE: PFD or MFD information is available in DU 4.

COM 1 on Digital Audio Panel 1.....PUSH IN

Do not set Thrust Lever 2 below idle in flight.

Monitor fuel quantity indication 2 through FMS.

Relevant Inoperative Items:

ADF 2/DME 2/VOR 2/VHF 2/ILS 2/MB 2	
Audio System 2	ISIS/Standby Altimeter
Brakes Inbd	RMU 2
DU 2 and 5	Standby Attitude Indicator
FMS 2	Steering
Ground Spoiler Inbd	Transponder 2

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

NOTE: Landing gear lever can not be moved up.

Landing configuration:

Anticipate flap actuation.

If landing gear has not been selected down:

Gear Electrical Override .....DOORS

.....WAIT 3 SECONDS

Gear Electrical Override .....GEAR/DOORS

Flaps .....45°

V_{REF} .....$V_{REF\ 45°}$ + 5 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.95.

Do not actuate Thrust Reverser 2.

Brake effectiveness will be reduced.

END

IS SUITABLE AIRPORT DISTANT?

No

Yes

Generators 2 and 4 .....PUSH IN

Battery 2 .....AUTO

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

Fuel Pump 1 .....1B

Fuel Pump 2 .....2A OR 2C

VTRL PUMP SEL (if applicable) .....SET TO B

Battery 1 .....OFF

Generators 1 and 3 .....PUSH OUT

Shed Buses, Central DC Bus**,** DC Bus 1 and Essential Bus 1 deenergized.

Emergency lights.....OFF

SMOKE STOPS OR DECREASES?

No

Yes

Icing Conditions.....EXIT/AVOID

SG On Reversionary Panel 1.....PUSH IN

NOTE: PFD or MFD information is available in DU 2.

COM 2 on Digital Audio Panel 2.....PUSH IN

Do not set Thrust Lever 1 below idle in flight.

Monitor fuel quantity indication 1 through FMS.

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

Relevant Inoperative Items:

ADF 1/DME 1/VOR 1/VHF 1/ILS 1/MB 1	
Audio System 1	Ground Spoiler Outbd
Autopilot	Main Pitch Trim
Brakes Outbd	RMU 1
DU 1 and 4	Speed Brake
FMS 1	Transponder 1

NOTE: Landing gear lever can only be moved up using downlock release button (DN Lock Rel).

Landing configuration:

Anticipate flap actuation.

Emergency lights.....ON

Flaps .....45°

V_{REF} .....$V_{REF\ 45°}$ + 5 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.95.

Do not actuate Thrust Reverser 1.

Brake effectiveness will be reduced.

END

Generators 1 and 3 .....PUSH IN

Battery 1 .....AUTO

Backup Battery .....PUSH OUT

WARNING: CONSIDER AN IMMEDIATE LANDING.

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

Landing configuration:

Emergency lights.....ON

Flaps .....45°

V_{REF} .....$V_{REF\ 45°}$

END

Icing Conditions.....EXIT/AVOID

Airspeed .....MAX 250 KIAS

SG On Reversionary Panel 2.....PUSH IN

NOTE: PFD or MFD information is available in DU 4.

COM 1 on Digital Audio Panel 1.....PUSH IN

Do not set Thrust Lever 2 below idle in flight.

Monitor fuel quantity indication 2 through FMS.

Relevant Inoperative Items:

ADF 2/DME 2/VOR 2/VHF 2/ILS 2/MB 2	
Audio System 2	ISIS/Standby Altimeter
Brakes Inbd	RMU 2
DU 2 and 5	Standby Attitude Indicator
FMS 2	Steering
Ground Spoiler Inbd	Transponder 2

NOTE: Landing gear lever can not be moved up.

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

Landing configuration:

Anticipate flap actuation.

If landing gear has not been selected down:

Gear Electrical Override .....DOORS

..... Wait 3 SECONDS

Gear Electrical Override .....GEAR/DOORS

Flaps .....45°

V_{REF} .....$V_{REF45°}$ + 5 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.95.

Do not actuate Thrust Reverser 2.

Brake effectiveness will be reduced.

END

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

AILERON RUNAWAY NAP-3
AIRPLANE OVERSPEED NAP-4
APU OVERTEMPERATURE NAP-4

DITCHING NAP-5
DUAL ENGINE FAILURE ..... refer to EAP 6-3

EMERGENCY DESCENT NAP-6
EMERGENCY EVACUATION NAP-6
ENGINE FIRE, SEVERE DAMAGE OR SEPARATION ..... refer to EAP 6-6
ENGINE OIL LOW PRESSURE ..... refer to EAP 6-7
FORCED LANDING NAP-7
FUEL LEAK NAP-8
INADVERTENT SPOILER OPEN ..... refer to EAP 8-3
JAMMED AILERON NAP-8A
JAMMED ELEVATOR NAP-8A
JAMMED RUDDER NAP-9
PITCH TRIM INOPERATIVE refer to EAP 8-4
PITCH TRIM RUNAWAY NAP-10
RAPID CABIN DEPRESSURIZATION NAP-11
ROLL TRIM RUNAWAY NAP-3
SMOKE EVACUATION refer to S-4
SMOKE / FIRE / FUMES refer to S-6

ABNORMAL ENGINE START NAP-12
ABNORMAL LANDING GEAR EXTENSION NAP-13
ADS-B OUT FAIL OR DEGRADED NAP-14
AILERON ARTIFICIAL FEEL INOPERATIVE NAP-14
APPROACH WARNING NAP-14
ASYMMETRIC RUDDER OPERATION NAP-14
CABIN DEPRESSURIZATION ..... refer to EAP 1-13
CABIN RATE ABNORMAL FLUCTUATIONS ..... refer to EAP 1-13
CAS MESSAGE MISCOMPARISON NAP-14
CDU DATA BUS FAIL FMS ANNUNCIATION ... NAP-15
CRACKED WINDSHIELD NAP-24
DISPLAY FAILURE ..... refer to EAP 2-8
EMERGENCY/PARKING BRAKE HANDLE DISAGREE NAP-15
ENGINE ABNORMAL VIBRATION NAP-15
ENGINE AIRSTART NAP-16

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

ENGINE CONTROL FAILURE..... refer to EAP 6-9
ENGINE FAILURE/SHUTDOWN NAP-19
ENGINE HIGH OIL PRESSURE NAP-20
ENGINE HIGH OIL TEMPERATURE.....NAP-20
ENGINE LOW OIL LEVEL NAP-20
ENGINE OIL LOW PRESSURE.....NAP-20
ENGINE OVERTEMPERATURE NAP-21
ENGINE TAILPIPE FIRE NAP-21
ERRONEOUS STALL PROTECTION ACTUATION.....NAP-22
GEAR LEVER CANNOT MOVE UP AFTER TAKEOFF NAP-22
GUST LOCK FAILURE NAP-23
IC BUS FAILURE ..... refer to EAP 2-11
IC FAILURE ..... refer to EAP 2-11
IMPAIRED OR CRACKED WINDSHIELD NAP-24
IRS/MSU FAILURE ANNUNCIATION.....NAP-25
LOSS OF ENGINE INDICATIONS.....NAP-26
LOSS OF PRESSURIZATION INDICATION NAP-27
MAIN DOOR BLOCKED NAP-28
NAV/FLIGHT INSTRUMENTS FAILURE NAP-28
NOSE LANDING GEAR UP DOOR OPEN NAP-30
ONE ENGINE INOPERATIVE APPROACH AND LANDING.....NAP-30
OVERWEIGHT LANDING.....NAP-31
OXYGEN LEAKAGE NAP-31
PARTIAL OR GEAR UP LANDING.....NAP-32
PRESSURIZATION AUTOMATIC SYSTEM FAILURE ..... refer to EAP 1-13
RADIO ALTIMETER FAIL ..... refer to EAP 2-16
RUDDER ARTIFICIAL FEEL INOPERATIVE NAP-33
RUDDER RUNAWAY NAP-33
SINGLE ENGINE BLEED OPERATION IN ICING CONDITIONS NAP-34
STEERING SYSTEM INOPERATIVE.... refer to EAP 12-7
STIFFENED ELEVATOR NAP-35
STRUCTURAL DAMAGE.....NAP-36
TRANSPONDER FAIL NAP-36
UNCOMMANDED AILERON DISCONNECTION .NAP-36

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

UNCOMMANDED
ELEVATOR DISCONNECTION NAP-36
UNCOMMANDED SWERVING
ON GROUND.....refer to EAP 12-7
UNRELIABLE AIRSPEED.....NAP-37
VOLCANIC ASH NAP-38
YAW TRIM RUNAWAY.....NAP-33

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

AILERON RUNAWAY/ ROLL TRIM RUNAWAY

Condition: Sudden roll.

Quick Disconnect Button..... PRESS AND HOLD
Aileron Shutoff 1 and 2 ..... PUSH OUT

Roll Trim CB (F23) ..... PULL
Quick Disconnect Button..... RELEASE
Airspeed ..... MAX 250 KIAS
Roll Trim Position..... CHECK

ROLL TRIM IN NEUTRAL POSITION? **No**

Yes

Roll Trim CB (F23) ..... PUSH
Prepare to overcome uncommanded roll.
Aileron Shutoff 1..... PUSH IN

RUNAWAY PERSISTS? **No**

Yes

Aileron Shutoff 1..... PUSH OUT
Prepare to overcome uncommanded roll.
Aileron Shutoff 2..... PUSH IN

RUNAWAY STILL PERSISTS? **No**

Yes

Aileron Shutoff 2..... PUSH OUT

Expect greater aileron control force. If required, both pilots should act together to control airplane.

Avoid airports with anticipated turbulence or crosswind.
Perform a long final approach.
Landing configuration:
Landing Gear ..... DOWN
Flaps ..... 22°
V_{REF}..... V_{REF45} + 30 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.85.

END

Aileron Shutoff 1 and 2 ..... PUSH IN
Roll trim is inoperative. Use aileron and rudder to control the airplane.

END

Copyright © by Embraer. Refer to cover page for details.

AIRPLANE OVERSPEED

Aural Warning: Voice Message HIGHSPEED

INDICATION: Airspeed and Mach indications in red range.

Airspeed.....MAX V_{MO}/M_{MO}

END

APU OVERTEMPERATURE

EICAS Indication: EGT enter amber or red range.

EICAS Caution: APU FAIL may be presented.

APU Bleed .....PUSH OUT

.....WAIT 10 SECONDS

EGT REMAINS HIGH?

No

Yes

APU Fuel Shutoff Valve .....PUSH IN

APU Master Knob.....OFF

CAUTION: DO NOT ATTEMPT TO RESTART APU.

END

Consider the APU shutdown if it is not necessary.

END

Copyright © by Embraer. Refer to cover page for details.

DITCHING

ATC ..... NOTIFY

Transponder 7700

FSTN Belts ..... ON

Cabin Crew ..... NOTIFY

Passengers (and Crew) ..... PREPARE FOR DITCHING

Below 10'000 ft:

- **Pressurization Dump Button ..... PUSH IN (ON)**
- **GPWS CB (J7 or J8) ..... PULL**
- **Aural Warn CBs (B4 and E30) ..... PULL**
- **Emerg Lts ..... ON**
- **ELT ..... ON**

At 1'000 ft:

- **VTRL TK XFER (if applicable) ..... OFF**
- **Packs 1 and 2 ..... PUSH OUT**
- **Engine Bleeds 1 and 2 ..... PUSH OUT**

Plan ditching parallel to the line of the wave crests. On final, level the wings and avoid skidding. Touchdown with 4° nose up attitude, and rate of descent less than 180 ft/min.

Ditching configuration:

- **Landing Gear ..... UP**
- **Flaps ..... 45°**

If it is not possible to achieve the selected flap position, maintain airspeed according to the following:

FLAPS POSITION	MIN AIRSPEED
0 to 8°	V_{REF45} + 30 KIAS
9° to 21°	V_{REF45} + 10 KIAS
22° to 44°	V_{REF45} + 5 KIAS
45°	V_{REF45}

Just before touchdown:

- **Cabin ..... ANNOUNCE IMPACT**

WARNING: USE ONLY OVERWING EMERGENCY EXITS FOR PASSENGER EVACUATION. DO NOT OPEN REMAINING DOORS.

Upon water contact:

- **Thrust Levers 1 and 2 ..... IDLE**
- **Start/Stop Selectors 1 and 2 ..... STOP**
- **APU ..... SHUTDOWN**
- **Fire Extinguishing Handles ..... PULL**
- **APU Fuel Shutoff ..... PUSH IN**
- **Engine and APU Fire Extinguishing Bottles (if necessary) ..... DISCHARGE**
- **Fuel Pumps Pwr ..... OFF**
- **Hydraulic Elec Pumps ..... OFF**
- **Evacuation ..... INITIATE**

Before leaving the airplane:

- **Batteries 1 and 2 ..... OFF**

END

Copyright © by Embraer. Refer to cover page for details.

EMERGENCY DESCENT

Cabin Crew	**NOTIFY**
FSTN Belts	**ON**
Thrust Levers	**IDLE**
Speed Brakes	**OPEN**
Airspeed	**MAX 250 KIAS**
Landing Gear	**DOWN**
Descent	**INITIATE**
Altitude	**MEA OR 10'000 FT, WHICHEVER IS HIGHER**

Transponder **7700**

ATC ..... **NOTIFY**

CAUTION: IF STRUCTURAL DAMAGE IS SUSPECTED, USE THE FLIGHT CONTROLS WITH CAUTION AVOIDING HIGH MANEUVERING LOADS AND REDUCING AIRSPEED AS APPROPRIATE.

END

EMERGENCY EVACUATION

Parking Brake ..... **APPLY**

Cabin ..... **DEPRESSURIZE**

Fire Extinguishing Handles ..... **PULL**

APU Fuel Shutoff Valve ..... **PUSH IN**

Engines and APU Fire Extinguishing Bottles (if necessary) ..... **DISCHARGE**

VTRL TK XFER (if applicable) ..... **OFF**

Fuel Pumps Pwr 1 and 2 ..... **OFF**

Hydraulic Elec Pumps 1 and 2 ..... **OFF**

Cabin Crew ..... **NOTIFY**

Emerg Lts ..... **ON**

Evacuation ..... **INITIATE**

ATC ..... **NOTIFY**

Before leaving the airplane:

Batteries 1 and 2 ..... **OFF**

END

Copyright © by Embraer. Refer to cover page for details.

FORCED LANDING

ATC ..... NOTIFY

Transponder 7700

FSTN Belts ..... ON

Cabin Crew ..... NOTIFY

Passengers (and Crew) ..... PREPARE FOR FORCED LANDING

Below 10'000 ft:

- **Pressurization Dump Button ..... PUSH IN (ON)**
- **GPWS CB (J7 or J8) ..... PULL**
- **Aural Warn CBs (B4 and E30) ..... PULL**
- **Emerg Lts ..... ON**
- **ELT ..... ON**

When committed to land:

- **Landing Gear ..... AS REQUIRED**

  The decision to land with all gear up or with any gear extended is left to pilots. The choice of configuration is based on the number of gear available, airplane load distribution, controllability and conditions of the landing field. Ground spoilers and thrust reversers will not operate if any main gear is up.

- **Flaps ..... 45°**

  If it is not possible to achieve the selected flap position, maintain airspeed according to the following:

FLAPS POSITION	MIN AIRSPEED
0 to 8°	V_{REF45} + 30 KIAS
9° to 21°	V_{REF45} + 10 KIAS
22° to 44°	V_{REF45} + 5 KIAS
45°	V_{REF45}

Just before touchdown:

- **Cabin ..... ANNOUNCE IMPACT**
- **Fire Extinguishing Handles ..... PULL**
- **APU Fuel Shutoff Valve ..... PUSH IN**
- **VTRL TK XFER (if applicable) ..... OFF**

When the airplane stops:

- **Engines and APU Fire Extinguishing Bottles (if necessary) ..... DISCHARGE**
- **Fuel Pumps Pwr 1 and 2 ..... OFF**
- **Hydraulic Elec Pumps 1 and 2 ..... OFF**
- **Evacuation ..... INITIATE**

Before leaving the airplane:

- **Batteries 1 and 2 ..... OFF**

END

Copyright © by Embraer. Refer to cover page for details.

FUEL LEAK

Condition:
- FMS Fuel Remaining quantity is above the MFD or EICAS total fuel indication.
- Excessive Fuel flow from one of the engines.
- Fuel imbalance develops.
- Fuel quantity of a tank decreases at an abnormal rate.

LAND AT THE NEAREST SUITABLE AIRPORT.

XFEED .....OFF

Affected Fuel Tank.....IDENTIFY

Asymmetric Thrust .....AS REQUIRED

LEAK ON THE RH TANK? **No**

Yes

APU.....OFF

LEAKAGE CONTINUES AND IS SEVERE? **No**

Yes

ENGINE SHUTDOWN IS CONSIDERED? **No**

Yes

At pilot's discretion:

Associated Thrust Lever.....IDLE
Start/Stop Selector .....STOP
Fire Extinguishing Handle .....PULL (DO NOT ROTATE)
Thrust Rating .....CON
TCAS .....TA ONLY
Altitude .....MAX 25'000 FT, MIN MEA

ICING CONDITIONS? **No**

Yes

XBleed.....OPEN
Altitude .....MAX 15'000 FT, MIN MEA

If it is not possible to descend below 15'000 ft:
Icing Conditions .....EXIT

ONE ENGINE INOPERATIVE APPROACH AND LANDING Procedure (NAP-30) .....ACCOMPLISH

END

Copyright © by Embraer. Refer to cover page for details.

JAMMED AILERON

Condition: Both control wheels can not be moved to either side.

Aileron Disconnection Handle.....PRESS AND PULL

Autopilot.....DISENGAGE

Airspeed .....MAX 200 KIAS

If the right control wheel is jammed, roll trim and artificial feel are not available.

Maintain bank angle below 20°.

If both ailerons are jammed, use rudder to control the airplane.

Avoid abrupt and large aileron inputs.

Avoid airports with anticipated turbulence or crosswind.

Landing configuration:

Flaps .....45°

V_{REF}.....V_{REF45} + 5 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.10.

END

JAMMED ELEVATOR

Condition: Both control columns can not be moved either forward or backward.

EICAS Warning: SPS 1-2 INOP may be presented.

EICAS Caution: STICK PUSHER FAIL may be presented.

Elevator Disconnection Handle.....PRESS AND PULL

Autopilot.....DISENGAGE

Airspeed .....MAX 200 KIAS

Pitch Trim .....AS REQUIRED

Avoid airports with anticipated turbulence or crosswind.

Landing configuration:

Flaps .....22°

V_{REF}.....V_{REF45} + 10 KIAS

If both elevators are jammed, pitch trim may be used to land the airplane.

If left elevator is jammed, Stick Pusher will not be available.

Elevator authority to flare during landing may be reduced.

Do not reengage autopilot.

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.45.

END

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

JAMMED RUDDER

Condition: Pedals can not be moved.

Command rudder through yaw trim.

If required, use asymmetric thrust to control the airplane. Maintain engine asymmetric thrust until nose gear contact in order to avoid lateral and directional miscontrol.

Rudder Shutoff Sys 2 PUSH OUT

RUDDER FREE? — **No**

Yes

END

Rudder Shutoff Sys 2 ..... PUSH IN

Rudder Shutoff Sys 1 ..... PUSH OUT

RUDDER FREE? — **No**

Yes

END

Airspeed ..... MAX 200 KIAS

Avoid airports with anticipated turbulence or crosswind.

During final approach and landing run:

Pilot not flying:

Steering Disengage Button ..... PRESS AND HOLD

Steering Handle ..... AS REQUIRED

Use Steering Handle still keeping the Steering Disengage Button pressed.

CAUTION: DO NOT RELEASE THE NOSEWHEEL STEERING HANDLE UNTIL THE AIRPLANE IS COMPLETELY STOPPED.

Thrust Levers ..... IDLE

If necessary, use differential braking to steer the airplane.

Landing configuration:

Flaps ..... 22°

V_{REF} ..... V_{REF45} + 5 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.62.

END

Copyright © by Embraer. Refer to cover page for details.

PITCH TRIM RUNAWAY

Condition: Uncommanded pitch and trim indication changes.

EICAS Warning: AUTOPILOT FAIL may be presented.

EICAS Caution: AUTO TRIM FAIL may be presented.

Quick Disconnect Button .....PRESS AND HOLD

NOTE: Do not change flap setting.

At safe altitude:

Pitch Trim Main Sys Cutout .....PUSH OUT

Pitch Trim Backup Sys Cutout .....PUSH OUT

Quick Disconnect Button .....RELEASE

WARNING: DO NOT OPEN SPEED BRAKE.

Prepare to overcome unwanted pitch change.
Pitch Trim Backup Sys Cutout .....PUSH IN

RUNAWAY PERSISTS? **No**

Yes

Pitch Trim Backup Sys Cutout .....PUSH OUT
Prepare to overcome unwanted pitch change.
Pitch Trim Main Sys Cutout .....PUSH IN

RUNAWAY STILL PERSISTS? **No**

Yes

Pitch Trim Main Sys Cutout .....PUSH OUT
PITCH TRIM INOPERATIVE
Procedure (EAP 8-4) .....ACCOMPLISH

END

Autopilot can be used.

END

Autopilot .....DISENGAGE

END

Copyright © by Embraer. Refer to cover page for details.

RAPID CABIN DEPRESSURIZATION

Aural Warning: Voice Message **CABIN**
EICAS Indication: CAB ALT Value in red
Condition: Cabin altitude has exceeded 10'000 ft.

Crew Oxygen Masks ..... DON
Crew Communication ..... ESTABLISH
If Emergency Descent is necessary:
- **Cabin Crew ..... NOTIFY**
- **FSTN Belts ..... ON**
- **Thrust Levers ..... IDLE**
- **Speed Brakes ..... OPEN**
- **Airspeed ..... MAX 250 KIAS**
- **Landing Gear ..... DOWN**
- **Descent ..... INITIATE**
- **Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER**

CAUTION: IF STRUCTURAL DAMAGE IS SUSPECTED, USE THE FLIGHT CONTROLS WITH CAUTION AVOIDING HIGH MANEUVERING LOADS AND REDUCING AIRSPEED AS APPROPRIATE.

Passenger Oxygen ..... AS REQUIRED
Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER

END

Copyright © by Embraer. Refer to cover page for details.

ABNORMAL ENGINE START

Condition: Any abnormal engine indication during engine start.

To abort start:

Associated Thrust Lever .....IDLE

Associated Start/Stop Selector .....STOP

FIRE OCCURS OR ENGINE STILL RUNNING? **No**

Yes

Fire Extinguishing Handle.....PULL

ENGINE FAILURE/SHUTDOWN

Procedure (NAP-19).....ACCOMPLISH

END

ANOTHER START CONSIDERED? **No**

Yes

FADEC in Command.....CHECK

FADEC Control Knob.....ALTN

FADEC ALTERNATES? **No**

Yes

Ignition .....OFF

Associated Start/Stop Selector .....START, THEN RUN

.....WAIT 10 SECONDS

N2 indication.....CHECK

N2 INDICATES ROTATION? **No**

Yes

.....WAIT 30 SECONDS

Associated Start/Stop Selector .....STOP

Ignition Selector.....AUTO

Engine .....START

END

Do not attempt another start.

ENGINE FAILURE/SHUTDOWN

Procedure (NAP-19).....ACCOMPLISH

END

Copyright © by Embraer. Refer to cover page for details.

ABNORMAL LANDING GEAR EXTENSION

Condition: Landing gear has not extended by normal means.

HYDRAULIC SYSTEM 1 AVAILABLE AND ELECTRICAL POWER AVAILABLE?

No

Yes

Airspeed ..... MAX 250 KIAS
Landing Gear Lever ..... DOWN
Gear Electrical Override ..... DOORS
..... WAIT 3 SECONDS
Gear Electrical Override ..... GEAR/DOORS
Landing Gear Indication ..... CHECK

THREE LEGS DOWN?

No

Yes

END

Airspeed ..... MAX 200 KIAS
Landing Gear Lever ..... UP
Gear Electrical Override ..... NORMAL
..... WAIT 10 SECONDS
Free Fall Lever ..... ACTUATE
Landing Gear Lever ..... DOWN
Landing Gear Indication ..... CHECK

ANY LEG STILL UP?

No

Yes

Free Fall Lever ..... CYCLE

Cycle the free fall lever as necessary, limited to 10 times, leaving it at the latched position for 20 seconds between each cycle.

ANY LEG STILL UP?

No

Yes

PARTIAL OR GEAR UP LANDING Procedure (NAP-32) ..... ACCOMPLISH

END

Copyright © by Embraer. Refer to cover page for details.

ADS-B OUT FAIL OR DEGRADED

Condition: ADS-B FAIL or ADS-B DGR annunciation displayed on the RMU Radio page or by ATC notification.

Transponder .....SELECT ANOTHER

END

AILERON ARTIFICIAL FEEL INOPERATIVE

Condition: Control Wheel excessively light and oversensitive.

Airspeed.....MAX 200 KIAS

Do not make abrupt and large aileron inputs.

END

APPROACH WARNING

Combiner Message: APCH WARN

MISSED APPROACH Procedure.....PERFORM

A Missed Approach Procedure must be performed, unless the approach is continued under visual conditions and the airplane position and attitude assure a safe landing.

In this case, the AIII guidance must not be followed.

END

ASYMMETRIC RUDDER OPERATION

Condition: Rudder pedals heavier to be moved to one side than the other.

Rudder Shutoff Sys 2 .....PUSH OUT

If the failure persists:

Rudder Shutoff Sys 2 .....PUSH IN

END

CAS MESSAGE MISCOMPARISON

PFD Indication: CAS MSG in amber.

MFD Knob on Reversionary Panel 1 .....EICAS

MFD Knob on Reversionary Panel 2 .....EICAS

Pilot's/Copilot's EICAS Messages .....COMPARE

Discrepant Message .....CHECK

Analyze the situation to check whether the discrepant message is spurious or not, and take the appropriate corrective action.

END

Copyright © by Embraer. Refer to cover page for details.

CDU DATA BUS FAIL FMS ANNUNCIATION

Condition: Affected FMS is not updating CDU.
CDU DATA BUS FAIL message presented on FMS.

Affected FMS.....OFF, then ON

Power down the FMS using the ON-OFF DIM key.

DO NOT use the circuit breaker to cycle power to the FMS.

CDU DATA BUS FAIL MESSAGE PERSISTS? **No**

Yes

Affected FMS.....OFF

END

EMERGENCY/PARKING BRAKE HANDLE DISAGREE

Light: BRAKE ON with Emergency/ Parking Brake Handle not actuated.

Do not take off.

END

ENGINE ABNORMAL VIBRATION

EICAS Indication: Engine vibration enters amber range.

Associated Thrust Lever.....REDUCE TO KEEP VIBRATION WITHIN LIMITS

VIBRATION REMAINS OUT OF LIMITS? **No**

Yes

Affected ENGINE FAILURE/SHUTDOWN Procedure (NAP-19).....ACCOMPLISH

END

TCAS ..... TA ONLY

Monitor engine vibration.

END

Copyright © by Embraer. Refer to cover page for details.

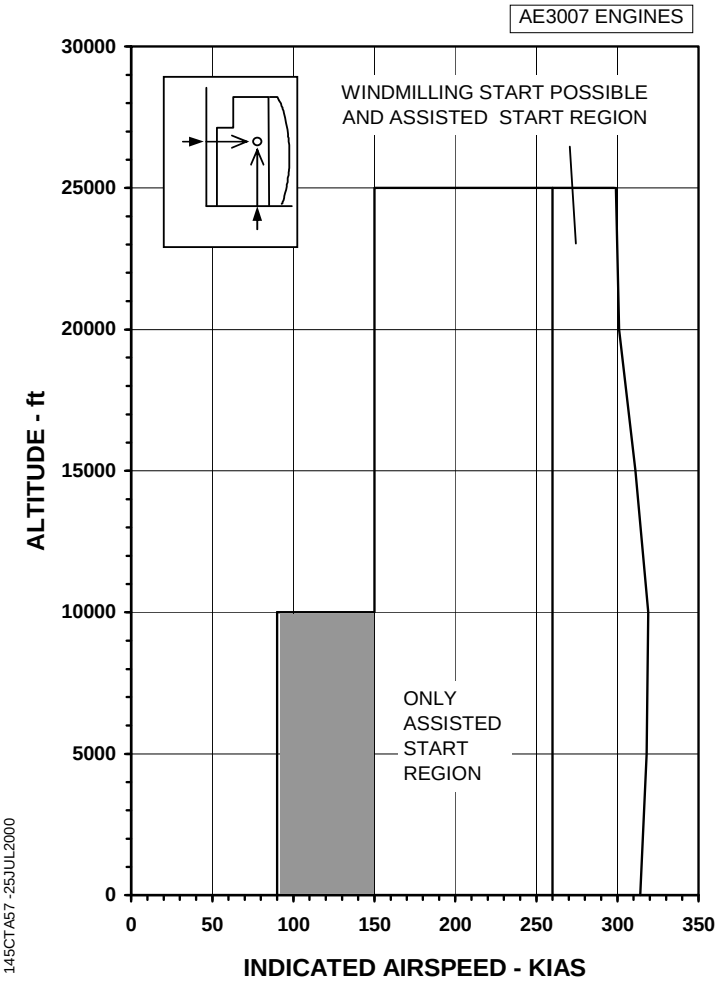
ENGINE AIRSTART

Inoperative engine:

Fuel Pump Selector .....A or B
Fuel Pump Pwr .....ON
Ignition .....AUTO
Start/Stop Selector.....STOP
Engine Bleed .....PUSH OUT
Thrust Lever .....IDLE

Engine Airstart Envelope .....CHECK

ENGINE AIRSTART ENVELOPE



NOTE: Shaded area may be below 1.23 V_{SR}.

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

START PROCEDURE CONSIDERED?

Windmilling

APU Assisted

Crossbleed Assisted

APU ..... START
Operative engine:
Engine Bleed ..... PUSH OUT
APU Bleed ..... PUSH IN
XBleed ..... AUTO OR OPEN

ONLY THE APU GENERATOR IS AVAIABLE?

No

Yes

Windshield heating ..... OFF
Inoperative engine:
Electric Hydraulic Pump ..... OFF

Inoperative engine:
Start/Stop Selector ..... START, THEN RUN

NORMAL START?

No

Yes

Engine Bleeds 1 and 2 ..... AS REQUIRED
XBleed ..... AUTO
APU Bleed ..... AS REQUIRED

END

Do not alternate FADEC.

Consider another start procedure attempt.
Affected ENGINE FAILURE/SHUTDOWN Procedure (NAP-19) ..... AS REQUIRED

END

Operative engine:
N2 ..... 80% MIN
Engine Bleed ..... PUSH IN
XBleed ..... AUTO OR OPEN
Inoperative engine:
Start/Stop Selector ..... START, THEN RUN

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

Non Annunciated

CONTINUED FROM PREVIOUS PAGE

NORMAL START? — No

Yes

Engine Bleeds 1 and 2.....AS REQUIRED
XBleed.....AUTO
APU Bleed.....AS REQUIRED

END

Consider another start procedure attempt.
Affected ENGINE FAILURE/SHUTDOWN Procedure (NAP-19).....AS REQUIRED

END

NOTE: Windmilling starts can be attempted in both engines simultaneously.

Airspeed.....MIN 260 KIAS
Inoperative engine:

N2 .....MIN 10%

Initiate windmilling start with N2 as high as possible. Once N2 is below 10%, it may not be recovered.

Start/Stop Selector.....START, THEN RUN

NORMAL START? — No

Yes

Engine Bleeds 1 and 2.....AS REQUIRED
XBleed.....AUTO
APU Bleed.....AS REQUIRED

END

Consider another start procedure attempt.
Affected ENGINE FAILURE/SHUTDOWN Procedure (NAP-19).....AS REQUIRED

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE FAILURE/SHUTDOWN

Condition: Loss of thrust on an engine or abnormal engine indication or precautionary shutdown.

Associated Thrust Lever.....IDLE
Associated Start/Stop Selector.....STOP

NOTE: If engine shutdown does not occur, pull the associated fire extinguishing handle.

Engine Thrust Rating .....CON
APU (if available) .....START
APU Bleed .....AS REQUIRED
XBleed.....AS REQUIRED
Fuel .....BALANCE

ENGINE RESTART CONSIDERED? **No**

Yes

Affected ENGINE AIRSTART
Procedure (NAP-16).....ACCOMPLISH

END

LAND AT THE NEAREST SUITABLE AIRPORT.

TCAS.....TA ONLY

Altitude .....MAX 25'000 FT, MIN MEA

ICING CONDITIONS? **No**

Yes

XBleed.....OPEN
Altitude .....MAX 15'000 FT, MIN MEA

If it is not possible to descend below 15'000 ft:

Icing Conditions .....EXIT

ONE ENGINE INOPERATIVE APPROACH AND LANDING
Procedure (NAP-30).....AS REQUIRED

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE HIGH OIL PRESSURE

EICAS Indication: Oil pressure pointer in amber range.

OIL TEMPERATURE, OIL LEVEL OR ENGINE VIBRATION OUT OF LIMITS?

No

Yes

Associated Procedure .....ACCOMPLISH

END

ENGINE HIGH OIL TEMPERATURE

EICAS Indication: Oil temperature pointer and digits become red.

Associated Thrust Lever .....REDUCE

FAILURE PERSISTS?

No

Yes

ABOVE 25'000 FT?

No

Yes

Altitude .....MAX 25'000 FT
MINIMUM MEA

FAILURE PERSISTS?

No

Yes

Affected ENGINE FAILURE/SHUTDOWN Procedure (NAP-19) ..... ACCOMPLISH

END

ENGINE LOW OIL LEVEL

MFD Indication: Oil quantity enters amber range.

Affected ENGINE FAILURE/SHUTDOWN Procedure (NAP-19) .....AS REQUIRED

Consider shutting the engine down to preserve oil quantity, and if required restart it prior to landing.

NOTE: The indication of oil-level is accurate above 3 quarts.

END

ENGINE OIL LOW PRESSURE

EICAS Indication: Oil pressure in amber range.

Associated Thrust Lever .....REDUCE

Reduce N2 below 88%.

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE OVERTEMPERATURE

Condition: ITT pointer and digits flashing amber or red.

Associated Thrust Lever.....REDUCE

ITT INDICATION WITHIN LIMITS? **No**

Yes

Operate at reduced thrust to keep ITT within limits.

END

Associated Bleed.....PUSH OUT

Altitude .....MAX 25'000 FT, MIN MEA

ITT INDICATION STILL OUT OF LIMITS? **No**

Yes

ENGINE FAILURE/SHUTDOWN Procedure (NAP-19).....ACCOMPLISH

END

Operate at reduced thrust to keep ITT within limits.

TCAS ..... TA ONLY

END

ENGINE TAILPIPE FIRE

Condition: Tailpipe fire was detected visually by crew or ground personnel. No EICAS message displayed.

Affected engine:

- **Thrust Lever.....IDLE**
- **Start/Stop Selector .....STOP**
- **Ignition.....OFF**
- **Fuel Pump .....OFF**
- **XFeed Selector Knob .....OFF**
- **Start/Stop Selector .....START, THEN RUN**

ITT .....MONITOR

ATC .....NOTIFY

.....WAIT 90 SECONDS

Associated Start/Stop Selector.....STOP

Associated Fire Extinguishing Handle .....PULL (DO NOT ROTATE)

NOTE: If fire is not extinguished while the engine is motored, ground personnel support must be requested.

END

Copyright © by Embraer. Refer to cover page for details.

ERRONEOUS STALL PROTECTION ACTUATION

Condition: Inadvertent shaker and/or pusher actuation.

Immediately and simultaneously:

Quick Disconnect Button.....PRESS

Stall Protection Cutout 1 and 2 .....PUSH OUT

Minimum Airspeed.....FLAP MANEUVERING SPEED (PD-2)

Avoid skidding the airplane.

To approach and go-around speeds, add 5 KIAS to V_{REF}.

Landing configuration:

Landing Gear ..... DOWN

Flaps ..... 45°

Airspeed ..... $V_{REF\ 45}$ + 5 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.10.

END

GEAR LEVER CANNOT MOVE UP AFTER TAKEOFF

Condition: Landing gear cannot be moved to up position after takeoff in the normal manner.

.....WAIT 10 SECONDS

LG AIR/GND FAIL MESSAGE DISPLAYED? **No**

Yes

Landing Gear Lever .....DO NOT MOVE

LANDING GEAR AIR/GROUND SYSTEM FAILURE Procedure (EAP 12-6) .....ACCOMPLISH

END

Downlock Release Button (DN LOCK REL) .....PRESS

Landing Gear Lever .....UP

END

Copyright © by Embraer. Refer to cover page for details.

GUST LOCK FAILURE

Light: GUST LOCK (amber).

CONDITION?

On Ground

In Flight

Consider diversion.
Do not push control column full Nose Down.
Avoid flying in turbulence conditions.
Avoid airports with anticipated turbulence or crosswind.

END

Do not takeoff.

END

Copyright © by Embraer. Refer to cover page for details.

IMPAIRED OR CRACKED WINDSHIELD

Associated Ice Protection

Windshield.....PUSH OUT

Cockpit Door.....CLOSE

ONLY OUTER LAYER CRACKED? **No**

Yes

END

Oxygen Masks.....AS REQUIRED

Smoke Goggle.....DON

Airspeed.....MAX 250 KIAS

Altitude.....MEA OR 10'000 FT, WHICHEVER IS HIGHER

Pressurization Manual Controller .....1 O'CLOCK POSITION

.....WAIT 15 SECONDS

Pressurization Mode Selector.....PUSH IN (MAN)

Pressurization Manual Controller .....CAB MAX ΔP = 1 PSI

FORWARD VISIBILITY GOOD IN ONE SIDE? **No**

Yes

Pilot flying must be on non impaired side.

END

When reaching 10'000 ft:

Pressurization Mode Selector .....PUSH OUT

Pressurization Dump Button .....PUSH IN

During approach and landing, when visibility is required:

Airspeed.....MAX 140 KIAS, MIN V_{REF45}

Check no loose objects in the cockpit.

Direct Vision Window.....REMOVE

Landing must be made by looking through Direct Vision Window. Intercommunication will be impossible with window removed.

END

Copyright © by Embraer. Refer to cover page for details.

IRS/MSU FAILURE ANNUNCIATION

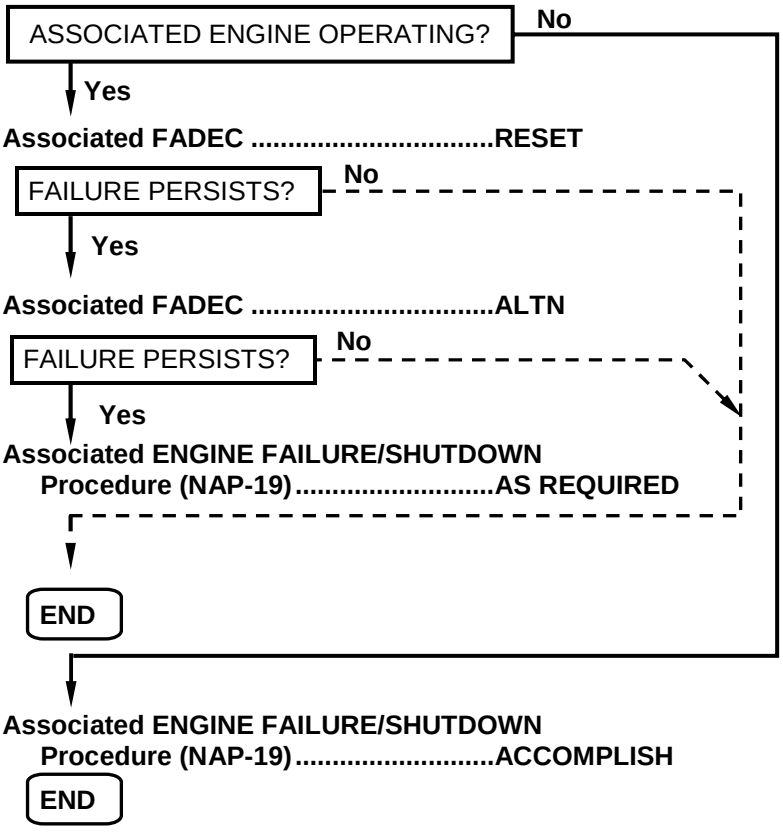
LIGHT	OPERATION PHASE		
	POWER ON	**ALIGNMENT**	**IN FLIGHT**
ALIGN	– No light: – Check IRS CBs. – Set mode select switch to ALIGN or NAV. – Press MSU Test switch. Annunciator bulb must be replaced if the other MSU annunciators do light.	Accomplish associated **IRS ALIGNMENT FAULT** Procedure (EAP 2-12)	– Flashes: – IRS data is not normal. – Accomplish associated **IRS OVERHEAT** Procedure (EAP 2-15)
FAULT	– Set mode select switch to OFF for at least 3 sec. Then set mode select switch back to ALIGN or NAV. – If the annunciator remains lighted, do not takeoff.	– Associated with ALIGN annunciation: – Check and reenter latitude. – Allow additional time for alignment. – Try new alignment. Set mode select switch to OFF for at least 3 sec, then to ALIGN, and enter present position. – If on ground, do not takeoff.	– Select the remaining IRU by pressing the IRS Button on the associated reversionary panel. – If necessary set mode select switch to ATT.
NO AIR	– Do not takeoff.	*******	– Operate IRU until flight completion. – If fault annunciator is ON or inertial data ceases to be transmitted by IRU, accomplish associated **IRS OVERHEAT** Procedure (EAP 2-15). IRS data is not normal.
BATT FAIL	– Do not takeoff.	*******	*******
ON BATT	– Check IRS CBs. – If the annunciator remains lighted, do not takeoff.	*******	Accomplish associated **IRS ON BATTERY** Procedure (EAP 2-14).

END

Copyright © by Embraer. Refer to cover page for details.

LOSS OF ENGINE INDICATIONS

Condition: Loss of Thrust Mode, ITT, N1 and N2 indications.



Copyright © by Embraer. Refer to cover page for details.

LOSS OF PRESSURIZATION INDICATION

Condition: Cabin altitude or cabin ΔP is not being presented or during use of the pressurization manual control.

Use the remaining indications to maintain cabin altitude below 10'000 ft, according to the table below:

AIRPLANE/CABIN ALTITUDE CONVERSION TABLE

AIRPLANE ALTITUDE (ft)	CABIN ALTITUDE (ft)	DIFFERENTIAL PRESSURE (psi)
10000	300	4.4
11000	500	4.7
12000	700	5.0
13000	900	5.2
14000	1100	5.5
15000	1300	5.7
16000	1500	5.9
17000	1700	6.1
18000	1900	6.3
19000	2200	6.5
20000	2400	6.7
21000	2700	6.8
22000	2900	7.0
23000	3200	7.1
24000	3400	7.2
25000	3800	7.3
26000	4100	7.4
27000	4400	7.5
28000	4700	7.6
29000	5000	7.6
30000	5400	7.7
31000	5700	7.7
32000	6100	7.7
33000	6500	7.7
34000	6800	7.8
35000	7200	7.8
36000	7600	7.8
37000	8000	7.8

If pressurization can not be maintained, refer to PRESSURIZATION AUTOMATIC SYSTEM FAILURE/CABIN DEPRESSURIZATION Procedure (EAP 1-13).

END

Copyright © by Embraer. Refer to cover page for details.

Non Annunciated

MAIN DOOR BLOCKED

Light: DOOR BLOCKED (Attendant's Panel).

CONDITION?

In Flight

On Ground

Hydraulic System 1.....OFF

Main Door Alternative Opening Valve.....TURN CLOCKWISE AND HOLD

.....WAIT 2 MINUTES

Outside Area.....CHECK CLEAR

Door.....OPEN

END

NAV/FLIGHT INSTRUMENTS FAILURE

ANNUNCIATOR/ FAILURE	LOCATION	ACTION
ATT: CAGE (amber)	**ISIS**	Press the CAGE push button in order to recover attitude indication. Caging the ISIS in flight will result in loss of attitude indication for up to 10 seconds and the amber message ATT 10s will be presented during this time. Use the primary indication source until attitude indication is available.
ATT, ALT, SPD, M, HDG (red)		Use the primary indication source. If on ground, do not takeoff.
MENU INOP (amber)	**MFD**	Do not takeoff.
HDG FAIL (red)	**PFD** **MFD**	Use cross-side heading by pressing the AHRS (IRS) button on associated reversionary panel or use RMU or standby attitude indicator.

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

ANNUNCIATOR/ FAILURE	LOCATION	ACTION
HDG (amber)	**PFD MFD**	Identify the failed side by comparing data with the Magnetic Compass and use cross-side data by pressing the AHRS (IRS) on associated reversionary panel.
ALT, ATT, IAS, PIT or ROL (amber)	**PFD**	Compare data with Standby Indicator. For altitude, compare the PFD altimeters setting also. If required, use cross-side data by pressing the appropriate button on associated reversionary panel.
ATT FAIL (red)		Push in the AHRS (IRS) button on associated reversionary panel or use standby attitude indicator.
RA (amber)		Compare both radio altimeter indications. If required, consider only the lower indication. Otherwise, disregard Radio Altitude. If RA is displayed in the center, the RA is failed.
RA1 (2) (amber)		Radio altimeter automatic reversion has occurred. No action is required.
VS (red)		Push in the ADC button on associated reversionary panel.
"X" (red) over IAS tape and/or altitude tape		Push in the ADC button on associated reversionary panel or use standby indications.
"X" (red) over course scale		Select another sensor.
Blank or "X" (red) on PFD or EICAS		Use the MFD Knob to present the required information on MFD.
Errors in attitude indication	**Standby Attitude Instrument**	Maintain a straight and leveled flight using the primary indication source. Wait 3 minutes. If error persists, cage the instrument and wait 5 min.

END

Copyright © by Embraer. Refer to cover page for details.

Non Annunciated

NOSE LANDING GEAR UP DOOR OPEN

Condition: Noise increase due to nose landing gear doors open.

Airspeed ..... MAX 250 KIAS

Icing Conditions ..... AVOID/EXIT

Fuel Consumption ..... MONITOR

ABNORMAL LANDING GEAR EXTENSION

Procedure (NAP-13) ..... AS REQUIRED

END

ONE ENGINE INOPERATIVE APPROACH AND LANDING

For CAT III mode or CAT II approaches using HGS, the normal CAT III approach procedure must be used.

Approach:

Altimeters ..... SET AND CROSS CHECKED

Approach Aids ..... SET AND CROSS CHECKED

Speed Bugs ..... SET

Pressurization ..... CHECK

Go-Around Procedure ..... REVIEW

- Disengage Autopilot.
- Press Go-Around Button.
- Advance Operative Engine Thrust Lever to MAX.
- Rotate airplane to 10° nose up.
- Set flaps to 9°.

With positive rate of climb:

- Landing gear up.
- Maintain Approach Climb Speed until reaching acceleration altitude (level off).

Before Landing:

Inoperative Engine Thrust Lever ..... IDLE

Landing Gear ..... DOWN

Thrust Rating ..... TAKEOFF MODE

Fuel XFeed ..... OFF

Autopilot/Yaw Damper ..... DISENGAGE

Landing configuration:

Flaps ..... 22°

V_{REF} ..... V_{REF45} + 10 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.48.

END

ORH-145/1115

Copyright © by Embraer. Refer to cover page for details.

OVERWEIGHT LANDING

Before touchdown:

Emerg Lts ..... AS REQUIRED
APU Fuel Shutoff Valve..... CLOSE
Rate of Descent ..... MAX 300 FT/MIN

Touch smoothly the runway surface.
Reduce the engine thrust only after the touchdown.
Landing configuration:

Flaps ..... 45°
V_{REF} ..... V_{REF45}

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY AT LEAST 1.10.

END

OXYGEN LEAKAGE

Condition: Evidence of oxygen leakage through the crew mask, mask hose, flow indicator (blinker), or oxygen line.

No Smoking ..... ON

IS THE LEAKAGE IN THE CREW MASK, MASK HOSE, OR FLOW INDICATOR?

No

Yes

Affected Mask ..... REMOVE FROM STOWAGE BOX
Stowage Box Doors..... CLOSE
Shutoff Sliding Control ..... ACTUATE
Oxygen Pressure ..... CHECK

OXYGEN PRESSURE BELOW MINIMUM REQUIRED FOR DISPATCH?

No

Yes

Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER

END

Keep one Portable Oxygen Cylinder available for the pilot of the affected side. The oxygen supply by the Portable Oxygen Cylinder will last at least 30 minutes.

END

Oxygen Cylinder Shutoff Valve ..... OFF
Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER

END

Copyright © by Embraer. Refer to cover page for details.

PARTIAL OR GEAR UP LANDING

Condition: Airplane committed to land with gear up or in transit.

EICAS Indication: Abnormal landing gear position.

EICAS Warning: LG/LEVER DISAGREE may be presented.

ATC.....NOTIFY

Burn fuel to reduce touchdown speed.

Transponder.....7700

FSTN Belts.....ON

Cabin Crew.....NOTIFY

Passengers (and Crew).....PREPARE FOR EMERGENCY LANDING AND EVACUATION

Below 10'000 ft:

- **GPWS CB (J7 or J8).....PULL**
- **Aural Warn CBs (B4 and E30).....PULL**
- **Emerg Lts.....ON**
- **ELT.....ON**

Prior to approach:

- **Hydraulic Elec Pumps 1 and 2.....OFF**
- **Cabin.....DEPRESSURIZE**
- **Engine Bleeds 1 and 2.....PUSH OUT**

When committed to land:

- **Landing Gear.....AS REQUIRED**

  The decision to land with all gear up or with any gear extended is left to pilots. Ground spoilers and thrust reversers will not operate if any main gear is up.

- **Flaps.....45°**

  If it is not possible to achieve the selected flap position, maintain airspeed according to the following:

FLAPS POSITION	MIN AIRSPEED
0 to 8°	V_{REF45} + 30 KIAS
9° to 21°	V_{REF45} + 10 KIAS
22° to 44°	V_{REF45} + 5 KIAS
45°	V_{REF45}

Just before touchdown:

- **Cabin..... ANNOUNCE IMPACT**

Apply thrust reverser (if available) after touchdown.

When the airplane stops:

- **Fire Extinguishing Handles.....PULL**
- **APU Fuel Shutoff Valve.....CLOSE**
- **Engines and APU Fire Extinguishing Bottles (if necessary).....DISCHARGE**
- **Fuel Pumps Pwr 1 and 2.....OFF**
- **Evacuation.....INITIATE**

Before leaving the airplane:

- **Batteries 1 and 2.....OFF**

END

Copyright © by Embraer. Refer to cover page for details.

RUDDER ARTIFICIAL FEEL INOPERATIVE

Condition: Rudder pedals become light and do not center by themselves. Yaw trim does not operate properly.

Rudder Shutoff Sys 2 ..... PUSH OUT

FAILURE PERSISTS? **No**

Yes

Rudder Shutoff Sys 1 ..... PUSH OUT

Expect greater rudder pedal force. If required, both pilots should act together to control the airplane.

Consider the use of aileron to help in yaw control, and asymmetric thrust to trim the airplane.

Avoid airports with anticipated turbulence or crosswind.

END

RUDDER/YAW TRIM RUNAWAY

Condition: Sudden uncommanded yaw.

EICAS Indication: Associated yaw trim indication changes.

Quick Disconnect Button ..... PRESS AND HOLD

Rudder Shutoff Sys 1 and 2 ..... PUSH OUT

Airspeed ..... MAX 250 KIAS

Yaw Trim Position ..... CHECK

YAW TRIM DISPLACED FROM NEUTRAL? **No**

Yes

Yaw Trim CB (F12) ..... PULL

Quick Disconnect Button ..... RELEASE

Rudder Shutoff Sys 1 and 2 ..... PUSH IN

END

Quick Disconnect Button ..... RELEASE

Prepare to overcome uncommanded yaw.

Rudder Shutoff Sys 1 ..... PUSH IN

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

Non Annunciated

CONTINUED FROM PREVIOUS PAGE

RUNAWAY PERSISTS? — No

Yes

Rudder Shutoff Sys 1 .....PUSH OUT

Prepare to overcome uncommanded yaw.

Rudder Shutoff Sys 2 .....PUSH IN

RUNAWAY STILL PERSISTS? — No

Yes

Rudder Shutoff Sys 2 .....PUSH OUT

LAND AT NEAREST SUITABLE AIRPORT.

Expect greater rudder pedals force. If required, both pilots should act together to control airplane.

Consider the use of aileron to help in yaw control, and asymmetric thrust to trim the airplane.

Avoid airports with anticipated turbulence or crosswind.

END

SINGLE ENGINE BLEED OPERATION IN ICING CONDITIONS

XBleed ..... OPEN

Altitude ..... MAX 15’000 FT, MIN MEA

If it is not possible to descend below 15’000 ft:

Icing Conditions ..... EXIT

END

Copyright © by Embraer. Refer to cover page for details.

STIFFENED ELEVATOR

Condition: Elevator control columns movement is stiffened.

EICAS Warning: AUTOPILOT FAIL (may be presented), SPS 1-2 INOP (may be presented)

EICAS Caution: STICK PUSHER FAIL (may be presented)

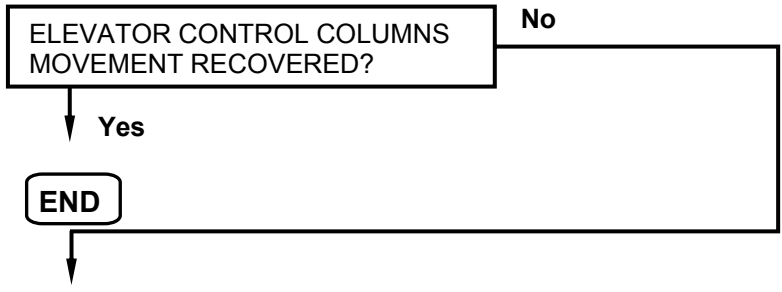
Autopilot ..... DISENGAGE

Pitch Trim ..... AS REQUIRED

Avoid abrupt and large elevator inputs.

Consider descent to a warmer altitude.

Freezing conditions may lead to de/anti-icing fluids residues to stiffen the elevator.



JAMMED ELEVATOR

Procedure (NAP-8)..... ACCOMPLISH

END

Copyright © by Embraer. Refer to cover page for details.

STRUCTURAL DAMAGE

Land at the nearest suitable airport.

At safe altitude, evaluate airplane aero dynamical behavior and take extra caution varying airspeed and attitude.

Use the flight controls with caution avoiding high maneuvering loads.

FUSELAGE DAMAGED OR SUSPECTED DAMAGED?

No

Yes

Oxygen Masks ..... AS REQUIRED

Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER

When reaching 10'000 ft:

Pressurization Dump Button ..... PUSH IN

Land as smooth as possible.

END

TRANSPONDER FAIL

Condition: The transponder mode annunciation is replaced with dashes on the RMU Radio page or by ATC notification.

TRANSPONDER ..... SELECT ANOTHER

END

UNCOMMANDED ELEVATOR OR AILERON DISCONNECTION

Light: Amber ELEV DISC or AIL DISC on Pedestal.

Condition: One control column or control wheel moves independently of the other.

Affected Surface Disconnection Handle ..... PULL

If aileron is affected, aileron artificial feel not available on left side.

Avoid airports with anticipated turbulence or crosswind.

END

Copyright © by Embraer. Refer to cover page for details.

UNRELIABLE AIRSPEED

Autopilot/Yaw Damper .....DISENGAGE

Both Flight Directors.....OFF

SPEED INDICATION ON BOTH PFD BELOW 135 KIAS?

No

Yes

Rudder Shutoff Sys 1 .....PUSH OUT

CAUTION: AVOID USING THE SPEEDBRAKE.

Attitude/Thrust .....ADJUST

Maintain airplane control. Refer to Unreliable Airspeed tables in the Performance Data section. Altitude and/or Vertical Speed indications may also be unreliable.

Ground speed indication is available in the FMS for reference.

END

Copyright © by Embraer. Refer to cover page for details.

VOLCANIC ASH

LAND AT THE NEAREST SUITABLE AIRPORT.

Volcanic Ash Area..... EXIT/AVOID

Consider performing a 180º turn.

Oxygen Masks (if necessary)..... DON, 100%

If a significant amount of volcanic ash fills the cockpit or if there is a strong smell of sulphur, don an oxygen mask and select 100%.

APU..... START

Ignitions ..... ON

Thrust Levers (if altitude permits)..... IDLE

Anti-Icing Buttons (Engine, Wing and Stabilizer) ..... CHECK PUSHED IN

Ice Detection Override Knob..... ALL

ITT..... MONITOR

ITT INCREASES? **No**

Yes

Affected Engine..... SHUT DOWN

After affected engine has cooled down:

Affected Engine..... RESTART

If the affected engine fails to restart, repeated attempts should be made immediately.

Airspeed..... MONITOR

ABNORMAL AIRSPEED INDICATIONS? **No**

Yes

UNRELIABLE AIRSPEED Procedure (NAP-37) ..... ACCOMPLISH

Restore systems to normal operation.

ATC..... NOTIFY

END

ORH-145/1115

Copyright © by Embraer. Refer to cover page for details.

INDEX BY EICAS MESSAGE

TYPE	MESSAGE	PAGE OR INSTRUCTION
WARNING	APU FIRE	EAP 3-3
	ATTCS FAIL	EAP 6-3
	AUTOPILOT FAIL	EAP 2-3
	BAGG SMOKE	S-3
	BATT 1 OVTEMP	EAP 5-3
	BATT 2 OVTEMP	EAP 5-3
	BLD 1 LEAK	EAP 1-4
	BLD 1 OVTEMP	EAP 1-5
	BLD 2 LEAK	EAP 1-4
	BLD 2 OVTEMP	EAP 1-5
	BLD APU LEAK	EAP 1-3
	E1 ATTCS NO MRGN	EAP 6-5
	E1 LOW N1	**Crew Awareness**
	E1 OIL LOW PRESS	EAP 6-7
	E2 ATTCS NO MRGN	EAP 6-5
	E2 LOW N1	**Crew Awareness**
	E2 OIL LOW PRESS	EAP 6-7
	ELEC ESS XFR FAIL	EAP 5-3
	ENG 1 FIRE	EAP 6-6
	ENG 2 FIRE	EAP 6-6
	ENG 1-2 OUT	EAP 6-3
	FUEL 1 LO LEVEL	EAP 9-3
	FUEL 2 LO LEVEL	EAP 9-3
	GPWS	**Crew Awareness**
	ICE COND-A/I INOP	EAP 11-3
	LAV SMOKE	S-3
	LG/LEVER DISAGREE	EAP 12-3
	MAIN DOOR OPN	EAP 4-3
	NO TAKEOFF CONFIG	EAP 14-4
	PIT TRIM 1 INOP	EAP 8-4
	PIT TRIM 2 INOP	EAP 8-4
	PTRIM BACKUP INOP	EAP 8-4
	PTRIM MAIN INOP	EAP 8-4
	SERVICE DOOR OPN	EAP 4-3
	SPS 1 INOP	EAP 14-3
	SPS 1-2 INOP	EAP 14-4
	SPS 2 INOP	EAP 14-3

Copyright © by Embraer. Refer to cover page for details.

TYPE	MESSAGE	PAGE OR INSTRUCTION
CAUTION	115 VAC BUS OFF	EAP 5-5
	A/ICE LOW CAPACIT	EAP 11-4
	A/ICE SWITCH OFF	EAP 11-4
	ACCESS DOORS OPN	EAP 4-4
	AHRS 1 ALN FAULT	EAP 2-3
	AHRS 1 FAIL	EAP 2-4
	AHRS 1 OVERHEAT	EAP 2-4
	AHRS 2 ALN FAULT	EAP 2-3
	AHRS 2 FAIL	EAP 2-4
	AHRS 2 OVERHEAT	EAP 2-4
	AIL SYS 1 INOP	EAP 8-6
	AIL SYS 1-2 INOP	EAP 10-3
	AIL SYS 2 INOP	EAP 8-6
	AOA 1 HEAT INOP	EAP 11-4
	AOA 2 HEAT INOP	EAP 11-4
	AP AIL MISTRIM	EAP 2-5
	AP ELEV MISTRIM	EAP 2-5
	APU BLD VLV FAIL	EAP 1-5
	APU CNTOR CLSD	EAP 3-4
	APU EXTBTL INOP	EAP 7-3
	APU FAIL	EAP 3-4
	APU FIREDET FAIL	EAP 7-3
	APU FUEL LO PRESS	EAP 3-4
	APU FUEL SOV INOP	EAP 9-4
	APU GEN OFF BUS	EAP 5-12
	APU GEN OVLD	EAP 5-12
	APU OIL HI TEMP	EAP 3-5
	APU OIL LO PRESS	EAP 3-5
	AURAL WARN FAIL	EAP 14-6
	AUTO TRIM FAIL	EAP 2-6
	BAGG EXTBTL INOP	EAP 7-4
	BAGGAGE DOOR OPN	EAP 4-4
	BATT1 OFF BUS	EAP 5-5
	BATT2 OFF BUS	EAP 5-5
	BKUP BATT OFF BUS	EAP 5-5
	BLD 1 LOW TEMP	EAP 1-6
	BLD 1 VLV FAIL	EAP 1-7
	BLD 2 LOW TEMP	EAP 1-6
	BLD 2 VLV FAIL	EAP 1-7
	BRAKE DEGRADED	EAP 12-5
	BRAKE OVERHEAT	EAP 12-4
	BRK INBD INOP	EAP 12-5
	BRK OUTBD INOP	EAP 12-5
	CHECK IC 1 SW	**Crew Awareness**
	CHECK IC 2 SW	**Crew Awareness**
	CHECK PFD 1	EAP 2-8
	CHECK PFD 2	EAP 2-8

Copyright © by Embraer. Refer to cover page for details.

TYPE	MESSAGE	PAGE OR INSTRUCTION
	CHK IC CONFIG	**Crew Awareness**
	CROSS BLD FAIL	EAP 1-8
	CROSS BLD SW OFF	**Crew Awareness**
	CREW OXYGEN LO PRESS	EAP 13-3
	DAU AC ID MISCMP	**Crew Awareness**
	DAU1 A FAIL	EAP 2-6
	DAU1 ENG MISCOMP	EAP 2-7
	DAU1 SYS MISCOMP	EAP 2-7
	DAU1 WRN MISCOMP	EAP 2-7
	DAU2 A FAIL	EAP 2-6
	DAU2 ENG MISCOMP	EAP 2-7
	DAU2 SYS MISCOMP	EAP 2-7
	DAU2 WRN MISCOMP	EAP 2-7
	DC BUS 1 OFF	EAP 5-6
	DC BUS 2 OFF	EAP 5-7
	DEFUEL NOT CLOSED	EAP 9-4
	DFDR FAIL	**Crew Awareness**
	E1 A/ICE FAIL	EAP 11-5
	E1 ATS SOV OPN	EAP 6-8
	E1 CTL FAIL	EAP 6-9
	E1 EXCEEDANCE	**Crew Awareness**
	E1 EXTBTLA INOP	EAP 7-5
	E1 EXTBTLB INOP	EAP 7-5
	E1 FIREDET FAIL	EAP 7-4
	E1 FPMU NO DISP	**Crew Awareness**
	E1 FUEL LO PRESS	EAP 9-5
	E1 FUEL LO TEMP	EAP 9-5
	E1 FUEL SOV INOP	EAP 9-6
	E1 NO DISP	**Crew Awareness**
	E2 A/ICE FAIL	EAP 11-5
	E2 ATS SOV OPN	EAP 6-8
	E2 CTL FAIL	EAP 6-9
	E2 EXCEEDANCE	**Crew Awareness**
	E2 EXTBTLA INOP	EAP 7-5
	E2 EXTBTLB INOP	EAP 7-5
	E2 FIREDET FAIL	EAP 7-4
	E2 FPMU NO DISP	**Crew Awareness**
	E2 FUEL LO PRESS	EAP 9-5
	E2 FUEL LO TEMP	EAP 9-5
	E2 FUEL SOV INOP	EAP 9-6
	E2 NO DISP	**Crew Awareness**
	ELEC EMERG ABNORM	EAP 5-8
	ELEKBAY OVTEMP	EAP 2-9
	EMERG EXIT OPN	EAP 4-5
	EMERG LT NOT ARMD	EAP 5-8
	EMRG BRK LO PRES	EAP 12-5
	ENG1 REV DISAGREE	EAP 6-12
	ENG1 REV FAIL	EAP 6-12

Copyright © by Embraer. Refer to cover page for details.

TYPE	MESSAGE	PAGE OR INSTRUCTION
CAUTION	ENG2 REV DISAGREE	EAP 6-12
	ENG2 REV FAIL	EAP 6-12
	ENG NO TO DATA	**Crew Awareness**
	ENG REF A/I DISAG	**Crew Awareness**
	ENG1 OUT	EAP 6-10
	ENG1 TLA FAIL	EAP 6-11
	ENG2 OUT	EAP 6-10
	ENG2 TLA FAIL	EAP 6-11
	ESS BUS 1 OFF	EAP 5-9
	ESS BUS 1-2 OFF	EAP 5-11
	ESS BUS 2 OFF	EAP 5-10
	FADEC ID NO DISP	**Crew Awareness**
	FLAP FAIL	EAP 8-7
	FUEL XFEED FAIL	EAP 9-6
	FUELING DOOR OPN	**Crew Awareness**
	FUEL EQ XFEED OPN	EAP 9-6
	FUEL IMBALANCE	EAP 9-7
	FUEL TANK LO TEMP	EAP 9-7
	GEN 1-2-3-4 OFF BUS APU GEN OFF BUS	EAP 5-4
	GEN 1 OFF BUS	EAP 5-12
	GEN 1 OVLD	EAP 5-12
	GEN 2 OFF BUS	EAP 5-12
	GEN 2 OVLD	EAP 5-12
	GEN 3 OFF BUS	EAP 5-12
	GEN 3 OVLD	EAP 5-12
	GEN 4 OFF BUS	EAP 5-12
	GEN 4 OVLD	EAP 5-12
	GPWS INOP	EAP 14-6
	HGS FAIL	**Crew Awareness**
	HS VLV 1 FAIL	EAP 1-8
	HS VLV 2 FAIL	EAP 1-8
	HYD SYS 1 FAIL	EAP 10-5
	HYD SYS 1 OVHT	EAP 10-7
	HYD SYS 1-2 FAIL	EAP 10-3
	HYD SYS 2 FAIL	EAP 10-6
	HYD SYS 2 OVHT	EAP 10-7
	IC 1 OVERHEAT	EAP 2-12
	IC 1 WOW INOP	**Crew Awareness**
	IC 2 OVERHEAT	EAP 2-12
	IC 2 WOW INOP	**Crew Awareness**
	IC BUS FAIL	EAP 2-11
	ICE DET1 FAIL	EAP 11-5
	ICE DET2 FAIL	EAP 11-5
	ICE DETECTORS FAIL	EAP 11-5
	IRS 1 ALN FAULT	EAP 2-12
	IRS 1 FAIL	EAP 2-14
	IRS 1 OVERHEAT	EAP 2-15

Copyright © by Embraer. Refer to cover page for details.

TYPE	MESSAGE	PAGE OR INSTRUCTION
CAUTION	IRS 2 ALN FAULT	EAP 2-12
	IRS 2 FAIL	EAP 2-14
	IRS 2 OVERHEAT	EAP 2-15
	LATERAL MODE OFF	EAP 2-11
	LG AIR/GND FAIL	EAP 12-6
	NLG UP/DOOR OPN	EAP 12-3
	NO ICE-A/ICE ON	EAP 11-6
	OXYGEN LO PRESS	EAP 13-3
	PACK 1 OVHT	EAP 1-9
	PACK 1 OVLD	EAP 1-10
	PACK 1 VLV FAIL	EAP 1-12
	PACK 2 OVHT	EAP 1-9
	PACK 2 OVLD	EAP 1-10
	PACK 2 VLV FAIL	EAP 1-12
	PAX OXYGEN LO PRESS	EAP 13-3
	PITOT 1 INOP	EAP 11-6
	PITOT 2 INOP	EAP 11-6
	PITOT 3 INOP	EAP 11-6
	PRESN AUTO FAIL	EAP 1-13
	PTRIM BKP SW FAIL	EAP 8-8
	PTRIM CPT SW FAIL	EAP 8-8
	PTRIM F/O SW FAIL	EAP 8-8
	RAM AIR VLV FAIL	EAP 1-14
	RUD HDOV PROTFAIL	EAP 8-8
	RUDDER OVERBOOST	EAP 8-9
	RUDDER SYS 1 INOP	EAP 8-10
	RUDDER SYS 1-2 INOP	EAP 8-10
	RUDDER SYS 2 INOP	EAP 8-10
	SHED BUS 1 OFF	**Crew Awareness**
	SHED BUS 2 OFF	**Crew Awareness**
	SPBK LVR DISAGREE	EAP 8-10
	SPOILER FAIL	EAP 8-3
	SPS ADVANCED	EAP 14-5
	STAB A/ICE FAIL	EAP 11-8
	STEER INOP	EAP 12-7
	STICK PUSHER FAIL	EAP 14-4
	TAT 1 HEAT INOP	EAP 11-7
	TAT 2 HEAT INOP	EAP 11-7
	TERR INOP	EAP 14-6
	VERTICAL MODE OFF	EAP 2-11
	W/S 1 HEAT FAIL	EAP 11-7
	W/S 2 HEAT FAIL	EAP 11-7
	WG A/ICE FAIL	EAP 11-8
	WINDSHEAR INOP	EAP 14-6
	YAW DAMPER FAIL	EAP 2-16

Copyright © by Embraer. Refer to cover page for details.

TYPE	MESSAGE	PAGE OR INSTRUCTION
ADVISORY	AHRS 1 ALN	**Crew Awareness**
	AHRS 1 ATT MODE	EAP 2-3
	AHRS 1 BASIC MODE	**Crew Awareness**
	AHRS 1 EXC MOTION	**Crew Awareness**
	AHRS 1 NO MAG HDG	**Crew Awareness**
	AHRS 1 NO PPOS	**Crew Awareness**
	AHRS 1 ON BATT	EAP 2-4
	AHRS 2 ALN	**Crew Awareness**
	AHRS 2 ATT MODE	EAP 2-3
	AHRS 2 BASIC MODE	**Crew Awareness**
	AHRS 2 EXC MOTION	**Crew Awareness**
	AHRS 2 NO MAG HDG	**Crew Awareness**
	AHRS 2 NO PPOS	**Crew Awareness**
	AHRS 2 ON BATT	EAP 2-4
	AIII NOT AVAIL	**Crew Awareness**
	APU FUEL SOV CLSD	**Crew Awareness**
	BLD 1 VLV CLSD	EAP 1-7
	BLD 2 VLV CLSD	EAP 1-7
	CHECK A PERF	**Crew Awareness**
	CHECK A1 PERF	**Crew Awareness**
	CHECK A1/3 PERF	**Crew Awareness**
	CHECK A1P PERF	**Crew Awareness**
	CHECK A3 PERF	**Crew Awareness**
	CMC FAIL	**Crew Awareness**
	CONFIG MISMATCH	**Crew Awareness**
	CROSS BLD OPEN	**Crew Awareness**
	DAU1 B FAIL	**Crew Awareness**
	DAU 1 REVERSION	**Crew Awareness**
	DAU2 B FAIL	**Crew Awareness**
	DAU 2 REVERSION	**Crew Awareness**
	DU 1 FAN FAIL	**Crew Awareness**
	DU 1 OVHT	**Crew Awareness**
	DU 2 FAN FAIL	**Crew Awareness**
	DU 2 OVHT	**Crew Awareness**
	DU 3 FAN FAIL	**Crew Awareness**
	DU 3 OVHT	**Crew Awareness**
	DU 4 FAN FAIL	**Crew Awareness**
	DU 4 OVHT	**Crew Awareness**
	DU 5 FAN FAIL	**Crew Awareness**
	DU 5 OVHT	**Crew Awareness**
	E1 FUEL IMP BYP	EAP 6-9
	E1 FUELSOV CLSD	**Crew Awareness**
	E1 HYD PUMP FAIL	**Crew Awareness**
	E1 HYDSOV CLSD	**Crew Awareness**

Copyright © by Embraer. Refer to cover page for details.

TYPE	MESSAGE	PAGE OR INSTRUCTION
ADVISORY	E1 IDL STP FAIL	EAP 6-9
	E1 OIL IMP BYP	**Crew Awareness**
	E1 SHORT DISP	**Crew Awareness**
	E1-2 FUEL IMP BYP	EAP 6-9
	E1-2 HYD PUMP FAIL	**Crew Awareness**
	E2 FUEL IMP BYP	EAP 6-9
	E2 FUELSOV CLSD	**Crew Awareness**
	E2 HYD PUMP FAIL	**Crew Awareness**
	E2 HYDSOV CLSD	**Crew Awareness**
	E2 IDL STP FAIL	EAP 6-9
	E2 OIL IMP BYP	**Crew Awareness**
	E2 SHORT DISP	**Crew Awareness**
	ENG A/ICEOVERPRES	**Crew Awareness**
	FDAU FAIL	**Crew Awareness**
	FLAP LOW SPEED	EAP 8-8
	FUEL XFEED OPEN	**Crew Awareness**
	GEN1 BRG FAIL	**Crew Awareness**
	GEN2 BRG FAIL	**Crew Awareness**
	GEN3 BRG FAIL	**Crew Awareness**
	GEN4 BRG FAIL	**Crew Awareness**
	HYD PUMP SELEC OFF	**Crew Awareness**
	HYD1 LO QTY	EAP 10-7
	HYD2 LO QTY	EAP 10-7
	IC1 CONFIG FAIL	**Crew Awareness**
	IC 1 FAN FAIL	**Crew Awareness**
	IC2 CONFIG FAIL	**Crew Awareness**
	IC 2 FAN FAIL	**Crew Awareness**
	ICE CONDITION	**Crew Awareness**
	IRS 1 ALN	EAP 2-12
	IRS 1 ATT MODE	EAP 2-13
	IRS 1 EXC MOTION	**Crew Awareness**
	IRS 1 NO MAG HDG	**Crew Awareness**
	IRS 1 NO PPOS	**Crew Awareness**
	IRS 1 ON BATT	EAP 2-14
	IRS 2 ALN	EAP 2-12
	IRS 2 ATT MODE	EAP 2-13
	IRS 2 EXC MOTION	**Crew Awareness**
	IRS 2 NO MAG HDG	**Crew Awareness**
	IRS 2 NO PPOS	**Crew Awareness**
	IRS 2 ON BATT	EAP 2-14
	PACK 1 VLV CLSD	EAP 1-11
	PACK 2 VLV CLSD	EAP 1-11
	RAD ALT 1 FAIL	EAP 2-16
	RAD ALT 2 FAIL	EAP 2-16
	RAD ALT FAIL	EAP 2-16
	SPS/ICE SPEEDS	**Crew Awareness**

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

BLEED APU LEAK ..... EAP 1-3
BLEED ENGINE LEAK ..... EAP 1-4
BLEED OVERTEMPERATURE ..... EAP 1-5
SMOKE / FIRE / FUMES ..... refer to S-6
APU BLEED VALVE FAILURE ..... EAP 1-5
BLEED LOW TEMPERATURE ..... EAP 1-6
BLEED VALVE CLOSED ..... EAP 1-7
BLEED VALVE FAILURE ..... EAP 1-7
CABIN DEPRESSURIZATION ..... EAP 1-13
CABIN RATE ABNORMAL FLUCTUATIONS ..... EAP 1-13
CROSSBLEED FAILURE ..... EAP 1-8
HIGH STAGE VALVE FAILURE ..... EAP 1-8
PACK OVERHEAT ..... EAP 1-9
PACK OVERLOAD ..... EAP 1-10
PACK VALVE CLOSED ..... EAP 1-11
PACK VALVE FAILURE ..... EAP 1-12
PRESSURIZATION AUTOMATIC SYSTEM FAILURE ..... EAP 1-13
RAM AIR VALVE FAILURE ..... EAP 1-14

NON ANNUNCIATED PROCEDURES

RAPID CABIN DEPRESSURIZATION ..... refer to NAP-11
LOSS OF PRESSURIZATION INDICATION ..... refer to NAP-27
SINGLE ENGINE BLEED OPERATION IN ICING CONDITIONS .. refer to NAP-34

Copyright © by Embraer. Refer to cover page for details.

LIST OF EICAS MESSAGES

BLD APU LEAK ..... EAP 1-3
BLD 1 (2) LEAK ..... EAP 1-4
BLD 1 (2) OVTEMP ..... EAP 1-5
APU BLD VLV FAIL ..... EAP 1-5
BLD 1 (2) LOW TEMP ..... EAP 1-6
BLD 1 (2) VLV CLSD ..... EAP 1-7
BLD 1 (2) VLV FAIL ..... EAP 1-7
CROSS BLD FAIL ..... EAP 1-8
HS VLV 1 (2) FAIL ..... EAP 1-8
PACK 1 (2) OVHT ..... EAP 1-9
PACK 1 (2) OVLD ..... EAP 1-10
PACK 1 (2) VLV CLSD ..... EAP 1-11
PACK 1 (2) VLV FAIL ..... EAP 1-12
PRESN AUTO FAIL ..... EAP 1-13
RAM AIR VLV FAIL ..... EAP 1-14

Copyright © by Embraer. Refer to cover page for details.

BLEED APU LEAK

EICAS Warning: BLD APU LEAK

APU Bleed ..... PUSH OUT
Bleed 1 and 2 ..... AS REQUIRED

..... WAIT 3 MINUTES

MESSAGE PERSISTS? **No**

Yes

APU ..... SHUTDOWN

MESSAGE STILL PERSISTS? **No**

Yes

BLEED ENGINE LEAK
Procedure (EAP 1-4) ..... ACCOMPLISH
Start the procedure considering BLD 1 LEAK.

END

END

Copyright © by Embraer. Refer to cover page for details.

BLEED ENGINE LEAK

EICAS Warning: BLD 1 (2) LEAK
Light: Red LEAK inscription in affected push button.

XBleed ..... CLOSED
Affected Bleed ..... PUSH OUT
APU Bleed ..... PUSH OUT
Altitude ..... MAX 25'000 FT, MIN MEA
Icing Conditions ..... EXIT/AVOID

..... WAIT 3 MINUTES

BLD 1 (2 OR APU) LEAK MESSAGE EXTINGUISHES? — **No**

Yes

END

ASSOCIATED MESSAGE? — **BLD 1 (2) VLV FAIL**

BLD 1 (2) VLV CLSD

Opposite Bleed ..... PUSH OUT
Associated Bleed ..... PUSH IN

..... WAIT 3 MINUTES

BLD 1 (2 OR APU) LEAK MESSAGE PERSISTS? — **No**

Yes

Associated Bleed ..... PUSH OUT
Oxygen Masks ..... AS REQUIRED
Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER

END

Associated Thrust Lever ..... IDLE

..... WAIT 3 MINUTES

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

BLD 1 (2) LEAK MESSAGE PERSISTS? **No**

Yes

Associated ENGINE FAILURE/SHUTDOWN Procedure (NAP-19)..... ACCOMPLISH

END

BLEED OVERTEMPERATURE

EICAS Warning: BLD 1 (2) OVTEMP

MFD Indication: Bleed Temp pointer may be in red range or out of view.

XBleed ..... OPEN
Affected Bleed ..... PUSH OUT
Altitude ..... MAX 25'000 FT, MIN MEA

SINGLE ENGINE BLEED OPERATION IN ICING CONDITIONS Procedure (NAP-34)..... AS REQUIRED

END

APU BLEED VALVE FAILURE

EICAS Caution: APU BLD VLV FAIL

FAILED IN THE OPEN POSITION? **No**

Yes

ANY BLEED LEAK EXISTS? **No**

Yes

APU ..... SHUTDOWN

Bleeds 1 and 2 ..... AS REQUIRED

END

Copyright © by Embraer. Refer to cover page for details.

BLEED LOW TEMPERATURE

EICAS Caution: BLD 1 (2) LOW TEMP

MFD Indication: Pointer may be amber or may be out of view.

BLEED TEMP POINTER OUT OF VIEW OR IN GREEN RANGE? — **No**

Yes

Message ..... DISREGARD

END

Ice Detection Override Knob ..... ALL

Associated Thrust Lever ..... ADVANCE

Adjust affected Thrust Lever until message disappears or pointer of the affected side reaches green range or matches the opposite pointer.

..... WAIT 1 MINUTE

MESSAGE PERSISTS? — **No**

Yes

Affected Bleed ..... PUSH OUT

Altitude ..... MAX 25'000 FT, MIN MEA

ICING CONDITIONS? — **No**

Yes

XBleed ..... OPEN

Altitude ..... MAX 15'000 FT, MIN MEA

If it is not possible to descend below 15'000 ft:

Ice Conditions ..... EXIT

..... WAIT 2 MINUTES

Ice Detection Override Knob ..... AUTO

END

Ice Detection Override Knob ..... AUTO

END

Copyright © by Embraer. Refer to cover page for details.

BLEED VALVE CLOSED

EICAS Advisory: BLD 1 (2) VLV CLSD

Associated Pack..... PUSH OUT, THEN PUSH IN
Associated Bleed ..... PUSH IN

MESSAGE PERSISTS? **No**

Yes

Affected BLEED VALVE FAILURE Procedure (EAP 1-7)..... AS REQUIRED

END

BLEED VALVE FAILURE

EICAS Caution: BLD 1 (2) VLV FAIL

BLD 1 (2) VLV CLSD MESSAGE PRESENTED? **No**

Yes

AFFECTED BLEED VALVE? **Bleed 2**

Bleed 1

APU SERVICEABLE? **No**

Yes

XBleed ..... CLOSED
APU..... START
APU Bleed..... PUSH IN
SINGLE ENGINE BLEED OPERATION IN ICING CONDITIONS Procedure (NAP-34)..... AS REQUIRED

END

XBleed ..... OPEN
Altitude..... MAX 25'000 FT, MIN MEA
SINGLE ENGINE BLEED OPERATION IN ICING CONDITIONS Procedure (NAP-34)..... AS REQUIRED

END

Copyright © by Embraer. Refer to cover page for details.

CROSSBLEED FAILURE

EICAS Caution: CROSS BLD FAIL

CROSS BLD OPEN MESSAGE PRESENTED?
- **No** → (skip to bleed actions)
- **Yes** → ANY BLEED LEAKAGE SUSPECTED OR ANNUNCIATED?
  - **No** → (skip to bleed actions)
  - **Yes** → ANY ENGINE OVERTEMPERATURE INDICATED?
    - **No** → (skip to bleed actions)
    - **Yes**

Bleeds 1 and 2 ..... PUSH OUT
APU Bleed ..... PUSH OUT
Icing Conditions ..... EXIT/AVOID
Oxygen Masks ..... AS REQUIRED
Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER

END

HIGH STAGE VALVE FAILURE

EICAS Caution: HS VLV 1 (2) FAIL

ASSOCIATED MFD BLEED POINTER INDICATES?
- **White Range** → (skip to Associated BLEED LOW TEMPERATURE)
- **Red Range**

Associated BLEED OVERTEMPERATURE Procedure (EAP 1-5) ..... ACCOMPLISH

END

Associated BLEED LOW TEMPERATURE Procedure (EAP 1-6) ..... ACCOMPLISH

END

Copyright © by Embraer. Refer to cover page for details.

PACK OVERHEAT

EICAS Caution: PACK 1 (2) OVHT

Associated Temperature & Mode Selector .....AUTO/FULL COLD
Affected Pack .....PUSH OUT
.....WAIT 3 MINUTES
Affected Pack .....PUSH IN

MESSAGE PERSISTS? **No**

Yes

Associated Temperature & Mode Selector .....MANUAL/FULL COLD
Affected Pack .....PUSH OUT
.....WAIT 3 MINUTES
Affected Pack .....PUSH IN

MESSAGE PERSISTS? **No**

Yes

Affected Pack .....PUSH OUT

BOTH PACKS AFFECTED? **No**

Yes

Oxygen Masks .....AS REQUIRED
Altitude.....MEA OR 10'000 FT, WHICHEVER IS HIGHER
At least one bleed source must be kept open.
When reaching 10'000 ft:
Cabin.....DEPRESSURIZE

END

Altitude.....MAX 25'000 FT, MIN MEA

END

END

Copyright © by Embraer. Refer to cover page for details.

PACK OVERLOAD

EICAS Caution: PACK 1 (2) OVLD

Recirculation Fan ..... PUSH IN
Associated Temperature & Mode Selector ..... AUTO / 12 O'CLOCK POSITION
Affected Pack ..... PUSH OUT
..... WAIT 3 MINUTES
Affected Pack ..... PUSH IN

MESSAGE PERSISTS? **No**

Yes

Associated Temperature & Mode Selector ..... MANUAL/ 12 O'CLOCK POSITION
Affected Pack ..... PUSH OUT
..... WAIT 3 MINUTES
Affected Pack ..... PUSH IN

MESSAGE STILL PERSISTS? **No**

Yes

Affected Pack ..... PUSH OUT
Associated Engine Bleed ..... CHECK PUSHED IN

BOTH PACKS AFFECTED? **No**

Yes

Oxygen Masks ..... AS REQUIRED
Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER
At least one bleed source must be kept open.
When reaching 10'000 ft:
Cabin ..... DEPRESSURIZE

END

Altitude ..... MAX 25'000 FT, MIN MEA

END

QRH-145/1115

Copyright © by Embraer. Refer to cover page for details.

PACK VALVE CLOSED

EICAS Advisory: PACK 1 (2) VLV CLSD

Associated Pack..... PUSH OUT, THEN PUSH IN

MESSAGE PERSISTS? **No**

Yes

BOTH PACKS AFFECTED? **No**

Yes

Oxygen Masks ..... AS REQUIRED
Altitude..... MEA OR 10’000 FT, WHICHEVER IS HIGHER

END

Altitude.....MAX 25’000 FT, MIN MEA

END

Copyright © by Embraer. Refer to cover page for details.

PACK VALVE FAILURE

EICAS Caution: PACK 1 (2) VLV FAIL

PACK 1 (2) VLV CLSD MESSAGE PRESENTED?

No

Yes

Altitude ..... MAX 25'000 FT, MIN MEA

BOTH PACKS AFFECTED?

No

Yes

Oxygen Masks ..... AS REQUIRED
Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER

When reaching 10'000 ft:
Cabin ..... DEPRESSURIZE

END

NEED TO CLOSE THE AFFECTED PACK?

No

Yes

Altitude ..... MAX 25'000 FT, MIN MEA

AFFECTED PACK(S)?

Pack 2

Pack 1 or Both Packs

APU Bleed ..... PUSH OUT

XBleed ..... CLOSED
Associated Bleed(s) ..... PUSH OUT
Icing Conditions ..... EXIT/AVOID

BOTH PACKS AFFECTED?

No

Yes

Oxygen Masks ..... AS REQUIRED
Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER

At least one bleed source must be kept open.
When reaching 10'000 ft:
Cabin ..... DEPRESSURIZE

END

QRH-145/1115

Copyright © by Embraer. Refer to cover page for details.

PRESSURIZATION AUTOMATIC SYSTEM FAILURE/CABIN DEPRESSURIZATION/ CABIN RATE ABNORMAL FLUCTUATIONS

EICAS Caution: PRESN AUTO FAIL may be presented.

EICAS Indication: Abnormal cabin altitude (amber or red) may be presented.
Erratic cabin rate fluctuations may be presented.

Pressurization Manual Controller.....11 O'CLOCK POSITION
Pressurization Mode Selector.....PUSH IN (MAN)
Pressurization Manual Controller.....AS REQUIRED
Wait for system response and stabilization.

MANUAL CONTROL POSSIBLE?

No

Yes

Cabin Altitude.....CONTROL
Set and control cabin altitude according to the AIRPLANE/CABIN ALTITUDE conversion table on NAP-27.

NOTE: On ground, select the manual controller to full up to avoid residual cabin pressure interfering on the door and cockpit slide windows operation.

END

CABIN AURAL WARNING VOICE MESSAGE SOUNDS?

No

Yes

RAPID CABIN DEPRESSURIZATION
Procedure (NAP-11).....ACCOMPLISH

END

Oxygen Masks.....AS REQUIRED
Altitude.....MEA OR 10’000 FT, WHICHEVER IS HIGHER

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

At least one bleed source must be kept open.
When reaching 10'000 ft:
Pressurization Manual Controller ...DOWN
Pressurization Mode Selector .....PUSH OUT
Pressurization Dump Button .....PUSH IN (ON)

END

RAM AIR VALVE FAILURE

EICAS Caution: RAM AIR VLV FAIL

Air Conditioning System .....MONITOR

ANY **PACK 1 (2) OVLD** OR **PACK 1 (2) OVHT** MESSAGE PRESENTED?

No

Yes

Associated Pack.....PUSH OUT
Altitude.....MAX 25'000 FT, MIN MEA

BOTH PACKS AFFECTED?

No

Yes

Oxygen Masks ..... AS REQUIRED
Altitude..... MEA OR 10'000 FT, WHICHEVER IS HIGHER
At least one bleed source must be kept open.
When reaching 10'000 ft:
Cabin..... DEPRESSURIZE

END

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

AUTOPILOT FAILURE.....EAP 2-3
AHRS ALIGNMENT FAULT.....EAP 2-3
AHRS ATTITUDE MODE.....EAP 2-3
AHRS FAIL.....EAP 2-4
AHRS ON BATTERY.....EAP 2-4
AHRS OVERHEAT.....EAP 2-4
AUTOPILOT AILERON MISTRIM.....EAP 2-5
AUTOPILOT ELEVATOR MISTRIM.....EAP 2-5
AUTOPILOT TRIM FAILED.....EAP 2-6
DAU FAILURE.....EAP 2-6
DAU MISCOMPARE.....EAP 2-7
DISPLAY FAILURE.....EAP 2-8
ELECTRONIC BAY OVERTEMPERATURE.....EAP 2-9
FD LATERAL MODE OFF.....EAP 2-11
FD VERTICAL MODE OFF.....EAP 2-11
IC BUS FAILURE.....EAP 2-11
IC FAILURE.....EAP 2-11
IC OVERHEAT.....EAP 2-12
IRS ALIGNMENT.....EAP 2-12
IRS ALIGNMENT FAULT.....EAP 2-12
IRS ATTITUDE MODE.....EAP 2-13
IRS FAIL.....EAP 2-14
IRS ON BATTERY.....EAP 2-14
IRS OVERHEAT.....EAP 2-15
RADIO ALTIMETER FAIL.....EAP 2-16
YAW DAMPER FAILURE.....EAP 2-16

NON ANNUNCIATED PROCEDURES

APPROACH WARNING.....refer to NAP-14
CAS MESSAGE MISCOMPARISON.....refer to NAP-14
IRS/MSU FAILURE ANNUNCIATION.....refer to NAP-25
NAV/FLIGHT INSTRUMENTS FAILURE.....refer to NAP-28

Copyright © by Embraer. Refer to cover page for details.

LIST OF EICAS MESSAGES

AUTOPILOT FAIL .....EAP 2-3
AHRS 1 (2) ALN FAULT.....EAP 2-3
AHRS 1 (2) FAIL .....EAP 2-4
AHRS 1 (2) OVERHEAT .....EAP 2-4
AP AIL MISTRIM.....EAP 2-5
AP ELEV MISTRIM .....EAP 2-5
AUTO TRIM FAIL.....EAP 2-6
CHECK PFD 1 (2).....EAP 2-8
DAU1 (2) A FAIL .....EAP 2-6
DAU1 (2) ENG MISCOMP .....EAP 2-7
DAU1 (2) SYS MISCOMP .....EAP 2-7
DAU1 (2) WRN MISCOMP.....EAP 2-7
ELEKBAY OVTEMP.....EAP 2-9
IC BUS FAIL.....EAP 2-11
IC 1 (2) OVERHEAT.....EAP 2-12
LATERAL MODE OFF .....EAP 2-11
IRS 1 (2) ALN FAULT .....EAP 2-12
IRS 1 (2) FAIL .....EAP 2-14
IRS 1 (2) OVERHEAT .....EAP 2-15
VERTICAL MODE OFF.....EAP 2-11
YAW DAMPER FAIL .....EAP 2-16
AHRS 1 (2) ATT MODE .....EAP 2-3
AHRS 1 (2) ON BATT .....EAP 2-4
DAU1 (2) B FAIL .....EAP 2-6
IRS 1 (2) ALN.....EAP 2-12
IRS 1 (2) ATT MODE .....EAP 2-13
IRS 1 (2) ON BATT .....EAP 2-14
RAD ALT 1 (2) FAIL .....EAP 2-16
RAD ALT FAIL .....EAP 2-16

Copyright © by Embraer. Refer to cover page for details.

AUTOPILOT FAILURE

EICAS Warning: AUTOPILOT FAIL
Aural Warning: **AUTOPILOT** Voice message

Autopilot .....DISENGAGE

END

AHRS ALIGNMENT FAULT

EICAS Caution: AHRS 1 (2) ALN FAULT

Check and reenter present position. If necessary, reenter present position once again.

END

AHRS ATTITUDE MODE

EICAS Advisory: AHRS 1 (2) ATT MODE

CROSS-SIDE AHRS AVAILABLE? **No**

Yes

Associated AHRS Button on Reversionary Panel.....PUSH IN

END

Maintain wings level and constant airspeed until AHRS 1 (2) ALN message is no longer displayed and attitude is recovered (approximately 20 seconds).

CAUTION:
- ATTITUDE OUTPUTS ARE NOT AS ACCURATE AS IN THE NORMAL OPERATIONAL MODE.
- AHRS MAGNETIC HEADING IS NOT AVAILABLE.

NOTE: The Autopilot is not available while AHRS 1 (2) ALN message is being displayed.

END

Copyright © by Embraer. Refer to cover page for details.

AHRS FAIL

EICAS Caution: AHRS 1 (2) FAIL

Relevant Inoperative Item: **Autopilot**

ONLY ONE AFFECTED?

No

Yes

Associated AHRS
on Reversionary Panel .....PUSH IN

END

Use standby instruments.

END

AHRS ON BATTERY

EICAS Advisory: AHRS 1 (2) ON BATT

Affected AHRS will operate for 40 minutes.

END

AHRS OVERHEAT

EICAS Caution: AHRS 1 (2) OVERHEAT

ASSOCIATED TO SMOKE OR FIRE?

No

Yes

Associated Circuit Breaker .....PULL
Backup Battery.....PUSH OUT
AHRS FAIL Procedure (EAP 2-4) .....ACCOMPLISH

AHRS data are valid for 30 minutes.

END

Copyright © by Embraer. Refer to cover page for details.

AUTOPILOT AILERON MISTRIM

EICAS Caution: AP AIL MISTRIM for more than 10 s.

Condition: Autopilot is engaged with aileron out of trim.

Control Wheel.....HOLD FIRMLY
Quick Disconnect Button .....PRESS
Roll Trim.....AS REQUIRED
Yaw Trim .....AS REQUIRED
Autopilot .....AS REQUIRED

END

AUTOPILOT ELEVATOR MISTRIM

EICAS Caution: AP ELEV MISTRIM

Condition: Autopilot is engaged with pitch out of trim.

Control Column .....HOLD FIRMLY
Quick Disconnect Button .....PRESS
Pitch Trim.....AS REQUIRED
Autopilot .....AS REQUIRED

END

Copyright © by Embraer. Refer to cover page for details.

AUTOPILOT TRIM FAILED

EICAS Caution: AUTO TRIM FAIL

Control Column .....HOLD FIRMLY
Quick Disconnect Button .....PRESS
Pitch Trim.....AS REQUIRED
Autopilot .....AS REQUIRED

END

DAU FAILURE

EICAS Caution: DAU1 (2) A FAIL
EICAS Advisory: DAU1 (2) B FAIL

ONLY CHANNEL **B** AFFECTED? — **No**

Yes

END

ONLY CHANNEL **A** AFFECTED? — **No**

Yes

Associated DAU on EICAS Rev (Pedestal Panel) .....PUSH IN

WHICH DAU IS AFFECTED? — **DAU 2**

DAU 1

- **Lost Indications:** engine 1 oil (temperature and pressure), battery 1 and 2 temperature, fuel tank temperature, roll trim position, cockpit temperature and bleed 1 temperature.
- **Lost Messages:** FUEL TANK LO TEMP, E1 FUEL LO TEMP, BLEED 1 OVTEMP.
- BLD 1 LOW TEMP message will appear.

END

- **Lost Indications:** engine 2 oil (temperature and pressure), hydraulic quantity 1 and 2, yaw trim position, cabin temperature and bleed 2 temperature.
- **Lost Messages:** BLEED 2 OVTEMP, E2 FUEL LO TEMP, HYD 1 LO QTY, HYD 2 LO QTY.
- BLD 2 LOW TEMP message will appear.
- APU OIL HI TEMP message will appear in case APU is OFF.

END

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

LAND AT THE NEAREST SUITABLE AIRPORT.

WHICH DAU IS AFFECTED?

DAU 2

DAU 1

- **All messages and indications of the following systems are lost:** Oxygen, Steering, Landing Gear, Roll Trim, and the message EMERG LIGHT NOT ARMED.
- **Some messages and indications of the following system are lost:** Doors, Stall Protection, Electrical, Fire Protection, Fuel, APU, Power Plant, Thrust Reverser, Spoiler, Brakes, Air Conditioning, Ice and Rain Protection.

END

- **All messages and indications of the following systems are lost:** Smoke, Hydraulic, Rudder and Yaw Trim.
- **Some messages and indications of the following systems are lost:** Doors, Stall Protection, Electrical, Fire Protection, Fuel, APU, Power Plant, Thrust Reverser, Flap, Brakes, Air Conditioning, Ice and Rain Protection.

END

DAU MISCOMPARE

EICAS Caution: DAU1 (2) ENG MISCOMP or
DAU1 (2) SYS MISCOMP or
DAU1 (2) WRN MISCOMP

Associated DAU on
EICAS Rev (Pedestal Panel) .....PUSH IN

Analyze the situation before and after the reversion, and take the appropriate corrective action.

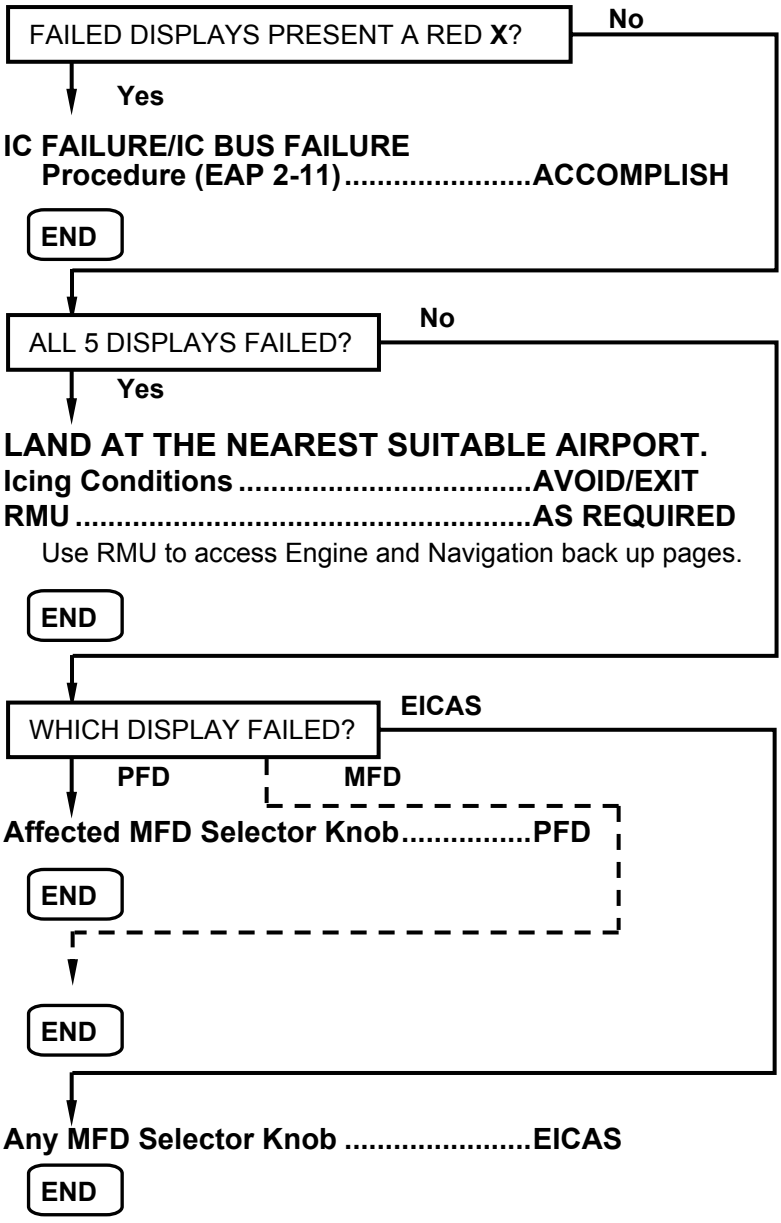
NOTE: For each miscompare message and each side, check the following parameters before and after the reversion:
- Engine: N1, N2, ITT.
- System: Battery voltage and temperature, Takeoff temperature, Hydraulic pressure, Oxygen pressure.
- Warning: all warning messages, if any.

END

Copyright © by Embraer. Refer to cover page for details.

DISPLAY FAILURE

EICAS Caution: CHECK PFD 1 (2) message is presented if PFD is the failed display.



Copyright © by Embraer. Refer to cover page for details.

ELECTRONIC BAY OVERTEMPERATURE

EICAS Caution: ELEKBAY OVTEMP

The following equipment is installed in the forward electronic compartment:

- Air Data Computer (ADC);
- Transponder Mode S;
- Integrated Communication Unit (ICU);
- Aural Warning Computer (AWC);
- Flight Management System (FMS);
- Attitude and Heading Reference System (AHRS);
- Passenger Address;
- Integrated Navigation Unit (INU);
- Inverters;
- Dimmers;
- Backup Battery.

It is recommended to turn off the systems that are unessential, using the table below to assess which system could be turned off. Turn off only systems that are unessential to the present phase of flight.

SYSTEM	POWER OFF CONTROL
Passenger Address	PA **CB** ..... PULL
Dimmers	Panel lights knob (pilot, pedestal and copilot) at left and right side of the glareshield panel ..... OFF OR PUSH BUTTONS **CB** ..... PULL
Integrated Navigation Unit	For INU 1: ADF 1 CB, DME 1 CB and VOR/ILS 1 **CB** ..... PULL OR For INU 2: ADF 2 CB, DME 2 CB and VOR/ILS 2 **CB** ..... PULL
Inverters	Push out AC PWR Push Button on overhead Electrical System panel. **NOTE**: TCAS and GPWS/Windshear may use 115V AC.

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

MESSAGE PERSISTS? **No**

Yes

It is recommended to turn off the redundant system and unessential equipment, using the table below to assess which system or equipment could be turned off. Turn off only systems and equipment that are unessential to the present phase of flight.

SYSTEM	POWER OFF CONTROL

FMS	- For Honeywell FMS:
	  - For FMS 1: CMPTR 1 CB.
	  - For FMS 2: CMPTR 2 CB.
	- For Universal FMS:
	  - FMS 1 CB.
	  - FMS 2 CB.
	NOTE: Some airplanes may not be equipped with dual FMS.

Aural Warning Computer	AWS CB.
Transponder Mode S	- For Transponder 1: XPDR 1 CB. - For Transponder 2: XPDR 2 CB.
Integrated Communication Unit	- For ICU 1: XPDR 1 CB and VHF 1 CB. - For ICU 2: XPDR 2 CB and VHF 2 CB.
Attitude and Reference System	- AHRS 1 CB. - AHRS 2 CB.
Air Data Computer	- ADC 1 CB. - ADC 2 CB.
Backup Battery	Backup Power Push Button on overhead Electrical System panel.

MESSAGE PERSISTS? **No**

Yes

LAND AT THE NEAREST SUITABLE AIRPORT.

Maintain a cross-check between main and standby instruments. In case of disagreement, follow the standby instruments indication.

END

Copyright © by Embraer. Refer to cover page for details.

FD LATERAL MODE OFF/ FD VERTICAL MODE OFF

EICAS Caution: LATERAL MODE OFF or VERTICAL MODE OFF

At crew discretion, re-select the affected Flight Director or select the other.

NOTE: For some EICAS versions this message will be presented if the crew turns the Flight Director off. In this case, the message must be disregarded.

END

IC FAILURE/IC BUS FAILURE

EICAS Caution: IC BUS FAIL may be presented.

Condition: Associated Display Units present a red X.

The following features will be inoperative:

- EICAS messages miscompare monitoring.
- Takeoff speeds synchronization.
- Flight Director mode synchronization.

FAILED DISPLAYS?

PFD 2 and MFD 2 (IC 2 Failed)

PFD 1, MFD 1 and EICAS (IC 1 Failed)

SG on Reversionary Panel 1 .....PUSH IN

NOTE: - The PIT TRIM 1 (2) INOP or PTRIM MAIN INOP and PTRIM BACKUP INOP messages may not be available, and

- The autopilot is not available.

END

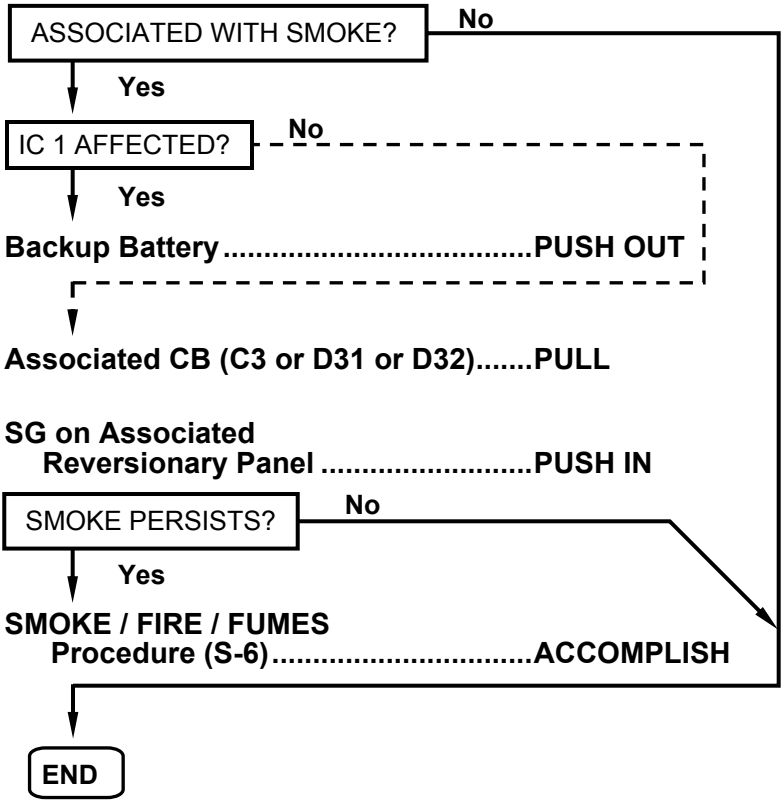
SG on Reversionary Panel 2 .....PUSH IN

END

Copyright © by Embraer. Refer to cover page for details.

IC OVERHEAT

EICAS Caution: IC 1 (2) OVERHEAT



IRS ALIGNMENT

EICAS Advisory: IRS 1 (2) ALN

IRU mode select switch .....CHECK NAV

This message is only presented during alignment phase or while the IRU mode select switch is set at ALIGN position.

END

IRS ALIGNMENT FAULT

EICAS Caution: IRS 1 (2) ALN FAULT

Check and reenter present position. If necessary, reenter present position once again.

END

Copyright © by Embraer. Refer to cover page for details.

IRS ATTITUDE MODE

EICAS Advisory: IRS 1 (2) ATT MODE

CONDITION? **In Flight**

On Ground

Affected IRU.....OFF

Wait MSU lights extinguish.

Affected IRU.....NAV

END

IRS Button on Associated Reversionary Panel .....PUSH IN

REMAINING IRU DATA NORMAL? **No**

Yes

END

Maintain wings level and constant airspeed for approximately 20 seconds until IRS 1 (2) ALN message is no longer displayed and attitude is recovered.

Magnetic Heading .....ENTER

CAUTION: FOR IRS IN ATTITUDE MODE, NAVIGATION AND ATTITUDE OUTPUTS ARE NOT AS ACCURATE AS IN THE NAV MODE. MAGNETIC HEADING MUST BE ENTERED AND UPDATED PERIODICALLY FROM THE BEST AVAILABLE ALTERNATIVE SOURCE, THROUGH THE FMS CDU.

Relevant Inoperative Item: **Autopilot**

END

Copyright © by Embraer. Refer to cover page for details.

IRS FAIL

EICAS Caution: IRS 1 (2) FAIL

OPERATION PHASE? **In Flight**

Power On or Alignment

Affected IRU.....OFF
Wait MSU lights extinguish.
Affected IRU.....NAV

END

IRS on Associated
Reversionary Panel .....PUSH IN

REMAINING IRU DATA NORMAL? **No**

Yes

END

Affected IRU.....ATT

Maintain wings level and constant airspeed for approximately 20 seconds until IRS 1 (2) ALN message is no longer displayed and attitude is recovered.

Magnetic Heading .....ENTER

CAUTION: FOR IRS IN ATTITUDE MODE, NAVIGATION AND ATTITUDE OUTPUTS ARE NOT AS ACCURATE AS IN THE NAV MODE. MAGNETIC HEADING MUST BE ENTERED AND UPDATED PERIODICALLY FROM THE BEST AVAILABLE ALTERNATIVE SOURCE, THROUGH THE FMS CDU.

Relevant Inoperative Item: **Autopilot**

END

IRS ON BATTERY

EICAS Advisory: IRS 1 (2) ON BATT

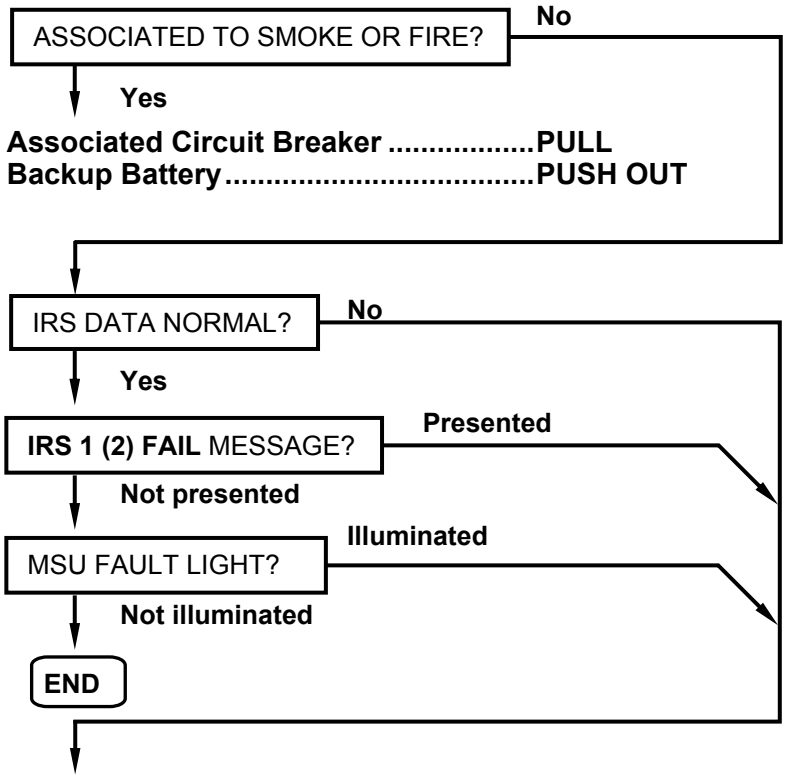
Associated IRU will operate for 40 minutes.

END

Copyright © by Embraer. Refer to cover page for details.

IRS OVERHEAT

EICAS Caution: IRS 1 (2) OVERHEAT



IRS on Associated
Reversionary Panel .....PUSH IN
Affected IRU.....OFF
Autopilot .....DISENGAGE

During final approach, if additional attitude reference is necessary:

Affected IRU .....ATT

For IRS in ATT mode, navigation and attitude outputs are not as accurate as in the NAV mode. Magnetic heading must be entered and updated periodically from the best available alternative source, through the FMS CDU.

Relevant Inoperative Item: **Autopilot**

END

Copyright © by Embraer. Refer to cover page for details.

RADIO ALTIMETER FAIL

EICAS Advisory: RAD ALT 1 (2) FAIL or
RAD ALT FAIL may be presented.

ASSOCIATED WITH ERRONEOUS EGPWS/GPWS ACTIVATION?

No

Yes

Consider pulling respective radio altimeter circuit breaker (D14 or D21) to inhibit undesired nuisance warnings.

ALL RADIO ALTIMETERS AFFECTED (**RAD ALT FAIL** MESSAGE PRESENTED)?

No

Yes

Do not perform CAT II or CAT III approaches.

END

RA1 (2) AMBER ANNUNCIATOR PRESENTED IN BOTH PFDs?

No

Yes

END

Do not perform CAT II or CAT III approaches.

END

YAW DAMPER FAILURE

EICAS Caution: YAW DAMPER FAIL

Yaw Damper.....DISENGAGE
Autopilot .....AS REQUIRED

FAILURE REOCCURS?

No

Yes

Autopilot .....DISENGAGE

END

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

APU FIRE ..... EAP 3-3
BLEED APU LEAK ..... refer to EAP 1-3
APU BLEED VALVE FAILURE ..... refer to EAP 1-5
APU CONTACTOR CLOSED ..... EAP 3-4
APU FAIL ..... EAP 3-4
APU FUEL LOW PRESSURE ..... EAP 3-4
APU FUEL SHUTOFF VALVE INOPERATIVE ..... refer to EAP 9-4
APU OIL LOW PRESSURE ..... EAP 3-5
APU OIL HIGH TEMPERATURE ..... EAP 3-5

NON ANNUNCIATED PROCEDURES

APU OVERTEMPERATURE ..... refer to NAP-4

LIST OF EICAS MESSAGES

APU FIRE ..... EAP 3-3
BLD APU LEAK ..... refer to EAP 1-3
APU BLD VLV FAIL ..... refer to EAP 1-5
APU CNTOR CLSD ..... EAP 3-4
APU FAIL ..... EAP 3-4
APU FUEL LO PRESS ..... EAP 3-4
APU FUEL SOV INOP ..... refer to EAP 9-4
APU OIL LO PRESS ..... EAP 3-5
APU OIL HI TEMP ..... EAP 3-5

APU FIRE

EICAS Warning: APU FIRE
Aural Warning: BELL

APU Fuel Shutoff Valve ..... PUSH IN

LAND AT THE NEAREST SUITABLE AIRPORT.

APU Master Knob ..... OFF

APU FUEL SOV CLSD MESSAGE?

Presented

Not Presented

XFeed ..... OFF
Fuel Pump Pwr 2 ..... OFF
Initiate the descent to MEA or 25'000 ft, whichever is higher.

..... WAIT 30 SECONDS

APU FIRE MESSAGE PERSISTS OR **APU FIREDET FAIL** IS DISPLAYED?

No

Yes

APU Fire Extinguishing Button ..... PRESS

WARNING: **DO NOT ATTEMPT TO RESTART APU.**

END

APU CONTACTOR CLOSED

EICAS Caution: APU CNTOR CLSD

Bus Ties ..... OFF
Battery 2 ..... OFF

END

APU FAIL

EICAS Caution: APU FAIL
Condition: APU automatically shuts down.

SHUTDOWN DURING APU START CYCLE? **No**

Yes

OBVIOUS SAFETY HAZARD? **No**

Yes

Do not try to restart the APU.

END

APU Start ..... AS REQUIRED

NOTE: - Up to three APU start attempts may be accomplished, including the first attempt in which the message was displayed.
- The APU Starter Cool-down Limit is 1 MINUTE OFF.

END

Do not try to restart the APU.

END

APU FUEL LOW PRESSURE

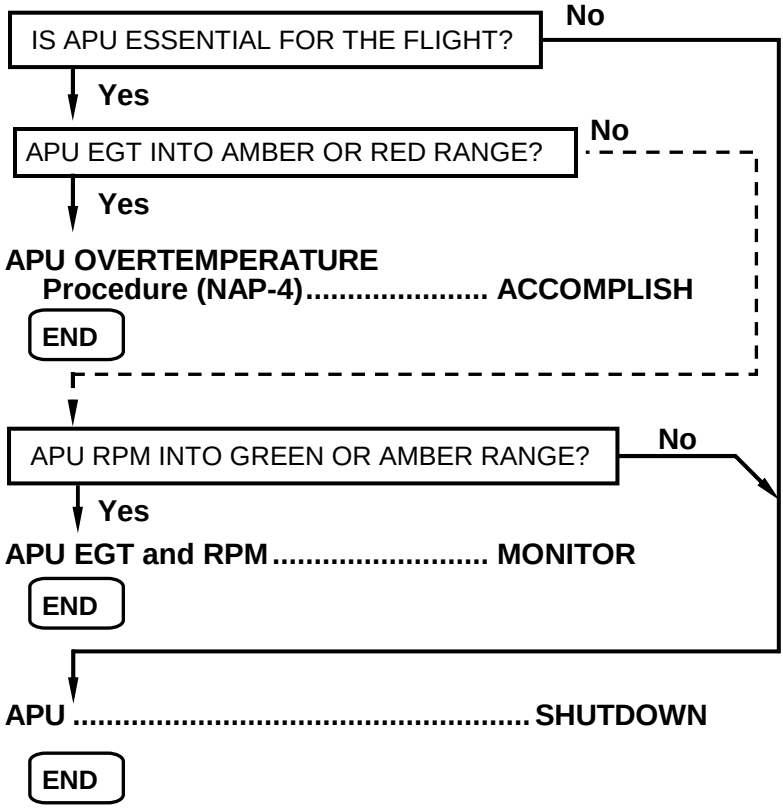
EICAS Caution: APU FUEL LO PRESS

Fuel Pump Sel 2 ..... SELECT ANOTHER
If the message persists, repeat the procedure.

END

APU OIL LOW PRESSURE/ APU OIL HIGH TEMPERATURE

EICAS Caution: APU OIL LO PRESS and/or
APU OIL HI TEMP



Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

MAIN DOOR OPEN ..... EAP 4-3
SERVICE DOOR OPEN..... EAP 4-3
ACCESS DOORS OPEN ..... EAP 4-4
BAGGAGE DOOR OPEN ..... EAP 4-4
EMERGENCY EXIT OPEN ..... EAP 4-5

NON ANNUNCIATED PROCEDURES

MAIN DOOR BLOCKED .....refer to NAP-28

LIST OF EICAS MESSAGES

MAIN DOOR OPN ..... EAP 4-3
SERVICE DOOR OPN..... EAP 4-3
ACCESS DOORS OPN ..... EAP 4-4
BAGGAGE DOOR OPN ..... EAP 4-4
EMERG EXIT OPN ..... EAP 4-5

MAIN OR SERVICE DOOR OPEN

EICAS Warning: MAIN DOOR OPN or SERVICE DOOR OPN

MFD Indication: Red DOOR OPEN

FSTN Belts ..... ON
Door Alignment Red Marks ..... CHECK

DOOR CONFIRMED — **No**

Yes

CABIN PRESSURIZATION NORMAL? — **No**

Yes

Cabin Pressure ..... MONITOR

END

LAND AT THE NEAREST SUITABLE AIRPORT.

Oxygen Masks ..... AS REQUIRED
Altitude ..... MEA OR 10’000 FT, WHICHEVER IS HIGHER

When reaching 10’000 ft:

Cabin ..... DEPRESSURIZE

END

ACCESS/BAGGAGE DOORS OPEN

EICAS Caution: ACCESS DOORS OPN or BAGGAGE DOOR OPN

MFD Indication: Red DOOR OPEN

Abrupt Maneuvers ..... AVOID

CABIN PRESSURIZATION NORMAL?

No

Yes

Message ..... DISREGARD

END

LAND AT THE NEAREST SUITABLE AIRPORT.

Oxygen Masks ..... AS REQUIRED

Altitude ..... MEA OR 10’000 FT, WHICHEVER IS HIGHER

When reaching 10’000 ft:

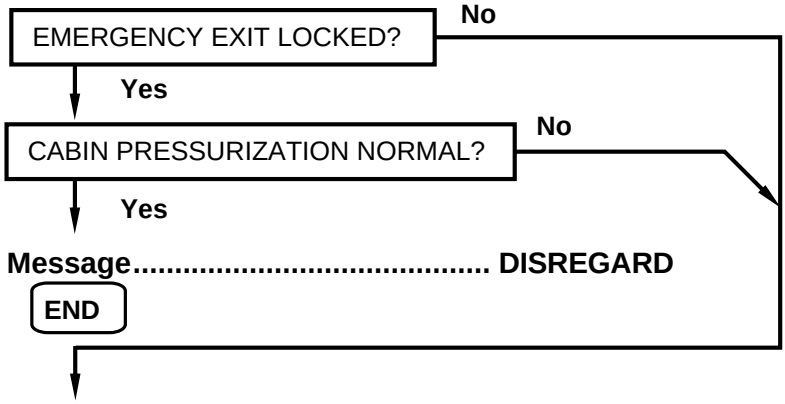
Cabin ..... DEPRESSURIZE

END

EMERGENCY EXIT OPEN

EICAS Caution: EMERG EXIT OPN
MFD Indication: Red DOOR OPEN

FSTN Belts ..... ON
Affected Exit ..... CHECK
Remove passenger(s) from exit vicinity.
Emergency exit handle ..... CHECK PUSHED IN



Message ..... DISREGARD

END

LAND AT THE NEAREST SUITABLE AIRPORT.

Oxygen Masks ..... AS REQUIRED
Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER

When reaching 10'000 ft:

Cabin ..... DEPRESSURIZE

END

INTENTIONALLY BLANK

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

BATTERY OVERTEMPERATURE ..... EAP 5-3
ELECTRIC ESSENTIAL TRANSFER FAILURE .. EAP 5-3
LOSS OF ALL GENERATORS ..... EAP 5-4
SMOKE / FIRE / FUMES ..... refer to S-6
115 V AC BUS OFF ..... EAP 5-5
APU CONTACTOR CLOSED ..... refer to EAP 3-4
BACK-UP BATTERY OFF BUS ..... EAP 5-5
BATTERY OFF BUS ..... EAP 5-5
DC BUS 1 OFF ..... EAP 5-6
DC BUS 2 OFF ..... EAP 5-7
ELECTRICAL EMERGENCY ABNORMAL TRANSFER ..... EAP 5-8
EMERGENCY LIGHTS NOT ARMED ..... EAP 5-8
ESSENTIAL BUS 1 OFF ..... EAP 5-9
ESSENTIAL BUS 2 OFF ..... EAP 5-10
ESSENTIAL BUS 1-2 OFF ..... EAP 5-11
GENERATOR OFF BUS ..... EAP 5-12
GENERATOR OVERLOAD ..... EAP 5-12

ANNEX 01 - AFFECTED EQUIPMENT

DC BUS FAILURE ..... EAP 5-13
ESS BUS FAILURE ..... EAP 5-14

Copyright © by Embraer. Refer to cover page for details.

LIST OF EICAS MESSAGES

BATT 1 (2) OVTEMP ..... EAP 5-3
ELEC ESS XFR FAIL ..... EAP 5-3
115 VAC BUS OFF ..... EAP 5-5
APU CNTOR CLSD ..... refer to EAP 3-4
APU GEN OFF BUS ..... EAP 5-12
APU GEN OVLD ..... EAP 5-12
BKUP BATT OFF BUS ..... EAP 5-5
BATT1 (2) OFF BUS ..... EAP 5-5
DC BUS 1 OFF ..... EAP 5-6
DC BUS 2 OFF ..... EAP 5-7
ELEC EMERG ABNORM ..... EAP 5-8
EMERG LT NOT ARMD ..... EAP 5-8
ESS BUS 1 OFF ..... EAP 5-9
ESS BUS 2 OFF ..... EAP 5-10
ESS BUS 1-2 OFF ..... EAP 5-11
GEN 1-2-3-4 OFF BUS ..... EAP 5-4
GEN 1 (2, 3, 4) OFF BUS ..... EAP 5-12
GEN 1 (2, 3, 4) OVLD ..... EAP 5-12

Copyright © by Embraer. Refer to cover page for details.

BATTERY OVERTEMPERATURE

EICAS Warning: BATT 1 (2) OVTEMP

MFD Indication: Battery temperature in red.

Affected Battery ..... OFF

ASSOCIATED **BATT 1 (2) OFF BUS** MESSAGE DISPLAYED?

No

Yes

END

LAND AT THE NEAREST SUITABLE AIRPORT.

END

ELECTRIC ESSENTIAL TRANSFER FAILURE

EICAS Warning: ELEC ESS XFR FAIL

If no generator is available:

Essential Power ..... PUSH IN

MESSAGE PERSISTS?

No

Yes

Bus Ties ..... OFF

MESSAGE STILL PERSISTS?

No

Yes

LAND AT THE NEAREST SUITABLE AIRPORT.

Shed Buses ..... OFF

Turn off non essential equipment. Begin with DC Buses 1 and 2 (CB Panel rows D, E, F, G and H).

LOSS OF ALL GENERATORS

Procedure (EAP 5-4) ..... ACCOMPLISH

END

Copyright © by Embraer. Refer to cover page for details.

LOSS OF ALL GENERATORS

EICAS Caution: GEN 1-2-3-4 OFF BUS,
APU GEN OFF BUS may be presented.

Condition: Noise increase due to nose landing gear doors open.

Generators..... PUSH OUT, THEN PUSH IN

APU ..... AS REQUIRED

Remember APU Maximum Start Altitude limitation.

LAND AT THE NEAREST SUITABLE AIRPORT.

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.95.

Airspeed ..... MAX 250 KIAS

Altitude ..... MEA OR 10'000 FT, WHICHEVER IS HIGHER

Essential Power ..... PUSH IN

Crew Oxygen..... AS REQUIRED

Passenger Oxygen..... AS REQUIRED

Emerg Lts ..... OFF

If required, turn on Emergency Lights before landing.

Icing Conditions..... EXIT/AVOID

Use standby instruments and RMU Navigation Backup Page.

CAUTION: BATTERY DURATION IS 40 MINUTES.

Do not set Thrust Levers below idle in flight.

Relevant Inoperative Items:

Autopilot	**W/S 1 and 2 Heating and Wiper**	**GPWS**
FMS 1 and 2	**Transponder 1 and 2**	**RA 1**
Speed Brake	**Main Pitch Trim**	**TCAS**
Pack 1 and 2	**Stick Pusher**	**Steering**
Spoilers	**ADF/DME/VHF/VOR/ILS/MB 2 and DME 1**	**Flaps**
Ventral Fuel Transfer Pump A and B (EMB-145XR only)		

Affected Equipment (EAP 5-13 and 5-14)..... CHECK

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

Landing configuration:

Landing Gear ..... DOWN

If necessary:

LG WRN Cutout ..... PRESS

FLAPS POSITION	MINIMUM AIRSPEED
0 to 8°	$V_{REF\ 45}$ + 30 KIAS
9° to 21°	$V_{REF\ 45}$ + 10 KIAS
22° to 44°	$V_{REF\ 45}$ + 5 KIAS
45°	$V_{REF\ 45}$

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.95.

Do not actuate Thrust Reversers.

END

115 V AC BUS OFF

EICAS Caution: 115 VAC BUS OFF

AC Power ..... PUSH OUT, THEN PUSH IN

MESSAGE PERSISTS? **No**

Yes

AC Power ..... PUSH OUT

Relevant Inoperative Items:

TCAS *	**Windshear Detection**	**GPWS**

* TCAS may be operative in some airplanes installations.

END

BACK-UP BATTERY OFF BUS

EICAS Caution: BKUP BATT OFF BUS

Backup Battery ..... CHECK PUSHED IN

END

BATTERY OFF BUS

EICAS Caution: BATT1 (2) OFF BUS

MFD Indication: Battery may be amber.

Affected Battery ..... AUTO

END

Copyright © by Embraer. Refer to cover page for details.

DC BUS 1 OFF

EICAS Caution: DC BUS 1 OFF
MFD Indication: DC BUS may be amber.

Bus Ties..... OVRD

MESSAGE PERSISTS? **No**

Yes

Bus Ties..... AUTO
Autopilot..... DISENGAGE
Icing Conditions..... EXIT/AVOID
Altitude ..... MAX 25'000 FT, MINIMUM MEA

At pilot's discretion:

MFD 1 Knob on Reversionary Panel..... PFD

Relevant Inoperative Items:

Ventral Fuel Transfer Pump A (EMB-145XR only)		
Autopilot	**W/S 1 Heating and Wiper**	**GPWS**
FMS 1	**Transponder 1**	**RA 1**
Speed Brake	**Main Pitch Trim**	**TCAS ***
Automatic Pressurization Control		**DME 1**
Pack 1		

* TCAS may be operative in some airplanes installations.

Do not set Thrust Lever 1 below idle in flight.

Affected Equipment (EAP 5-13)..... CHECK

Landing configuration:

Anticipate flap actuation.

Flaps ..... 22°

V_{REF} ..... $V_{REF\ 45°}$ + 10 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.50.

Do not actuate Thrust Reverser 1.

END

Copyright © by Embraer. Refer to cover page for details.

DC BUS 2 OFF

EICAS Caution: DC BUS 2 OFF

MFD Indication: DC BUS may be amber.

MFD Indication: Noise increase due to nose landing gear doors open.

Bus Ties ..... OVRD

MESSAGE PERSISTS? **No**

Yes

Bus Ties ..... AUTO
Icing Conditions ..... EXIT/AVOID
Airspeed ..... MAX 250 KIAS
Altitude ..... MAX 25'000 FT, MINIMUM MEA

The overhead panel lighting is inoperative, therefore, all striped bars will not illuminate.

SG on Reversionary Panel 2 ..... PUSH IN
At pilot's discretion:
MFD Knob on Reversionary Panel 2 ..... PFD
MFD Control is possible through MFD 1 Bezel.
Relevant Inoperative Items:

Stick Pusher	**W/S 2 Heating and Wiper**	**Steering**
FMS 2	**Transponder 2**	**Pack 2**
ADF/DME/VHF/VOR/ILS/MB 2		

Do not set Thrust Lever 2 below idle in flight.

Affected Equipment (EAP 5-13) ..... CHECK

Landing configuration:

Anticipate flap actuation.

Flaps ..... 22°

V_{REF} ..... $V_{REF\ 45°}$ + 10 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.50.

Do not actuate Thrust Reverser 2.

END

Copyright © by Embraer. Refer to cover page for details.

ELECTRICAL EMERGENCY ABNORMAL TRANSFER

EICAS Caution: ELEC EMERG ABNORM

Essential Power ..... CHECK PUSHED OUT

MESSAGE PERSISTS? **No**

Yes

LAND AT THE NEAREST SUITABLE AIRPORT.

Emerg Lts ..... OFF

If required, turn ON Emerg Lts just before landing.

Bus Ties..... OFF

Batteries 1 and 2 ..... OFF

MESSAGE STILL PERSISTS? **No**

Yes

Bus Ties..... AUTO

Batteries 1 and 2 ..... AUTO

APU (if serviceable) ..... START

APU Generator ..... CHECK PUSHED IN

CAUTION: IF APU GENERATOR IS NOT AVAILABLE AND THE MESSAGE REMAINS, BATTERIES DURATION WILL BE 40 MINUTES.

END

EMERGENCY LIGHTS NOT ARMED

EICAS Caution: EMERG LT NOT ARMD

EMERG LT Switch..... ARM

MESSAGE PERSISTS? **No**

Yes

EMERG LTS on Forward Attendant Control Panel..... Check NORM

END

Copyright © by Embraer. Refer to cover page for details.

ESSENTIAL BUS 1 OFF

EICAS Caution: ESS BUS 1 OFF

SG On Reversionary Panel 1 ..... PUSH IN
Fuel Pump 1 ..... 1B OR 1C
Fuel Pump 2 ..... 2A OR 2C
Altitude ..... MAX 25'000 FT, MIN MEA

ICING CONDITIONS? **No**

Yes

XBleed ..... OPEN
Altitude ..... MEA OR 15'000 FT, WHICHEVER IS HIGHER

If it is not possible to descend below 15'000 ft:
Icing Conditions ..... EXIT

Monitor fuel quantity indication 1 through FMS.

COM 2 on Digital Audio Panel 2 ..... PUSH IN
Relevant Inoperative Items:

ADF 1/VHF 1/VOR 1/ILS 1/MB 1	**Audio System 1**
ENG 1 Fire Detection System	**RMU 1**
Landing Gear Control (Down Override)	
Thrust Reverser 1	

Affected Equipment (EAP 5-14) ..... CHECK

EICAS DEENERGIZED? **No**

Yes

DAU 1 on EICAS Rev ..... PUSH IN
At pilots's discretion:
MFD Knob on
Reversionary Panel 1 ..... EICAS

Landing configuration:
Landing Gear ..... DOWN
Flaps ..... 45°
Airspeed ..... $V_{REF\ 45°}$
CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.45.
Brake effectiveness will be reduced.

END

Copyright © by Embraer. Refer to cover page for details.

ESSENTIAL BUS 2 OFF

EICAS Caution: ESS BUS 2 OFF
MFD Indication: ESS BUS may be amber.

Fuel Pump 1 .....1A OR 1C
Fuel Pump 2 .....2B OR 2C
Icing Conditions.....EXIT/AVOID
Altitude .....MAX 25'000 FT, MIN MEA

Monitor fuel quantity indication 2 through FMS.

CAUTION: DO NOT USE CROSSFEED.

Relevant Inoperative Items:

ISIS/Standby Altimeter (except for EMB-145XR Model)	
APU Fire Detection System	**Pitot Heating 3**
ENG 2 Fire Detection System	**Standby Attitude Indicator**
APU Control	**RMU 2**
Thrust Reverser 2	**Audio System 2**

Affected Equipment (EAP 5-14).....CHECK

When necessary to extend landing gear:

Landing Gear Lever.....DOWN
Gear Electrical Override.....DOORS
.....WAIT 3 SECONDS
Gear Electrical Override.....GEAR/DOORS

Landing configuration:

Landing Gear ..... DOWN
Flaps ..... 45°
Airspeed ..... $V_{REF\ 45°}$

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.45.

Brake effectiveness will be reduced.

END

Copyright © by Embraer. Refer to cover page for details.

ESSENTIAL BUS 1-2 OFF

EICAS Caution: ESS BUS 1-2 OFF

Bus Ties ..... OFF
MFD Electrical Page ..... CHECK

BOTH DC BUSES ENERGIZED?

No

Yes

Battery 1 and 2 ..... OFF

MESSAGE ESS BUS 1 OFF PERSISTS?

No

Yes

ESSENTIAL BUS 1 OFF
Procedure (EAP 5-9) ..... ACCOMPLISH

END

MESSAGE ESS BUS 2 OFF PERSISTS?

No

Yes

ESSENTIAL BUS 2 OFF
Procedure (EAP 5-10) ..... ACCOMPLISH

END

END

WHICH DC BUS IS ENERGIZED?

DC BUS 2

DC BUS 1

Battery 1 ..... OFF

Check which Essential Bus is off.
Associated ESSENTIAL BUS OFF
Procedure (EAP 5-9 or EAP 5-10) ... ACCOMPLISH

END

Battery 2 ..... OFF

Check which Essential Bus is off.
Associated ESSENTIAL BUS OFF
Procedure (EAP 5-9 or EAP 5-10) ... ACCOMPLISH

END

Copyright © by Embraer. Refer to cover page for details.

GENERATOR OFF BUS

EICAS Caution: GEN 1 (2, 3, 4) OFF BUS or APU GEN OFF BUS

MFD Indication: Generator voltage may be amber.

ALL GENERATORS OFF BUS? — **No**

Yes

LOSS OF ALL GENERATORS Procedure (EAP 5-4)..... ACCOMPLISH

END

Affected Generator ..... PUSH OUT, THEN PUSH IN

APU (if serviceable) ..... AS REQUIRED

END

GENERATOR OVERLOAD

EICAS Caution: GEN 1 (2, 3, 4) OVLD or APU GEN OVLD

MFD Indication: Generator may be amber.

Shed Buses ..... OFF

MESSAGE PERSISTS? — **No**

Yes

Electrical Load (affected generator).... REDUCE

The equipments that require more electrical load and may be turned OFF, at pilot's discretion, are: lights, hydraulic electric pump, ice protection and air conditioning related systems. Non-required equipment may also be turned OFF.

GEN 1 (2, 3, 4) OVLD EICAS MESSAGE DISPLAYED? — **No**

Yes

APU ON? — **No**

Yes

APU GEN ..... PUSH IN

END

APU ..... START

APU GEN ..... PUSH IN

END

Copyright © by Embraer. Refer to cover page for details.

ANNEX 01

In case of electrical bus failure, refer to the following table to verify the affected equipment.

DC BUS 1	DC BUS 2
AILERON CONTROL SYSTEM 1	ADC 2
AIR/GND POSITION SYSTEM A	ADF 2 *
AOA 1 SENSOR HEATING	AHRS 2 or IRS 2
AUTOMATIC PRESSURIZATION CONTROL	AILERON CONTROL SYSTEM 2
AUTOPILOT 1	AIR/GND POSITION SYSTEM C
BRAKES TEMP INDICATION OUTBD	AOA 2 SENSOR HEATING
CLEAR ICE DET - CHANNEL 1	AURAL WARNING SYSTEM 2
CMC	BAGGAGE SMOKE DETECTOR
CREW PEDAL ADJUSTMENT	BRAKES TEMP INDICATION INBD
CREW SEAT ADJUSTMENT 1	CABIN RECIRCULATION
DME 1	CLEAR ICE DET - CHANNEL 2
EICAS POWER (DAU 1B)	CLOCK COPILOT'S
ELECTRICAL FLIGHT IDLE STOP 1	CREW SEAT ADJUSTMENT 2
ELECTRONIC BAY COOLING (EXHAUST 1 AND RECIRC 2)	DEFUELING
EMER/PARKING BRAKE	DISPLAY PRCS/CTRL PWR 2 (IC 2)
ENG 1 FUEL PUMPS 1C	DME 2
ENG 1 THRUST REVERSER COMMAND	EICAS POWER (DAU 2B)
ENGINE 1 ANTI-ICE	ELECTRICAL FLIGHT IDLE STOP 2
FLAP CHANNEL 1	ELECTRONIC BAY COOLING (EXHAUST 2 AND RECIRC 1)
FMS SYSTEM 1 *	ENG 2 FUEL PUMPS 2C
FUEL PRESSURE REFUELING 1/2	ENG 2 THRUST REVERSER COMMAND
GROUND SPOILER OUTBD	ENGINE VIBRATION SENSORS
HEAD-UP GUIDANCE SYSTEM	ENGINE 2 ANTI-ICE
HYDR ELECTRIC PUMP 2	FLAP CHANNEL 2
HYDR GEN SYS 2 INDICATION	FMS SYSTEM 2 *
ICE DETECTOR 1	GASPER FAN
INVERTER	GPS *
LAVATORY FLUSH	GROUND SPOILER INBD
LAVATORY SMOKE DETECTOR	GUST LOCK (ELECTROMECHANICAL)
LAVATORY WATER DRAIN HEATER	HF POWER/CONTROL *
LIGHTING: CABIN 1, OVERHEAD PANEL, COCKPIT READING, COURTESY/STAIR 2, FLOOD/STORM, LAVATORY, LOGOTYPE	HYDR ELECTRIC PUMP 1
LIGHTS: LANDING 1 & NAVIGATION	HYDR GEN SYS 1 INDICATION
MAIN DOOR CONTROL 1	ICE DETECTOR 2
MFD 2 POWER	IRS 2
PACK VALVE 1	LANDING GEAR DOOR COMMAND
PASSENGER SIGNS	LIGHTING: OVERHEAD PANEL, COMPARTMENT, INSPECTION & PASSENGER CABIN 1/2/3
PFD 1 POWER	LIGHTS: RED BEACON & LANDING
PITCH TRIM MAIN	MFD 1 POWER
PITOT 1 HEATING	OBSERVER'S DAP (INTPH 3)
PNEUMATIC HSV 1	PACK VALVE 2
PUMP A (EMB-145XR)	PFD 2 POWER
RADAR SYSTEM	PITOT 2 HEATING
RADIO ALTIMETER 1	PNEUMATIC HSV 2
SPEED BRAKE	PUMP B (EMB-145XR)
STATIC PORT HEATING 1	RADIO ALTIMETER 2 *
STROBE LIGHTS	ROLL TRIM SYSTEM
TAT 1 SENSOR HEATING	SENSORS HEATING CONTROL
TCAS 2000	SPOILER INDICATION
TRANSPONDER 1	SPS (SHAKER 2/CHANNEL 2)
VHF SYSTEM 3 *	SPS PUSHER
WINDSHIELD HEATING 1	STABILIZER ANTI-ICE
WINDSHIELD WIPER 1	STATIC PORT HEATING 2
WING ANTI-ICE	STEERING
YAW TRIM	TAT 2 SENSOR HEATING
	TRANSPONDER 2
	TUNING BACKUP CONTROL HEAD
	VHF SYSTEM 2
	VOR 2/ILS 2/MB 2
	WINDSHIELD WIPER 2

Optional equipments are marked with an asterisk (*)

Copyright © by Embraer. Refer to cover page for details.

ANNEX 01

In case of electrical bus failure, refer to the following table to verify the affected equipment.

ESSENTIAL BUS 1	ESSENTIAL BUS 2
ADC 1	AIR/GND POSITION SYSTEM D
ADF 1	APU CONTROL
AHRS 1	APU FIRE DETECTION
AIR/GND POSITION SYSTEM B	APU FIRE EXTINGUISHING
APU BLEED	APU FUEL FEED
AURAL WARNING SYSTEM 1	BRAKES INBOARD
BRAKES OUTBOARD	CROSSBLEED
CLOCK PILOT'S	EICAS (DAU 2A)
DISPLAY PRCS/CTRL PWR 1 (IC 1)	ENG 1 FUEL PUMPS 1B
EICAS (DAU 1A)	ENG 2 FIRE DETECTION
EICAS DISPLAY	ENG 2 FUEL PUMPS 2A
ENG 1 FIRE DETECTION	ENGINE 2 STARTING
ENG 1 FUEL PUMPS 1A	ENGINES N2 SIGNALS 1B AND 2B
ENG 2 FUEL PUMPS 2B	ENGINES 1 AND 2 FADEC B
ENGINE 1 STARTING	FUEL CROSS FEED
ENGINES N2 SIGNALS 1A AND 2A	FUEL QUANTITY INDICATION 2
ENGINES 1 AND 2 FADEC A	ISIS (all models except for EMB-145XR)
FDR MANAGEMENT	LDG CONTROL
FUEL QUANTITY INDICATION 1	LDG NOSE INDICATION 2
IRS 1	LIGHTING EMERGENCY CTRL
LDG CONTROL (DOWN OVRD)	LIGHTING PANEL COPILOT'S AND PEDESTAL
LDG NOSE INDICATION 1	PASSENGER OXYGEN SYSTEM 2
LIGHTS COCKPIT DOME	PILOT/COPILOT'S DAP (INTPH 2)
PANEL LIGHTING PILOT'S	PITCH TRIM BACKUP
PASSENGER OXYGEN SYSTEM 1	PITOT HEATING 3
PILOT/COPILOT'S DAP (INTPH 1)	PNEUMATIC 2 (EBV 2)
PNEUMATIC 1 (EBV 1)	PUBLIC ADRESS
RAM AIR DISTRIBUTION	RMU 2
RMU 1	RUDDER CONTROL SYSTEM 1
RUDDER CONTROL SYSTEM 2	STANDBY ALTIMETER
SPS (SHAKER 1/CHANNEL 1)	STANDBY ATTITUDE INDICATOR
THRUST REVERSER 1	THRUST REVERSER 2
VHF SYSTEM 1	VOICE RECORDER
VOR/ILS/MB 1	

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

ATTCS FAILURE ..... EAP 6-3
BLEED ENGINE LEAK ..... refer to EAP 1-4
DUAL ENGINE FAILURE ..... EAP 6-3
ENGINE ATTCS NO MARGIN ..... EAP 6-5
ENGINE FIRE, SEVERE DAMAGE OR SEPARATION ..... EAP 6-6
ENGINE OIL LOW PRESSURE ..... EAP 6-7
ENGINE ANTI-ICING FAILURE ..... refer to EAP 11-5
ENGINE ATS SHUTOFF VALVE OPEN ..... EAP 6-8
ENGINE CONTROL FAILURE ..... EAP 6-9
ENGINE FUEL FILTER IMPENDING BYPASS ... EAP 6-9
ENGINE FUEL LOW PRESSURE ..... refer to EAP 9-5
ENGINE FUEL LOW TEMPERATURE .. refer to EAP 9-5
ENGINE FUEL SHUTOFF VALVE INOPERATIVE ..... refer to EAP 9-6
ENGINE IDLE STOP FAILURE ..... EAP 6-9
ENGINE OUT ..... EAP 6-10
ENGINE THRUST LEVER FAILURE ..... EAP 6-11
ENGINE THRUST REVERSER FAILURE/DISAGREE ..... EAP 6-12

NON ANNUNCIATED PROCEDURES

ABNORMAL ENGINE START ..... refer to NAP-12
ENGINE ABNORMAL VIBRATION ..... refer to NAP-15
ENGINE AIRSTART ..... refer to NAP-16
ENGINE FAILURE/SHUTDOWN ..... refer to NAP-19
ENGINE HIGH OIL PRESSURE ..... refer to NAP-20
ENGINE HIGH OIL TEMPERATURE ..... refer to NAP-20
ENGINE LOW OIL LEVEL ..... refer to NAP-20
ENGINE LOW OIL PRESSURE ..... refer to NAP-21
ENGINE OVERTEMPERATURE ..... refer to NAP-21
LOSS OF ENGINE INDICATIONS ..... refer to NAP-26
ONE ENGINE INOPERATIVE APPROACH AND LANDING ..... refer to NAP-30
SINGLE ENGINE BLEED OPERATION IN ICING CONDITIONS ..... refer to NAP-34

Copyright © by Embraer. Refer to cover page for details.

LIST OF EICAS MESSAGES

ATTCS FAIL ..... EAP 6-3
BLD 1 (2) LEAK ..... refer to EAP 1-4
E1 (2) ATTCS NO MRGN ..... EAP 6-5
E1 (2) OIL LOW PRESS ..... EAP 6-7
ENG 1 (2) FIRE ..... EAP 6-6
ENG 1-2 OUT ..... EAP 6-3
E1 (2) ATS SOV OPN ..... EAP 6-8
E1 (2) CTL FAIL ..... EAP 6-9
E1 (2) FUEL SOV INOP ..... refer to EAP 9-6

ENG1 (2) OUT ..... EAP 6-10
ENG1 (2) REV DISAGREE ..... EAP 6-12
ENG1 (2) REV FAIL ..... EAP 6-12
ENG1 (2) TLA FAIL ..... EAP 6-11
E1 (2) FUEL IMP BYP ..... EAP 6-9
E1 (2) IDL STP FAIL ..... EAP 6-9

Copyright © by Embraer. Refer to cover page for details.

ATTCS FAILURE

EICAS Warning: ATTCS FAIL

Thrust Levers ..... MAX

Another takeoff is not permitted.

END

DUAL ENGINE FAILURE

EICAS Warning: ENG 1-2 OUT may be presented.

Airspeed..... MIN 260 KIAS
Oxygen Mask..... AS REQUIRED

Altitude..... MAX 25'000 FT
Fuel Pumps Selectors 1 and 2 ..... CHECK A or B
Fuel Pumps Pwr 1 and 2..... CHECK ON

APU SERVICEABLE? — **No**

Yes

Thrust Levers ..... IDLE
Engine 1 and 2 Start/Stop Selectors... STOP
APU Bleed..... PUSH IN
Engine Bleeds 1 and 2 ..... PUSH OUT
Below 25'000 ft:
Engine 1 Start/Stop Selector ..... START, THEN RUN

ENGINE 1 STARTS? — **No**

Yes

Do not alternate FADEC 1.
ENGINE 2 AIRSTART Procedure (NAP-16)..... ACCOMPLISH

END

Engine 1 Start/Stop Selector ..... STOP
Engine 2 Start/Stop Selector ..... START, THEN RUN

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

ENGINE 2 STARTS?

No

Yes

Do not alternate FADEC 2.

ENGINE 1 AIRSTART
Procedure (NAP-16)..... AS REQUIRED

END

NOTE: Windmilling starts can be attempted in both engines simultaneously.

Airspeed ..... MIN 260 KIAS
Minimum N2 ..... 10%

Initiate windmilling start with N2 as high as possible.
Once N2 is below 10%, it may not be recovered.

Thrust Levers ..... IDLE
Engine 1 and 2 Start/Stop Selectors... STOP
Engine 1 and 2 Start/Stop Selectors... START, THEN RUN

ANY ENGINE STARTS?

No

Yes

INOPERATIVE ENGINE START CONSIDERED?

No

Yes

ENGINE AIRSTART (inoperative engine)
Procedure (NAP-16)..... AS REQUIRED

END

ENGINE FAILURE/SHUTDOWN
(inoperative engine)
Procedure (NAP-19)..... AS REQUIRED

END

DITCHING (NAP-5) OR FORCED
LANDING (NAP-7) Procedure..... ACCOMPLISH

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE ATTCS NO MARGIN

EICAS Warning: E1 (2) ATTCS NO MRGN

ENGINE MODEL?

A1P, A1/3 or A1E

A, A1, A1/1 or A3

Perform a Thrust Assurance Check, as described:

Associated Thrust Lever ..... MAX

ENGINE PARAMETERS IN NORMAL RANGE?

No

Yes

Another takeoff may be attempted. ALT T/O-1 mode is prohibited.

END

Do not takeoff.

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE FIRE, SEVERE DAMAGE OR SEPARATION

EICAS Warning: ENG 1 (2) FIRE (in case of fire) (may be presented)
Light: Engine Fire Handle (may be illuminated)
Aural Warning: BELL (in case of fire) (may sound)

Associated Thrust Lever ..... IDLE
Associated Start/Stop Selector ..... STOP
Associated Fire Extinguishing Handle ..... PULL (DO NOT ROTATE)

LAND AT THE NEAREST SUITABLE AIRPORT.

E1 (2) FUEL SOV CLSD MESSAGE? — **Displayed**

Not Displayed

XFeed Selector Knob ..... OFF
Associated Fuel Pumps ..... OFF

..... WAIT 30 SECS
Fire Extinguishing Handle (1st shot) ... ROTATE
..... WAIT 30 SECS

FIRE PERSISTS? — **No**

Yes

Fire Extinguishing Handle (2nd shot) .. ROTATE

CONDITION? — **On Ground**

In Flight

TCAS ..... TA ONLY
Affected Bleed ..... PUSH OUT

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

Remaining Engine Thrust Rating ..... CON
APU ..... START
APU Bleed ..... AS REQUIRED
XBleed ..... AS REQUIRED

ICING CONDITIONS? **No**

Yes

XBleed ..... OPEN
Altitude ..... MAX 15'000 FT, MIN MEA

If it is not possible to descend below 15'000 ft:

Icing Conditions ..... EXIT

For CAT III or CAT II approaches using HGS, the normal CAT III approach procedure must be used.

Approach:

Altimeters ..... SET AND CROSS CHECKED
Approach Aids ..... SET AND CROSS CHECKED
Speed Bugs ..... SET
Pressurization ..... CHECK
Go-Around Procedure ..... REVIEW

- Disengage Autopilot.
- Press Go-Around Button.
- Advance Operative Engine Thrust Lever to MAX.
- Rotate airplane to 10° nose up.
- Set flaps to 9°.

With positive rate of climb:

- Landing gear up.
- Maintain Approach Climb Speed until reaching acceleration altitude (level off).

Before Landing:

Inoperative Engine Thrust Lever .... IDLE
Landing Gear ..... DOWN
Thrust Rating ..... TAKEOFF MODE
Fuel XFeed ..... OFF
Autopilot/Yaw Damper ..... DISENGAGE

Landing configuration:

Flaps ..... 22°

V_{REF} ..... V_{REF45} + 10 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.48.

EMERGENCY EVACUATION

Procedure (NAP-6) ..... AS REQUIRED

END

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

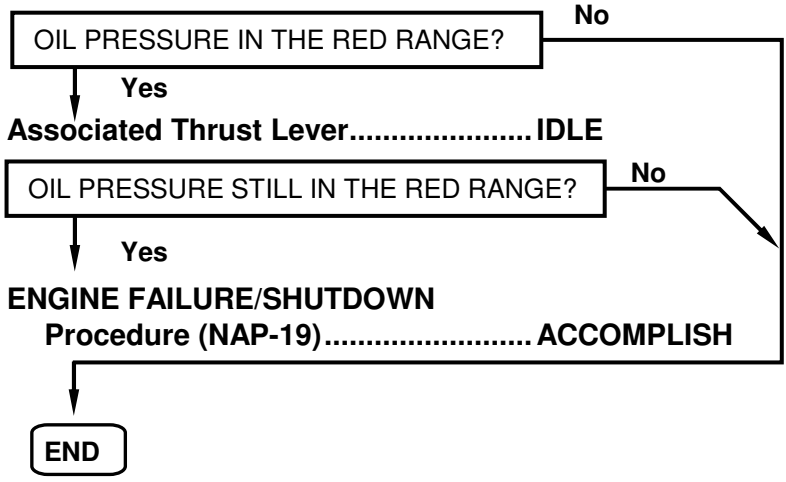
ENGINE OIL LOW PRESSURE

EICAS Warning: E1 (2) OIL LOW PRESS may be presented.

EICAS Indication: Oil pressure may be red.

Associated Thrust Lever..... REDUCE

Reduce Thrust Lever to at least N2 below 88%, until pressure is within limits.



Copyright © by Embraer. Refer to cover page for details.

ENGINE ATS SHUTOFF VALVE OPEN

EICAS Caution: E1 (2) ATS SOV OPN

Associated Bleeds
(including APU bleed) ..... PUSH OUT
XBleed ..... CLOSE

ON GROUND?

No

Yes

Thrust Levers ..... IDLE
Associated Start/Stop Selector ..... STOP

NOTE: If engine shutdown does not occur, pull the fire extinguishing handle.

END

Altitude ..... MAX 25'000 FT, MIN MEA

Icing Conditions ..... EXIT/AVOID

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE CONTROL FAILURE

EICAS Caution: E1 (2) CTL FAIL may be presented.

CAUTION: DO NOT MANUALLY ALTERNATE ASSOCIATED FADECS.

FADEC In Control ..... CHECK
Associated FADEC ..... RESET

FADEC IN CONTROL CHANGES? **No**

Yes

Avoid quick movements of the associated Thrust Lever.

NOTE: Thrust Lever movements may cause surge or an uncommanded engine shutdown.

Affected ENGINE FAILURE/SHUTDOWN Procedure (NAP-19) ..... AS REQUIRED

END

Engine control recovered.

END

ENGINE FUEL FILTER IMPENDING BYPASS

EICAS Advisory: E1 (2) FUEL IMP BYP

BOTH FILTERS AFFECTED? **No**

Yes

LAND AT NEAREST SUITABLE AIRPORT.

END

ENGINE IDLE STOP FAILURE

EICAS Advisory: E1 (2) IDL STP FAIL

Protection against thrust lever movement below flight idle is not available.

CAUTION: NEVER SET THRUST LEVER BELOW IDLE INFLIGHT.

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE OUT

EICAS Caution: ENG1 (2) OUT

Associated Thrust Lever ..... IDLE
Associated Start/Stop Selector..... STOP

NOTE: If engine shutdown does not occur, pull the associated fire extinguishing handle.

Engine Thrust Rating..... CON
APU (if available)..... START
APU Bleed..... AS REQUIRED
XBleed ..... AS REQUIRED
Fuel ..... BALANCE

ENGINE RESTART CONSIDERED? — **No**

Yes

Affected ENGINE AIRSTART
Procedure (NAP-16)..... ACCOMPLISH

END

LAND AT THE NEAREST SUITABLE AIRPORT.

TCAS..... TA ONLY
XBleed ..... OPEN
Altitude ..... MAX 25'000 FT, MIN MEA

ICING CONDITIONS? — **No**

Yes

Altitude ..... MEA OR 15'000 FT, WHICHEVER IS HIGHER

If it is not possible to descend below 15'000 ft:
Icing Conditions ..... EXIT

ONE ENGINE INOPERATIVE
APPROACH AND LANDING
Procedure (NAP-30)..... AS REQUIRED

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE THRUST LEVER FAILURE

EICAS Caution: ENG1 (2) TLA FAIL

Associated FADEC..... RESET

THRUST LEVER CONTROL RECOVERED?

No

Yes

END

Associated FADEC..... ALTN

THRUST LEVER CONTROL RECOVERED?

No

Yes

END

Thrust can be partially controlled through the Thrust Rating Buttons.

Affected ENGINE FAILURE/SHUTDOWN Procedure (NAP-19)..... AS REQUIRED

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE THRUST REVERSER FAILURE/DISAGREE

EICAS Caution: ENG1 (2) REV DISAGREE or
ENG1 (2) REV FAIL

CONDITION?

In Flight

On Ground

Do not takeoff.

END

ASSOCIATED N1 INDICATION BELOW THRUST LEVER SETTING?

No

Yes

LAND AT THE NEAREST SUITABLE AIRPORT.

Associated Thrust Lever ..... IDLE

Airspeed ..... MAX 200 KIAS

Associated ENGINE FAILURE/ SHUTDOWN Procedure (NAP-19) ... ACCOMPLISH

END

Message..... DISREGARD

END

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

BAGGAGE SMOKE ..... refer to S-3
ENGINE FIRE, SEVERE DAMAGE OR SEPARATION ..... refer to EAP 6-6
SMOKE / FIRE / FUMES..... refer to S-6
APU FIRE DETECTION FAILURE ..... EAP 7-3
APU FIRE EXTINGUISHING INOPERATIVE..... EAP 7-3
BAGGAGE COMPARTMENT FIRE EXTINGUISHING INOPERATIVE ..... EAP 7-4
ENGINE FIRE DETECTION FAILURE..... EAP 7-4
ENGINE FIRE EXTINGUISHING INOPERATIVE ..... EAP 7-5

Copyright © by Embraer. Refer to cover page for details.

LIST OF EICAS MESSAGES

BAGG SMOKE .....refer to S-3
ENG 1 (2) FIRE .....refer to EAP 6-6
APU FIREDET FAIL..... EAP 7-3
APU EXTBTL INOP ..... EAP 7-3
BAGG EXTBTL INOP..... EAP 7-4
E1 (2) FIREDET FAIL ..... EAP 7-4
E1 (2) EXTBTLA INOP ..... EAP 7-5
E1 (2) EXTBTLB INOP ..... EAP 7-5

Copyright © by Embraer. Refer to cover page for details.

APU FIRE DETECTION FAILURE

EICAS Caution: APU FIREDET FAIL

ASSOCIATED TO APU FAILURE?

No

Yes

APU FIRE Procedure (EAP 3-3).....ACCOMPLISH

END

ASSOCIATED TO SUSPECTED APU FIRE OR OVERHEAT?

No

Yes

APU FIRE Procedure (EAP 3-3).....ACCOMPLISH

END

The APU fire detection system is inoperative.

END

APU FIRE EXTINGUISHING INOPERATIVE

EICAS Caution: APU EXTBTL INOP
Condition: Affected bottle has not been discharged intentionally.

APU fire protection is not available.
Consider shutting the APU down.

END

Copyright © by Embraer. Refer to cover page for details.

BAGGAGE COMPARTMENT FIRE EXTINGUISHING INOPERATIVE

EICAS Caution: BAGG EXTBTL INOP (if installed).
Condition: Affected bottle has not been discharged intentionally.

BAGG SMOKE MESSAGE DISPLAYED? — **No**

Yes

LAND AT THE NEAREST SUITABLE AIRPORT.
BAGGAGE SMOKE Procedure (S-4)....AS REQUIRED

END

Baggage fire protection is not available.

END

ENGINE FIRE DETECTION FAILURE

EICAS Caution: E1 (2) FIREDET FAIL

ASSOCIATED TO ENGINE FAILURE? — **No**

Yes

Affected ENGINE FIRE, SEVERE DAMAGE OR SEPARATION Procedure (EAP 6-6).....ACCOMPLISH

END

ASSOCIATED TO SUSPECTED ENGINE FIRE OR OVERHEAT? — **No**

Yes

Affected ENGINE FIRE, SEVERE DAMAGE OR SEPARATION Procedure (EAP 6-6).....ACCOMPLISH

END

The associated engine fire detection system is inoperative.

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE FIRE EXTINGUISHING INOPERATIVE

EICAS Caution: E1 (2) EXTBTLA INOP or
E1 (2) EXTBTLB INOP

Condition: Affected bottle has not been discharged intentionally.

Only one bottle is available to protect both engines against fire.

END

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

INADVERTENT SPOILER OPEN ..... EAP 8-3
PITCH TRIM INOPERATIVE..... EAP 8-4
AILERON SYSTEM INOPERATIVE..... EAP 8-6
FLAP FAILURE..... EAP 8-7
FLAP LOW ACTUATION SPEED ..... EAP 8-8
PITCH TRIM SWITCH INOPERATIVE..... EAP 8-8
RUDDER HARDOVER PROTECTION FAILURE ..... EAP 8-8
RUDDER OVERBOOST ..... EAP 8-9
RUDDER SYSTEM INOPERATIVE ..... EAP 8-10
SPEED BRAKE LEVER DISAGREE..... EAP 8-11

NON ANNUNCIATED PROCEDURES

AILERON RUNAWAY ..... refer to NAP-3
JAMMED AILERON ..... refer to NAP-8A
JAMMED ELEVATOR..... refer to NAP-8A
JAMMED RUDDER..... refer to NAP-9
PITCH TRIM RUNAWAY ..... refer to NAP-10
ROLL TRIM RUNAWAY..... refer to NAP-3
AILERON ARTIFICIAL FEEL INOPERATIVE ..... refer to NAP-14
ASYMMETRIC RUDDER OPERATION . refer to NAP-14
RUDDER ARTIFICIAL FEEL INOPERATIVE ..... refer to NAP-33
RUDDER RUNAWAY ..... refer to NAP-33
STIFFENED ELEVATOR ..... refer to NAP-35
UNCOMMANDED AILERON DISCONNECTION ..... refer to NAP-36
UNCOMMANDED ELEVATOR DISCONNECTION ..... refer to NAP-36
YAW TRIM RUNAWAY..... refer to NAP-33

Copyright © by Embraer. Refer to cover page for details.

LIST OF EICAS MESSAGES

SPOILER FAIL ..... EAP 8-3
PIT TRIM 1 (2) INOP ..... EAP 8-4
PTRIM MAIN INOP ..... EAP 8-4
PTRIM BACKUP INOP ..... EAP 8-4
AIL SYS 1 (2) INOP ..... EAP 8-6
FLAP FAIL ..... EAP 8-7
PTRIM CPT SW FAIL ..... EAP 8-8
PTRIM F/O SW FAIL ..... EAP 8-8
PTRIM BKP SW FAIL ..... EAP 8-8
RUD HDOV PROTFAIL ..... EAP 8-8
RUDDER OVERBOOST ..... EAP 8-9
RUDDER SYS 1 (2) INOP ..... EAP 8-10
SPBK LVR DISAGREE ..... EAP 8-11
FLAP LOW SPEED ..... EAP 8-8

Copyright © by Embraer. Refer to cover page for details.

INADVERTENT SPOILER OPEN

EICAS Caution: SPOILER FAIL (may be presented)

Condition: Sudden airspeed or altitude loss, buffeting or roll tendency.

EICAS Indication: SPLRS OPN

Speed Brake ..... CLOSE

Spoilers CB's F13, F14 and F21 ..... PULL

Do not reduce Thrust during flare.

Landing Configuration:

Flaps ..... 22°

V_{REF45} ..... V_{REF45} + 10 KIAS

PANEL FAILURE MODE?

Jammed Open

Jammed Closed or Floating

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.50.

END

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.45.

END

Copyright © by Embraer. Refer to cover page for details.

PITCH TRIM INOPERATIVE

EICAS Warning: PTRIM MAIN INOP (may be presented) and/or PTRIM BACKUP INOP (may be presented) or PIT TRIM 1 (2) INOP (may be presented).

EICAS Caution: AUTO TRIM FAIL (may be presented).

EMB-145XR MODEL? **No**

Yes

PTRIM BACKUP INOP MESSAGE DISPLAYED? **No**

Yes

Airspeed ..... MAX 0.78 MACH

ANY EICAS MESSAGE DISPLAYED? **No**

Yes

MESSAGE DISPLAYED WITHOUT PRESSING ANY PITCH TRIM SWITCH? **No**

Yes

INOPERATIVE SYSTEM(S)? **Both**

Main (1)

Backup (2)

Pitch Trim Main Sys Cutout ..... PUSH OUT

Autopilot ..... DISENGAGE

Back up system can be used to trim the airplane.

END

Pitch Trim Backup Sys Cutout..... PUSH OUT

Main system can be used to trim the airplane.
Autopilot is available.

END

Pitch Trim Main Sys Cutout ..... PUSH OUT

Pitch Trim Backup Sys Cutout..... PUSH OUT

Consider LANDING AT THE NEAREST SUITABLE AIRPORT.

Autopilot ..... DISENGAGE

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

AIRPLANE PITCH UP TENDENCY? **No**

Yes

ANY PITCH TRIM SWITCH (PILOT, COPILOT OR BACKUP) WORKING? **No**

Yes

Use the remaining Pitch Trim System.
If Main Pitch Trim System is inoperative, do not engage the Autopilot.

END

WARNING: **DO NOT OPEN SPEEDBRAKE.**

Airspeed..... REDUCE

Reduce airspeed to alleviate forward control column forces, observing Flap Maneuvering Speed (PD-2). Continuous turns also helps to alleviate forward control column forces. Extending flaps and landing gear helps to recover trimmed condition.

Landing Configuration:

Landing Gear ..... DOWN
Flaps..... 22°
V_{REF} ..... V_{REF45} + 10 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.45.

END

WARNING: **DO NOT OPEN SPEEDBRAKE.**

AIRPLANE PITCH DOWN TENDENCY? **No**

Yes

Airspeed..... REDUCE
Below 250 KIAS ..... FLAPS 9° (MAX 20'000 FT)
Below 200 KIAS ..... FLAPS 22°

Landing Configuration:

Landing Gear ..... DOWN
Delay gear extension as long as possible.
Flaps..... 22°
V_{REF} ..... V_{REF45} + 25 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.75.

END

Copyright © by Embraer. Refer to cover page for details.

AILERON SYSTEM INOPERATIVE

EICAS Caution: AIL SYS 1 (2) INOP

Affected Aileron Shutoff ..... PUSH OUT
Airspeed ..... MAX 250 KIAS

BOTH SYSTEMS FAILED?

No

Yes

Remaining Aileron Shutoff ..... PUSH OUT
Autopilot ..... DISENGAGE

Expect greater aileron control force. If required, both pilots should act together to control airplane.

Avoid landing at airports with anticipated turbulence, gusts or crosswind.

Perform a long final approach.

Landing Configuration:

Flaps ..... 22°
V_{REF} ..... V_{REF45} + 30 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.85.

END

Copyright © by Embraer. Refer to cover page for details.

FLAP FAILURE

EICAS Caution: FLAP FAIL
Condition: Flap operation is not possible.
EICAS Indication: Flap position may become amber.

If flap indication on EICAS is not available, use the RMU Engine Backup Page 2 or flap position marks on the wing.

With flaps at intermediate positions, limit airspeed according to the following:

ALL MODELS EXCEPT EMB-145XR

FLAPS POSITION	MAX AIRSPEED
1° to 9°	250 KIAS
10° to 22°	200 KIAS
23° to 45°	145 KIAS

EMB-145XR MODEL

FLAPS POSITION	MAXIMUM AIRSPEED	ABOVE 10'000 ft AND Y/D DISENGAGED
1° to 9°	250 KIAS	250 KIAS
10° to 22°	180 KIAS	180 KIAS
23° to 45°	160 KIAS	145 KIAS

V_{REF}:

FLAPS POSITION	V_{REF}
0 to 8°	V_{REF45} + 30 KIAS
9° to 21°	V_{REF45} + 10 KIAS
22° to 44°	V_{REF45} + 5 KIAS
45°	V_{REF45}

At crew discretion:

EGPWS/GPWS CB's (J7 or J8) ..... PULL

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY:

FLAPS POSITION	FACTOR
0 to 8°	1.65
9° to 21°	1.40
22° to 44°	1.40

END

Copyright © by Embraer. Refer to cover page for details.

FLAP LOW ACTUATION SPEED

EICAS Advisory: FLAP LOW SPEED

Anticipate flap actuation.

END

PITCH TRIM SWITCH INOPERATIVE

EICAS Caution: PTRIM CPT SW FAIL,
PTRIM F/O SW FAIL or
PTRIM BKP SW FAIL

CONDITION?

On Ground

In Flight

Use another serviceable switch.

END

Deenergize the airplane and energize it again.

END

RUDDER HARDOVER PROTECTION FAILURE

EICAS Caution: RUD HDOV PROTFAIL

Rudder hardover protection is not available.

END

Copyright © by Embraer. Refer to cover page for details.

RUDDER OVERBOOST

EICAS Caution: RUDDER OVERBOOST

Rudder Shutoff 2 .....PUSH OUT

RUDDER OVERBOOST MESSAGE PERSISTS? — **No**

Yes

Rudder Shutoff 2 .....PUSH IN
Rudder Shutoff 1 .....PUSH OUT
Below 135 KIAS:
Rudder Shutoff 1 .....PUSH IN

END

Below 135 KIAS:
Rudder Shutoff 2 .....PUSH IN

END

Copyright © by Embraer. Refer to cover page for details.

RUDDER SYSTEM INOPERATIVE

EICAS Caution: RUDDER SYS 1 (2) INOP or
RUDDER SYS 1-2 INOP

FOLLOWING A RUDDER RUNAWAY? **No**

Yes

RUDDER/YAW TRIM RUNAWAY
Procedure (NAP-33).....ACCOMPLISH

END

INOPERATIVE SYSTEM? **Both (1 and 2)**

Only one (1 or 2)

Affected Rudder Shutoff.....PUSH OUT

END

Rudder Shutoff 2 .....PUSH OUT

RUDDER 1-2 INOP MESSAGE PERSISTS? **No**

Yes

Rudder Shutoff 1 .....PUSH OUT
Rudder Shutoffs 1 and 2.....PUSH IN

RUDDER 1-2 INOP MESSAGE PERSISTS? **No**

Yes

Expect heavier than normal rudder pedals. If required, combine effort of both pilots.

Consider use of aileron and asymmetric thrust to control the airplane.

Avoid airports with reported gusts, turbulence or crosswind.

END

Copyright © by Embraer. Refer to cover page for details.

SPEED BRAKE LEVER DISAGREE

EICAS Caution: SPBK LVR DISAGREE

Speed Brake Lever.....CLOSE

END

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

FUEL LOW LEVEL ..... EAP 9-3
FUEL TRANSFER CRITICAL ..... EAP 9-3
APU FUEL LOW PRESSURE ..... refer to EAP 3-4
APU FUEL SHUTOFF VALVE INOPERATIVE ..... EAP 9-4
DEFUEL VALVE NOT CLOSED ..... EAP 9-4
ENGINE FUEL FILTER IMPENDING BYPASS ..... refer to EAP 6-9
ENGINE FUEL LOW PRESSURE ..... EAP 9-5
ENGINE FUEL LOW TEMPERATURE ..... EAP 9-5
ENGINE FUEL SHUTOFF VALVE INOPERATIVE ..... EAP 9-6
FUEL CROSSFEED FAILURE ..... EAP 9-6
FUEL CROSSFEED MISCOMMAND ..... EAP 9-6
FUEL IMBALANCE ..... EAP 9-7
FUEL TANK LOW TEMPERATURE ..... EAP 9-7
FUEL TRANSFER ISOLATION FAILURE EAP-9-8
FUEL TRANSFER OVERRIDE EAP-9-8
FUEL TRANSFER SYSTEM INOPERATIVE EAP-9-9
VENTRAL TANK VENTILATION OPEN EAP-9-10
WING TANKS OVERFLOW EAP-9-10

LIST OF EICAS MESSAGES

FUEL 1 (2) LO LEVEL ..... EAP 9-3
FUEL XFER CRITICAL ..... EAP 9-3
APU FUEL LO PRESS ..... refer to EAP 3-4
APU FUEL SOV INOP ..... EAP 9-4
DEFUEL NOT CLOSED ..... EAP 9-4
E1 (2) FUEL LO PRESS ..... EAP 9-5
E1 (2) FUEL LO TEMP ..... EAP 9-5
E1 (2) FUEL SOV INOP ..... EAP 9-6
FUEL EQ XFEED OPN ..... EAP 9-6
FUEL IMBALANCE ..... EAP 9-7
FUEL TANK LO TEMP ..... EAP 9-7
FUEL VENT OPEN ..... EAP 9-10
FUEL XFEED FAIL ..... EAP 9-6
FUEL XFER OVRD ..... EAP 9-8
FUEL XFER INOP ..... EAP 9-9
FUEL XFER OVERFLOW ..... EAP 9-10
XFER ISOL FAIL ..... EAP 9-8

FUEL LOW LEVEL

EICAS Warning: FUEL 1 (2) LO LEVEL
MFD Indication: Fuel quantity in red range.

LAND AT THE NEAREST SUITABLE AIRPORT.

Thrust Levers .....LONG RANGE CRUISE

Avoid attitudes in excess of 10° nose down or 12° nose up attitude, uncoordinated maneuvers and negative g's.

Xfeed Operation .....AS REQUIRED

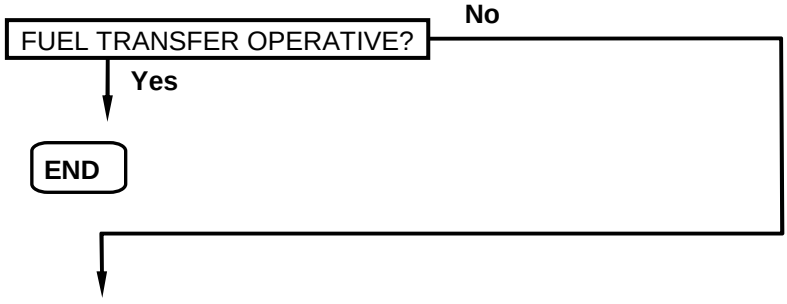
END

FUEL TRANSFER CRITICAL

EICAS Warning: FUEL XFER CRITICAL

Thrust Levers ..... LONG RANGE CRUISE

Fuel Transfer System..... CHECK STATUS



Consider diversion.

Remaining Fuel ..... CHECK

Avoid rapid maneuvers and flying in severe turbulence conditions.

Before touchdown:

Rate of Descent ..... MAX 450 FT/MIN

Touch smoothly the runway surface.

Reduce engine thrust only after touchdown.

END

Copyright © by Embraer. Refer to cover page for details.

APU FUEL SHUTOFF VALVE INOPERATIVE

EICAS Caution: APU FUEL SOV INOP

APU Fuel Shutoff Button .....CHECK NOT PRESSED

APU RUNNING NORMALLY?

No

Yes

Disregard the message.

END

APU SHUTOFF VALVE REMAINS CLOSED?

No

Yes

APU Master Knob.....OFF

END

DEFUEL VALVE NOT CLOSED

EICAS Caution: DEFUEL NOT CLOSED

CONDITION?

On Ground

In Flight

LAND AT THE NEAREST SUITABLE AIRPORT.

VTRL TK XFER.....OFF
XFEED .....OFF
Fuel Imbalance .....MONITOR
Asymmetric Thrust.....AS REQUIRED
Avoid uncoordinated maneuvers and negative g's.

END

Do not takeoff

END

Copyright © by Embraer. Refer to cover page for details.

ENGINE FUEL LOW PRESSURE

EICAS Caution: E1 (2) FUEL LO PRESS
Condition: One or more affected tank electric fuel pump may be inoperative.

Associated Fuel Pump Sel ..... SELECT ANOTHER

ALL PUMPS INOPERATIVE IN THE AFFECTED TANK?

No

Yes

Altitude ..... MAX 25'000 FT, MIN MEA

NOTE: If required, the flight may proceed above 25'000 ft using XFEED.

Avoid rapid thrust lever movements and set minimum required thrust.

END

END

ENGINE FUEL LOW TEMPERATURE

EICAS Caution: E1 (2) FUEL LO TEMP

WARNING: **IF NO ICING INHIBITOR WAS ADDED, ENGINE FLAMEOUT MAY OCCUR.**

ON GROUND?

No

Yes

Do not take off.

END

Descend to lower altitude and monitor engine indications as long as the message remains.

END

ENGINE FUEL SHUTOFF VALVE INOPERATIVE

EICAS Caution: E1 (2) FUEL SOV INOP

AFTER PULLING FIRE HANDLE? — **No**

Yes

XFEED ..... OFF
Associated Fuel Pump Pwr ..... OFF

END

Fire Extinguishing Handle ..... CHECK IN

ENGINE OPERATION NORMAL? — **No**

Yes

Message ..... DISREGARD

END

LAND AT THE NEAREST SUITABLE AIRPORT.
Associated ENGINE FAILURE/SHUTDOWN Procedure (NAP-19)..... ACCOMPLISH

END

FUEL CROSSFEED FAILURE

EICAS Caution: FUEL XFEED FAIL

Fuel Imbalance ..... MONITOR
Asymmetric Thrust..... AS REQUIRED

END

FUEL CROSSFEED MISCOMMAND

EICAS Caution: FUEL EQ XFEED OPN

XFEED ..... OFF
Fuel Imbalance ..... CHECK
XFEED ..... AS REQUIRED
Check XFEED selector knob properly positioned to correct wing fuel imbalance.

END

FUEL IMBALANCE

EICAS Caution: FUEL IMBALANCE

NOTE: Crossfeed must be off during takeoff and landing.

FUEL LEAK SUSPECTED? **No**

Yes

FUEL LEAK Procedure (NAP-8) ..... ACCOMPLISH

END

ATTITUDE ..... WINGS LEVELED

RH WING LOWER LEVEL? **No**

Yes

Fuel Xfeed Selector ..... LOW 2
Avoid rapid thrust levers movement.
Monitor fuel imbalance.
When the desired balance is achieved:
Fuel Xfeed Selector ..... OFF

END

Fuel Xfeed Selector ..... LOW 1
Avoid rapid thrust levers movement.
Monitor fuel imbalance.
When the desired balance is achieved:
Fuel Xfeed Selector ..... OFF

END

FUEL TANK LOW TEMPERATURE

EICAS Caution: FUEL TANK LO TEMP
MFD Indication: Fuel temperature in amber range.

WARNING: ENGINE FLAMEOUT MAY OCCUR.

ON GROUND? **No**

Yes

Do not take off.

END

Descend to lower altitude and monitor engine indications as long as the message remains.

END

FUEL TRANSFER ISOLATION FAILURE

EICAS Caution: XFER ISOL FAIL

CONDITION?

On Ground

In Flight

FUEL TRANSFER?

Operative

Not Operative

VTRL TK XFER.....OFF

FUEL XFER INOP message may appear.

Consider diversion.

Remaining Fuel.....CHECK

END

Do not take off.

END

FUEL TRANSFER OVERRIDE

EICAS Advisory: FUEL XFER OVRD

When Ventral Tank Fuel Quantity is lower than 220 lb:

VTRL TK XFER.....OFF

END

FUEL TRANSFER SYSTEM INOPERATIVE

EICAS Caution: FUEL XFER INOP

CONDITION?

On ground

In flight

VTRL TK XFER ..... AUTO

MESSAGE PERSISTS? **No**

Yes

Pitch attitude ..... MAX 5° NOSE UP

..... WAIT 30 SECONDS

MESSAGE PERSISTS? **No**

Yes

VTRL PUMP SEL ..... PRESS ONCE

MESSAGE PERSISTS? **No**

Yes

VENTRAL TANK QUANTITY IS ABOVE 220 LB? **No**

Yes

VTRL TK XFER ..... OVRD

Avoid pitch attitudes in excess of 5° nose up, uncoordinated maneuvers and negative g's.

When ventral tank fuel quantity is lower than 220 lb:

VTRL TK XFER..... OFF

Consider diversion.
Fuel Quantity..... MONITOR
Remaining Fuel ..... CHECK

Do not takeoff.

END

VENTRAL TANK VENTILATION OPEN

EICAS Caution: FUEL VENT OPEN

CONDITION?

On ground

In flight

While there is fuel in the Ventral Tank:

Pitch attitude .....MAX 10° NOSE DOWN

Avoid uncoordinated maneuvers.

END

Do not takeoff.

END

WING TANKS OVERFLOW

EICAS Caution: FUEL XFER OVERFLOW

VTRL TK XFER .....OFF
Wing Fuel Tank Quantity.....MONITOR
XFEED.....AS REQUIRED

When Wing Fuel Tanks quantity is at 4630 lb or below:

VTRL TK XFER .....AUTO
Fuel Imbalance.....MONITOR
XFEED.....AS REQUIRED

END

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

BOTH HYDRAULIC SYSTEMS FAILURE..... EAP 10-3

HYDRAULIC SYSTEM 1 FAILURE..... EAP 10-5
HYDRAULIC SYSTEM 2 FAILURE..... EAP 10-6
HYDRAULIC SYSTEM LOW QUANTITY..... EAP 10-7
HYDRAULIC SYSTEM OVERHEAT ..... EAP 10-7

Copyright © by Embraer. Refer to cover page for details.

LIST OF EICAS MESSAGES

HYD SYS 1-2 FAIL ..... EAP 10-3
HYD SYS 1 FAIL..... EAP 10-5
HYD SYS 2 FAIL..... EAP 10-6
HYD SYS 1 (2) OVHT..... EAP 10-7

HYD1 (2) LO QTY..... EAP 10-7

Copyright © by Embraer. Refer to cover page for details.

BOTH HYDRAULIC SYSTEMS FAILURE

EICAS Caution: HYD SYS 1-2 FAIL
MFD Indication: Hydraulic pressure may be amber.
Condition: Noise increase due to nose landing gear doors open.

The following messages will be displayed:

EICAS Caution: AIL SYS 1-2 INOP,
RUDDER SYS 1-2 INOP

EICAS Advisory: E1-2 HYD PUMP FAIL

CAUTION: DO NOT OPEN THE SPEED BRAKES.

Airspeed..... MAX 250 KIAS

ANY HYDRAULIC SYSTEM FLUID QUANTITY IN GREEN RANGE?

No

Yes

System(s) in green range:
Hydraulic Elec Pump(s)..... ON

AT LEAST ONE SYSTEM RECOVERED?

No

Yes

WHICH SYSTEM RECOVERED?

Hydraulic 2

Hydraulic 1

HYDRAULIC SYSTEM 2 FAILURE
Procedure (EAP 10-6) ..... AS REQUIRED

END

HYDRAULIC SYSTEM 1 FAILURE
Procedure (EAP 10-5) ..... AS REQUIRED

END

LAND AT THE NEAREST SUITABLE AIRPORT.

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 3.45.

Autopilot/Yaw Damper..... DISENGAGE

Both Hydraulic Elec Pumps..... OFF

Expect greater aileron and rudder control force. If required, both pilots should act together to control airplane. Consider using aileron and asymmetric thrust to help yaw control.

CONTINUES ON NEXT PAGE

Copyright © by Embraer. Refer to cover page for details.

CONTINUED FROM PREVIOUS PAGE

Relevant Inoperative Items:

Normal gear extension	**Thrust reversers**	**Anti-skid**
Normal brakes	**Spoilers**	**Steering**
Main door retraction		

Approach:

APU..... AS REQUIRED
Altimeters..... SET AND CROSS CHECKED
Approach Aids..... SET AND CROSS CHECKED
Speed Bugs..... SET
Pressurization..... CHECK

Go-Around Procedure ..... REVIEW

If necessary, accomplish a normal go-around procedure except that landing gear cannot be retracted.

Before Landing:

Free Fall Lever..... ACTUATE
Landing Gear Lever..... DOWN

Perform a long final approach.

Avoid landings at airports with anticipated crosswind or turbulence.

Excessive flare may require a bigger landing distance.

Use rudder for directional control on ground.

During landing run, pull Emergency Brake Handle carefully.

Landing Configuration:

Landing Gear ..... DOWN
Flaps..... 22°
V_{REF} ..... V_{REF45} + 30 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 3.45.

END

Copyright © by Embraer. Refer to cover page for details.

HYDRAULIC SYSTEM 1 FAILURE

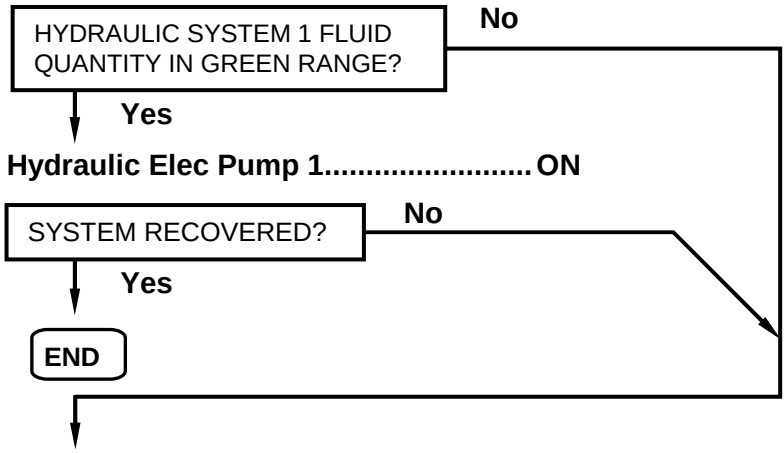
EICAS Caution: HYD SYS 1 FAIL

MFD Indication: Hydraulic pressure may be amber.
Condition: Noise increase due to nose landing gear doors open.

The following messages will be displayed:

EICAS Caution: AIL SYS 1 INOP,
RUDDER SYS 1 INOP

EICAS Advisory: E1 HYD PUMP FAIL



Hydraulic Elec Pump 1.....OFF
Airspeed.....MAX 250 KIAS
Relevant Inoperative Items:

Inboard Spoiler	**Normal gear extension**	**Thrust reverser 1**
Steering	**Main door retraction**	**Outboard brakes**

Before Landing:

Free Fall Lever .....ACTUATE

Landing Gear Lever.....DOWN

Brake effectiveness will be reduced.

Do not actuate engine 1 Thrust Reverser.

Landing Configuration:

Landing Gear .....DOWN

Flaps.....45°

V_{REF} ..... V_{REF45}

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.60.

END

Copyright © by Embraer. Refer to cover page for details.

HYDRAULIC SYSTEM 2 FAILURE

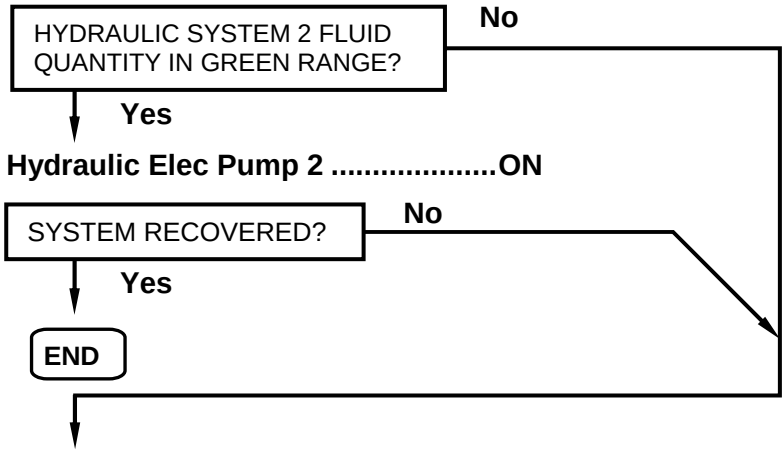
EICAS Caution: HYD SYS 2 FAIL
MFD Indication: Hydraulic pressure may be amber.

The following messages will be displayed:

EICAS Caution: AIL SYS 2 INOP,
RUDDER SYS 2 INOP

EICAS Advisory: E2 HYD PUMP FAIL

CAUTION: DO NOT OPEN THE SPEED BRAKES.



Hydraulic Elec Pump 2 .....OFF
Airspeed .....MAX 250 KIAS
Relevant Inoperative Items:

Outboard Spoiler	**Thrust reverser 2**	**Inboard brakes**

The Emergency/Parking Brake has accumulator pressure only.

Brake effectiveness will be reduced.

Do not actuate engine 2 Thrust Reverser.

Landing Configuration:

Landing Gear .....DOWN
Flaps.....45°
V_{REF} .....V_{REF45}

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.53.

END

Copyright © by Embraer. Refer to cover page for details.

HYDRAULIC SYSTEM LOW QUANTITY

EICAS Advisory: HYD1 (2) LO QTY

MFD Indication: Hydraulic fluid quantity may be amber.

Affected Hydraulic System ..... MONITOR

NOTE: If Hydraulic System 2 is affected, do not open the Speed Brakes.

END

HYDRAULIC SYSTEM OVERHEAT

EICAS Caution: HYD SYS 1 (2) OVHT

Turn the affected system OFF:

Associated Hydraulic
Eng Pump Shutoff..... PUSH IN

Associated Hydraulic
Elec Pump ..... OFF

EICAS Messages related to associated hydraulic system will be displayed while system is set to OFF.

Airspeed..... MAX 250 KIAS

For remainder of flight, if required:

Affected Hydraulic System..... 15 MINUTES OFF, 1 MINUTE ON

NOTE: To turn the hydraulic system ON, first turn the Hydraulic Elec Pump to AUTO. As soon as the system pressure is recovered, push out the Hydraulic Eng Pump Shutoff button.

During Approach and Landing or when required:

Affected Hydraulic System..... ON

After reaching taxi speed or when the system is no longer required:

Affected Hydraulic System..... OFF

END

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

ICING CONDITIONS WITH ANTI-ICING INOPERATIVE ..... EAP 11-3
ANTI-ICING LOW CAPACITY ..... EAP 11-4
ANTI-ICING SWITCH OFF ..... EAP 11-4
AOA HEATING INOPERATIVE ..... EAP 11-4
ENGINE ANTI-ICING FAILURE ..... EAP 11-5
ICE DETECTOR FAIL ..... EAP 11-5
NO ICE - ANTI-ICE ON ..... EAP 11-6
PITOT HEATING INOPERATIVE ..... EAP 11-6
STABILIZER ANTI-ICING FAILURE ..... EAP 11-8
TAT HEATING INOPERATIVE ..... EAP 11-7
WINDSHIELD HEATING FAILURE ..... EAP 11-7
WING ANTI-ICING FAILURE ..... EAP 11-8

NON ANNUNCIATED PROCEDURES

SINGLE ENGINE BLEED OPERATION IN ICING CONDITIONS ..... refer to NAP-34

LIST OF EICAS MESSAGES

ICE COND-A/I INOP ..... EAP 11-3
A/ICE LOW CAPACIT..... EAP 11-4
A/ICE SWITCH OFF ..... EAP 11-4
AOA 1 (2) HEAT INOP..... EAP 11-4
E1 (2) A/ICE FAIL ..... EAP 11-5
ICE DET1 (2) FAIL..... EAP 11-5
ICE DETECTORS FAIL ..... EAP 11-5
NO ICE-A/ICE ON..... EAP 11-6
PITOT 1 (2, 3) INOP ..... EAP 11-6
STAB A/ICE FAIL..... EAP 11-8
TAT 1 (2) HEAT INOP ..... EAP 11-7
W/S 1 (2) HEAT FAIL ..... EAP 11-7
WG A/ICE FAIL ..... EAP 11-8

ICING CONDITIONS WITH ANTI-ICING INOPERATIVE

EICAS Warning: ICE COND-A/I INOP

CONDITION?

On Ground

In Flight

Ice Detection Override Knob ..... ALL
Icing Conditions ..... EXIT
Associated Failure Procedure ..... ACCOMPLISH

END

Thrust Levers ..... IDLE
Ice Detection Override Knob ..... AUTO
Ice Detection Test Knob ..... 1, FOR 5 SECONDS, THEN 2, FOR 5 SECONDS

ALL OPEN INSCRIPTIONS FLASH AT LEAST ONCE AND ICE DET 1 (OR 2) FAIL AND ICE CONDITION DISPLAYED?

No

Yes

ANTI-ICING SYSTEM EICAS MESSAGES DISAPPEAR WITHIN 60 SECONDS?

No

Yes

TAKEOFF Procedure ..... ACCOMPLISH

END

Do not takeoff.

END

ANTI-ICING LOW CAPACITY

EICAS Caution: A/ICE LOW CAPACIT

Thrust Levers .....ADVANCE
Advance Thrust Levers to at least 55% N1.

.....WAIT 5 SECONDS

MESSAGE PERSISTS?

No

Yes

WING ANTI-ICING FAILURE OR
STABILIZER ANTI-ICING FAILURE
Procedure (EAP 11-8).....AS REQUIRED

END

ANTI-ICING SWITCH OFF

EICAS Caution: A/ICE SWITCH OFF

All Ice Protection Buttons .....PUSH IN

END

AOA HEATING INOPERATIVE

EICAS Caution: AOA 1 (2) HEAT INOP

Minimum Airspeed.....FLAP MANEUVERING SPEED (PD-2)

END

ENGINE ANTI-ICING FAILURE

EICAS Caution: E1 (2) A/ICE FAIL

Thrust Levers ..... ADVANCE

MESSAGE PERSISTS? **No**

Yes

Ice Detection Override Knob ..... ALL

MESSAGE STILL PERSISTS? **No**

Yes

Ice Protection Engine Air Inlet ..... PUSH OUT, THEN PUSH IN

MESSAGE STILL PERSISTS? **No**

Yes

Icing Conditions ..... AVOID/EXIT
Two minutes after exiting icing conditions:
Ice Detection Override Knob ..... AUTO

Engine Vibration ..... MONITOR

If vibration increases, advance thrust levers one at a time, to 60% N1 minimum for 5 seconds.

If vibration increases to unacceptable values or engine parameters indicate abnormal values, exit icing conditions.

END

ICE DETECTOR FAIL

EICAS Caution: ICE DET1 (2) FAIL or
ICE DETECTORS FAIL

When flying in icing conditions:
Ice Detection Override Knob ..... ALL
Two minutes after exiting icing conditions:
Ice Detection Override Knob ..... AUTO

END

NO ICE - ANTI-ICE ON

EICAS Caution: NO ICE-A/ICE ON

Ice Detection Override Knob.....AUTO

MESSAGE PERSISTS? **No**

Yes

ASSOCIATED TO EITHER **WG A/ICE FAIL** OR **STAB A/ICE FAIL** MESSAGE? **No**

Yes

Altitude .....MAX 25'000 FT, MIN MEA
APU .....START
APU Bleed .....PUSH IN
XBleed .....OPEN
Engine Bleeds.....PUSH OUT

END

Ice Protection Buttons.....PUSH OUT

Operate the Anti-Icing Systems through the buttons, pushing them in again whenever icing conditions are encountered.

END

PITOT HEATING INOPERATIVE

EICAS Caution: PITOT 1 (2, 3) INOP

Instruments' information supplied by the affected system may be unreliable. Cross-check and do not use the affected system if a disagreement is found.
If the Pitot 3 heating is inoperative, standby instruments and pressurization system may be affected.
If necessary:

ADC on Associated Reversionary Panel.....PUSH IN

END

TAT HEATING INOPERATIVE

EICAS Caution: TAT 1 (2) HEAT INOP

TAT, TAS and SAT indication may be unreliable.

END

WINDSHIELD HEATING FAILURE

EICAS Caution: W/S 1 (2) HEAT FAIL

Associated Ice Protection
Windshield ..... PUSH OUT

END

WING ANTI-ICING FAILURE OR STABILIZER ANTI-ICING FAILURE

EICAS Caution: STAB A/ICE FAIL or WG A/ICE FAIL

Ice Detector Override Knob..... ALL
Thrust Levers ..... ADVANCE

MESSAGE PERSISTS? **No**

Yes

Affected Ice Protection Button..... PUSH OUT, THEN PUSH IN

MESSAGE STILL PERSISTS? **No**

Yes

Associated Ice Protection Button..... PUSH OUT
Icing Conditions..... AVOID/EXIT
Two minutes after exiting icing conditions:
Ice Detector Override Knob ..... AUTO
Maximum Bank Angle..... 30°
Minimum Airspeed (Flaps 0° or 9°)..... 190 KIAS

LANDING IN ICING CONDITIONS OR WITH ICE ACCRETION? **No**

Yes

AFFECTED SYSTEM(S)? **Stab**

Wing or Wing+Stab

Landing configuration:
Flaps..... 22°
V_{REF} ..... V_{REF45} + 30 KIAS
CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.85.

END

Landing configuration:
Flaps..... 22°
V_{REF} ..... V_{REF45} + 15 KIAS
CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.55.

END

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

LANDING GEAR/LEVER DISAGREE ..... EAP 12-3
BRAKE OVERHEAT ..... EAP 12-4
BRAKES DEGRADED ..... EAP 12-5
BRAKES INOPERATIVE..... EAP 12-5
EMERGENCY/PARKING BRAKE LOW PRESSURE..... EAP 12-5
LANDING GEAR AIR/GROUND SYSTEM FAILURE..... EAP 12-6
STEERING SYSTEM INOPERATIVE ..... EAP 12-7
UNCOMMANDED SWERVING ON GROUND..... EAP 12-7

NON ANNUNCIATED PROCEDURES

ABNORMAL LANDING GEAR EXTENSION..... refer to NAP-13
EMERGENCY/PARKING BRAKE HANDLE DISAGREE..... refer to NAP-15
GEAR LEVER CANNOT MOVE UP AFTER TAKEOFF ..... refer to NAP-22
NOSE LANDING GEAR UP DOOR OPEN ..... refer to NAP-30
PARTIAL OR GEAR UP LANDING ..... refer to NAP-32

Copyright © by Embraer. Refer to cover page for details.

LIST OF EICAS MESSAGES

LG/LEVER DISAGREE ..... EAP 12-3
BRAKE OVERHEAT ..... EAP 12-4
BRAKE DEGRADED..... EAP 12-5
BRK INBD INOP ..... EAP 12-5
BRK OUTBD INOP ..... EAP 12-5
EMRG BRK LO PRES ..... EAP 12-5
LG AIR/GND FAIL..... EAP 12-6
STEER INOP ..... EAP 12-7

Copyright © by Embraer. Refer to cover page for details.

LANDING GEAR/LEVER DISAGREE

EICAS Warning: LG/LEVER DISAGREE
EICAS Caution: NLG UP/DOOR OPN may be displayed.
EICAS Indication: Landing gear abnormal indication.
Condition: Landing gear cannot move to desired position.

ON GROUND?
No
Yes
Consider to stop the airplane. Takeoff is not permitted.
END

Airspeed..... MAX 200 KIAS
Landing Gear..... CYCLE

LG/LEVER DISAGREE MESSAGE PERSISTS?
No
Yes
DESIRED LANDING GEAR POSITION?
Extended
Retracted
LAND AT THE NEAREST SUITABLE AIRPORT.
Landing Gear Lever ..... DOWN
Airspeed..... MAX 250 KIAS
Icing Conditions..... AVOID/EXIT
Fuel Consumption..... MONITOR
END

ABNORMAL LANDING GEAR EXTENSION
Procedure (NAP-13)..... ACCOMPLISH

END

Copyright © by Embraer. Refer to cover page for details.

BRAKE OVERHEAT

EICAS Caution: BRAKE OVERHEAT

ON GROUND? **No**

Yes

If necessary, reduce brake pressure.
Use of full reverse thrust is recommended.
Airplane ..... STOP
Chocks..... ON
Parking Brake..... RELEASE

END

Airspeed ..... MAX 250 KIAS
Landing Gear Lever ..... DOWN
Brakes Temperature ..... CHECK IN THE GREEN RANGE

If MFD is not available, retract gear after message is removed.
Airspeed ..... MAX 200 KIAS
Landing Gear Lever ..... UP

END

Copyright © by Embraer. Refer to cover page for details.

BRAKES DEGRADED

EICAS Caution: BRAKE DEGRADED

Brake effectiveness and symmetry may be affected.

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.24.

During landing run:

Brakes ..... APPLY NORMALLY

Use thrust reverser if available.

END

BRAKES INOPERATIVE

EICAS Caution: BRK OUTBD (INBD) INOP

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.45.

During landing run:

If available, use thrust reverser.

BRAKES INOPERATIVE?

Both (OUTBD and INBD)

Only one (OUTBD or INBD)

Brakes ..... APPLY NORMALLY

END

Emergency Brake Handle ..... PULL CAREFULLY

Relevant Inoperative Item: **Anti-skid**

END

EMERGENCY/PARKING BRAKE LOW PRESSURE

EICAS Caution: EMRG BRK LO PRES

Emergency/Parking Brake performance may be degraded.

When parking the airplane, use wheel chocks.

END

Copyright © by Embraer. Refer to cover page for details.

LANDING GEAR AIR/GROUND SYSTEM FAILURE

EICAS Caution: LG AIR/GND FAIL

NOTE: - Thrust Reversers, Steering and Ground Spoiler may not be available.
- Depending on the failed condition, Ground Idle may not be selectable.
- If the message is presented on ground, a loss of the main brake capacity may occur (below 10 kt ground speed) and steering may not be available.
- Refer to the associated procedures for each case.

ICING CONDITIONS? **No**

Yes

Anti-Icing System..... MONITOR

If any anti-ice valve does not open or anti-ice failure messages appear, exit and avoid icing conditions.

After exiting icing conditions, proceed as follows:

Maximum Bank Angle..... 30°

Minimum Airspeed for Flaps up or 9° ..... 190 KIAS

Landing Configuration:

Flaps ..... 22°

Airspeed..... $V_{REF\ 45}$ + 30 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 2.40.

END

Icing Conditions..... EXIT/AVOID

Landing Configuration:

Flaps ..... 45°

Airspeed..... $V_{REF\ 45}$

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.70.

END

Copyright © by Embraer. Refer to cover page for details.

STEERING SYSTEM INOPERATIVE OR UNCOMMANDED SWERVING ON GROUND

EICAS Caution: STEER INOP may be displayed.

Steering Handwheel.....DO NOT USE
Steering Disengagement Button.....PRESS

Control the airplane using differential brakes and rudder.

Consider the use of differential thrust if serviceable.

END

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

CREW OXYGEN LOW PRESSURE..... EAP 13-3
OXYGEN LOW PRESSURE..... EAP 13-3
PASSENGER OXYGEN LOW PRESSURE..... EAP 13-3

NON ANNUNCIATED PROCEDURES

OXYGEN LEAKAGE..... refer to NAP-31

LIST OF EICAS MESSAGES

CREW OXYGEN LO PRESS..... EAP 13-3
OXYGEN LO PRESS..... EAP 13-3
PAX OXYGEN LO PRESS ..... EAP 13-3

(CREW/PASSENGER) OXYGEN LOW PRESSURE

EICAS Caution: OXYGEN LO PRESS
CREW (PAX) OXYGEN LO PRESS

MFD Indication: Oxygen pressure red or amber.

Altitude .....MEA OR 10’000 FT, WHICHEVER IS HIGHER

END

INTENTIONALLY BLANK

TABLE OF CONTENTS

ANNUNCIATED PROCEDURES

STALL PROTECTION INOPERATIVE ..... EAP 14-3
STICK PUSHER FAILURE ..... EAP 14-4
TAKEOFF CONFIGURATION WARNING ..... EAP 14-4
ADVANCED STALL PROTECTION ..... EAP 14-5
AURAL WARNING FAIL ..... EAP 14-6
GPWS INOPERATIVE ..... EAP 14-6
WINDSHEAR DETECTION INOPERATIVE ..... EAP 14-6

NON ANNUNCIATED PROCEDURES

AIRPLANE OVERSPEED refer to NAP-4
ERRONEOUS STALL PROTECTION ACTUATION refer to NAP-22

LIST OF EICAS MESSAGES

SPS 1 (2) INOP ..... EAP 14-3
SPS 1-2 INOP ..... EAP 14-4
NO TAKEOFF CONFIG ..... EAP 14-4
AURAL WARN FAIL ..... EAP 14-6
GPWS INOP ..... EAP 14-6
SPS ADVANCED ..... EAP 14-5
STICK PUSHER FAIL ..... EAP 14-4
TERR INOP ..... EAP 14-6
WINDSHEAR INOP ..... EAP 14-6

STALL PROTECTION INOPERATIVE

EICAS Warning: SPS 1 (2) INOP
EICAS Caution: SPS ADVANCED

Affected Stall Protection Cutout Button ..... PUSH OUT

The following messages will be displayed:

EICAS Warning: SPS 1 (2) INOP
EICAS Caution: STICK PUSHER FAIL

Minimum Airspeed ..... FLAP MANEUVERING SPEED (PD-2)

Avoid skidding the airplane.

Add 5 KIAS to approach and go-around speeds.

Landing configuration:

Landing Gear ..... DOWN
Flaps ..... 45°
Airspeed ..... $V_{REF\ 45}$ + 5 KIAS

NOTE: - The remaining stick shaker is available.
- Stick pusher is not available.

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.10.

END

STICK PUSHER FAILURE

EICAS Warning: SPS 1-2 INOP
EICAS Caution: STICK PUSHER FAIL

Stall ..... RECOVER

Minimum Airspeed ..... FLAP MANEUVERING SPEED (PD-2)

STICK PUSHER JAMMED?

No

Yes

JAMMED ELEVATOR Procedure (NAP-8) ..... ACCOMPLISH

END

Both stick shaker 1 and 2 are still available.

Add 5 KIAS to V_{REF}, approach and approach-climb speeds.

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.15.

END

TAKEOFF CONFIGURATION WARNING

EICAS Warning: NO TAKEOFF CONFIG
Aural Warning: Voice Messages TAKEOFF-BRAKES, TAKEOFF-FLAPS, TAKEOFF-TRIM, TAKEOFF-SPOILERS
EICAS Indication: Spoiler and pitch trim may be red.

Do not take off.

Airplane Configuration ..... CORRECT

TO Config Button ..... PRESS

END

ADVANCED STALL PROTECTION

EICAS Caution: SPS ADVANCED

Above 25’000 ft:
Minimum Airspeed..... 150 KIAS

Below 25’000 ft:
Minimum Airspeed..... FLAP MANEUVERING SPEED (PD-2)

Add 5 KIAS to approach and go-around speeds.

Landing configuration:
Landing Gear ..... DOWN
Flaps..... 45°
Airspeed ..... V_{REF45} + 5 KIAS

CAUTION: MULTIPLY THE FLAPS 45° UNFACTORED LANDING DISTANCE BY 1.10.

END

AURAL WARNING FAIL

EICAS Caution: AURAL WARN FAIL

Visually monitor every EICAS, MFD and PFD indication specially related to TCAS, Windshear Detection, GPWS, IC-600, Fire Detection, Stall Protection, Trims, Flaps, Brakes, Spoilers, Radio Altimeter, Autopilot, Landing gear, ADC, Pressurization, SELCAL. No aural warning will be available.

Do not perform CAT II or CAT III approaches.

END

GPWS INOPERATIVE

EICAS Caution: GPWS INOP or
GPWS INOP and TERR INOP (for EGPWS)

Monitor visually any trend toward terrain contact, excessive sink rate, marginal flight path and airplane configuration. No aural warning related to the system will be available.

END

WINDSHEAR DETECTION INOPERATIVE

EICAS Caution: WINDSHEAR INOP

Windshear detection is not available.

END

WIND COMPONENT TABLE

ANGLE BETWEEN WIND DIRECTION AND HEADING (LEFT OR RIGHT)

WIND SPEED (kt)	10	20	30	40	50	60	70	80	100	110	120	130	140	150	160	170	10	20	30	40	50	60	70	80	100	110	120	130	140	150	160	170	WIND SPEED (kt)
	HEADWIND COMPONENT (kt)								TAILWIND COMPONENT (kt)								CROSSWIND COMPONENT (kt)																
5	5	5	4	4	3	3	2	1	-1	-2	-3	-3	-4	-4	-5	-5	1	2	3	3	4	4	5	5	5	5	5	6	6	7	8	9	5
10	10	9	9	8	6	5	3	2	-2	-3	-5	-6	-8	-9	-9	-10	2	3	5	6	8	9	9	10	10	10	11	12	14	15	17	17	10
15	15	14	13	11	10	8	5	3	-3	-5	-8	-10	-11	-13	-14	-15	3	5	8	10	11	13	14	15	15	16	17	18	20	22	23	25	15
20	20	19	(17)	15	13	10	7	3	-3	-7	-10	-13	-15	-17	-19	-20	3	7	(10)	13	15	17	19	20	20	21	22	24	26	29	31	33	20
25	25	23	22	19	16	13	9	4	-4	-9	-13	-16	-19	-22	-23	-25	4	9	13	16	19	22	23	25	25	26	28	30	33	36	39	41	25
30	30	28	26	23	19	15	10	5	-5	-10	-15	-19	-23	-26	-28	-30	5	10	15	19	23	26	28	30	30	31	33	36	39	43	47	49	30
35	34	33	30	27	22	18	12	6	-6	-12	-18	-22	-27	-30	-33	-34	6	12	18	22	27	30	33	34	35	36	39	42	46	50	54	57	35
40	39	38	35	31	26	20	14	7	-7	-14	-20	-26	-31	-35	-38	-39	7	14	20	26	31	35	38	39	40	41	44	48	52	57	62	65	40
45	44	42	39	34	29	23	15	8	-8	-15	-23	-29	-34	-39	-42	-44	8	15	23	29	34	39	42	44	45	46	50	54	59	64	69	73	45
50	49	47	43	38	32	25	17	9	-9	-17	-25	-32	-38	-43	-47	-49	9	17	25	32	38	43	47	49	50	51	55	60	65	71	77	81	50

Example: Given Wind Speed=20 kt and Angle (between wind and nose)= 30°, the Headwind Component is 17 kt and the Crosswind component is 10 kt.

Shaded areas are not allowed for CAT II operations.

REFERENCE CROSSWIND VALUES

Surface Condition	Dry/ Wet	Compacted Snow	Standing Water/ Slush/ Dry Snow	-	Ice
Brake Action	-	Good	-	Medium	Poor
Friction Coefficient	-	0.40	-	0.30	<0.20
Reference Crosswind	30 kt	25 kt	20 kt	17 kt	10 kt

Copyright © by Embraer. Refer to cover page for details.

PITCH TRIM UNITS						
EMB-145	CG POSITION (%)	LESS THAN OR EQUAL TO 27.5	27.6 UP TO 32.5	32.6 UP TO 36.5	36.6 UP TO 41.5	ABOVE OR EQUAL TO 41.6
PITCH TRIM UNITS		8	7	6	5	4

FLAP RETRACTION SCHEDULE		
ALL ENGINES	*For a flaps 9° takeoff:*	
	Flaps 9° to UP	V_2 + 15 KIAS
A1P and A1/3 ENGINES	*For a flaps 18° takeoff:*	
	Flaps 18° to 9°	V_2 + 10 KIAS
	Flaps 9° to UP	V_2 + 30 KIAS
A, A1, A1/1 ENGINES	*For a flaps 22° takeoff:*	
	Flaps 22° to 9°	V_2 + 5 KIAS
	Flaps 9° to UP	V_2 + 25 KIAS
A1P ENGINES	*For a flaps 22° takeoff:*	
	Flaps 22° to 9°	V_2 + 10 KIAS
	Flaps 9° to UP	V_2 + 30 KIAS

FLAP MANEUVERING SPEED		
GEAR-FLAP	No Icing Conditions	Icing Conditions
UP-0°	180 KIAS	200 KIAS
UP/DN-9°	160 KIAS	160 KIAS
UP/DN-18°/22°	140 KIAS	150 KIAS
DN-45°	140 KIAS	140 KIAS

Copyright © by Embraer. Refer to cover page for details.

UNRELIABLE AIRSPEED TABLES (CLB Thrust Mode)

Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: OFF

PRESSURE ALTITUDE (ft)		WEIGHT (lb)					
		30000	35000	40000	45000	50000	55000
0 (240 KIAS)	Pitch (deg)	13	12	11	10	10	9
	V/S (ft/min)	4800	4000	3400	2900	2500	2200
10000 (240 KIAS)	Pitch (deg)	10	9	8	8	8	7
	V/S (ft/min)	3800	3100	2700	2200	1900	1600
20000 (0.56 M)	Pitch (deg)	7	6	6	5	5	5
	V/S (ft/min)	3500	2800	2300	1800	1500	1200
30000 (0.56 M)	Pitch (deg)	7	7	7	8	7	7
	V/S (ft/min)	2700	2000	1600	1200	900	600
37000 (0.56 M)	Pitch (deg)	7	7	8	8	8	-
	V/S (ft/min)	1800	1300	900	400	100	

UNRELIABLE AIRSPEED TABLES (Cruise)

Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: OFF

PRESSURE ALTITUDE (ft)		WEIGHT (lb)					
		30000	35000	40000	45000	50000	55000
15000 (250 KIAS)	Pitch (deg)	1	1	2	2	2	3
	N1 (%)	68.7	69.9	71.1	72.4	73.8	75.3
20000 (250 KIAS)	Pitch (deg)	1	1	2	2	2	3
	N1 (%)	72.0	73.3	74.8	76.4	78.0	79.6
25000 (250 KIAS)	Pitch (deg)	1	1	2	2	2	3
	N1 (%)	76.1	77.6	79.3	80.7	81.9	83.1
30000 (0.63 M)	Pitch (deg)	1	1	2	2	2	3
	N1 (%)	78.5	80.0	81.2	82.5	84.0	85.5
37000 (0.63 M)	Pitch (deg)	2	2	2	3	3	4
	N1 (%)	79.8	81.7	83.8	86.4	89.8	92.8

UNRELIABLE AIRSPEED TABLES (Flight Idle Descent)

Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: OFF

PRESSURE ALTITUDE (ft)		WEIGHT (lb)					
		30000	35000	40000	45000	50000	55000
0 (240 KIAS)	Pitch (deg)	-3	-2	-2	-1	-1	0
	V/S (ft/min)	-1800	-1700	-1600	-1500	-1500	-1500
10000 (240 KIAS)	Pitch (deg)	-3	-2	-2	-1	0	0
	V/S (ft/min)	-2000	-1900	-1800	-1800	-1700	-1700
20000 (240 KIAS)	Pitch (deg)	-2	-2	-1	-1	0	0
	V/S (ft/min)	-2200	-2100	-2000	-1900	-1900	-1900
30000 (240 KIAS)	Pitch (deg)	-2	-2	-1	0	0	1
	V/S (ft/min)	-2500	-2400	-2200	-2100	-2100	-2000
37000 (240 KIAS)	Pitch (deg)	-1	-1	0	0	1	1
	V/S (ft/min)	-2400	-2200	-2100	-2100	-2000	-2000

UNRELIABLE AIRSPEED TABLES (Holding)

Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: OFF

PRESSURE ALTITUDE (ft)		WEIGHT (lb)					
		30000	35000	40000	45000	50000	55000
5000 (200 KIAS)	Pitch (deg)	3	3	4	5	5	6
	N1 (%)	54.0	55.9	57.8	59.9	62.2	64.4
10000 (200 KIAS)	Pitch (deg)	3	3	4	5	5	6
	N1 (%)	57.5	59.5	61.8	63.9	66.2	68.4

UNRELIABLE AIRSPEED TABLES (Terminal Area)					
Terminal Area (5000 ft) - %N1 for Level Flight					
Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: OFF					
FLAP POSITION (V_{REF} + INCREMENT)		**WEIGHT (lb)**			
		30000	**35000**	**40000**	**45000**
0 (V_{REF45} + 30)	Pitch (deg) N1 (%)	7 48.9	7 52.5	7 55.7	8 58.8
9 (V_{REF45} + 15)	Pitch (deg) N1 (%)	7 51.6	7 55.3	7 58.6	8 61.7

UNRELIABLE AIRSPEED TABLES (Final Approach)					
Final Approach (1500 ft) - %N1 for 3° Glideslope					
Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: OFF					
FLAP POSITION (V_{REF} + INCREMENT)		**WEIGHT (lb)**			
		30000	**35000**	**40000**	**45000**
22 (V_{REF22} + 10)	Pitch (deg) N1 (%)	3 47.7	3 51.1	3 53.9	3 56.6
45 (V_{REF45} + 10)	Pitch (deg) N1 (%)	-1 58.6	0 62.4	0 65.7	0 68.6

UNRELIABLE AIRSPEED TABLES (CLB Thrust Mode)

Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: ON

PRESSURE ALTITUDE (ft)		WEIGHT (lb)					
		30000	35000	40000	45000	50000	55000
0 (240 KIAS)	Pitch (deg)	13	11	10	10	9	9
	V/S (ft/min)	4700	3900	3400	2900	2500	2200
10000 (240 KIAS)	Pitch (deg)	8	8	7	7	7	6
	V/S (ft/min)	3500	2800	2400	1900	1600	1400
20000 (0.56 M)	Pitch (deg)	6	5	5	5	5	5
	V/S (ft/min)	3000	2300	1900	1500	1200	900
30000 (0.56 M)	Pitch (deg)	6	6	6	6	6	6
	V/S (ft/min)	2000	1400	1100	700	400	200
37000 (0.56 M)	Pitch (deg)	6	6	7	7	-	-
	V/S (ft/min)	1200	700	400	0		

UNRELIABLE AIRSPEED TABLES (Cruise)

Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: ON

PRESSURE ALTITUDE (ft)		WEIGHT (lb)					
		30000	35000	40000	45000	50000	55000
15000 (250 KIAS)	Pitch (deg)	1	1	2	2	2	3
	N1 (%)	68.7	69.9	71.1	72.4	73.8	75.3
20000 (250 KIAS)	Pitch (deg)	1	1	2	2	2	3
	N1 (%)	72.0	73.4	74.8	76.4	78.0	79.6
25000 (250 KIAS)	Pitch (deg)	1	1	2	2	2	3
	N1 (%)	76.1	77.6	79.3	80.7	81.9	83.1
30000 (0.63 M)	Pitch (deg)	1	1	2	2	2	3
	N1 (%)	78.5	79.9	81.2	82.5	84.0	85.5
37000 (0.63 M)	Pitch (deg)	2	2	2	3	3	4
	N1 (%)	79.8	81.7	83.8	86.4	89.8	92.8

UNRELIABLE AIRSPEED TABLES (Flight Idle Descent)

Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: ON

PRESSURE ALTITUDE (ft)		WEIGHT (lb)					
		30000	35000	40000	45000	50000	55000
0 (240 KIAS)	Pitch (deg)	-1	0	0	0	1	1
	V/S (ft/min)	-900	-900	-900	-900	-900	-1000
10000 (240 KIAS)	Pitch (deg)	-1	0	0	0	1	1
	V/S (ft/min)	-1200	-1200	-1200	-1200	-1200	-1200
20000 (240 KIAS)	Pitch (deg)	-1	0	0	0	1	1
	V/S (ft/min)	-1300	-1300	-1300	-1300	-1300	-1400
30000 (240 KIAS)	Pitch (deg)	0	0	0	1	1	2
	V/S (ft/min)	-1400	-1400	-1400	-1400	-1400	-1400
37000 (240 KIAS)	Pitch (deg)	0	0	0	1	1	2
	V/S (ft/min)	-1700	-1600	-1600	-1500	-1500	-1500

UNRELIABLE AIRSPEED TABLES (Holding)

Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: ON

PRESSURE ALTITUDE (ft)		WEIGHT (lb)					
		30000	35000	40000	45000	50000	55000
5000 (200 KIAS)	Pitch (deg)	3	3	4	5	5	6
	N1 (%)	54.0	55.9	57.8	59.9	62.2	65.7
10000 (200 KIAS)	Pitch (deg)	3	3	4	5	5	6
	N1 (%)	57.5	59.5	61.8	65.7	66.2	69.7

UNRELIABLE AIRSPEED TABLES (Terminal Area)					
Terminal Area (5000 ft) - %N1 for Level Flight					
Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: ON					
FLAP POSITION (V_{REF} + INCREMENT)		**WEIGHT (lb)**			
		30000	**35000**	**40000**	**45000**
0 (V_{REF45} + 30)	Pitch (deg)	7	7	7	8
	N1 (%)	48.9	52.5	55.7	61.3
9 (V_{REF45} + 15)	Pitch (deg)	7	7	7	8
	N1 (%)	52.3	55.3	58.6	64.3

UNRELIABLE AIRSPEED TABLES (Final Approach)					
Final Approach (1500 ft) - %N1 for 3° Glideslope					
Airplane: EMB-145 – Engine: AE3007A1 & A1P – Anti-Ice: ON					
FLAP POSITION (V_{REF} + INCREMENT)		**WEIGHT (lb)**			
		30000	**35000**	**40000**	**45000**
22 (V_{REF22} + 10)	Pitch (deg)	3	3	3	3
	N1 (%)	48.9	52.5	55.7	61.3
45 (V_{REF45} + 10)	Pitch (deg)	-1	0	0	0
	N1 (%)	58.6	62.4	65.7	70.9

TAKEOFF SPEEDS (Balanced Field Length)

V1/VR/V2 SPEEDS
ALT T/O-1 MODE - NORMAL V_2 - FLAPS 9°

PRESSURE ALTITUDE (ft)	STATIC AIR TEMPERATURE (°C)			
SL ➔	-40 to 46	47 to 48	49 to 50	-
1000 ➔	-40 to 43	44 to 45	46 to 48	-
2000 ➔	-40 to 39	40 to 42	43 to 45	46 to 46
3000 ➔	-40 to 35	36 to 38	39 to 41	42 to 44
4000 ➔	-40 to 31	32 to 34	35 to 38	39 to 41
5000 ➔	-40 to 26	27 to 30	31 to 33	34 to 37
6000 ➔	-	-40 to 25	26 to 29	30 to 32
7000 ➔	-	-40 to 20	21 to 24	25 to 28
8000 ➔	-	-	-40 to 20	21 to 24
10000 ➔	-	-	-	-40 to 23
	↓	↓	↓	↓
WEIGHT (lb)	**V1 VR V2**	**V1 VR V2**	**V1 VR V2**	**V1 VR V2**
26500	100 104 122	98 102 119	97 100 117	95 98 114
27500	100 104 121	99 102 119	97 100 116	95 98 113
28500	100 104 120	99 102 118	97 100 115	95 98 113
29500	100 104 120	99 102 117	97 100 115	97 99 114
30500	100 103 119	98 101 117	98 101 115	99 101 115
31500	100 103 119	99 101 116	100 102 116	101 103 116
32500	101 103 118	102 104 118	103 104 118	103 105 118
33500	104 106 120	104 106 120	105 107 120	106 107 120
34500	106 108 122	107 108 122	108 109 122	108 109 122
35500	108 110 123	109 110 123	110 111 123	111 111 123
36500	111 112 125	111 112 125	112 113 125	113 114 125
37500	113 114 127	114 115 127	115 115 127	115 116 127
38500	115 116 128	116 117 128	117 117 128	117 118 128
39500	118 118 130	118 119 130	119 119 130	119 120 130
40500	120 120 132	120 121 132	121 121 132	121 122 132
41500	122 122 133	122 122 133	123 123 133	123 124 133
42500	124 124 135	124 124 135	125 125 135	125 125 135
43500	126 126 136	126 126 136	127 127 136	127 127 136
44500	128 128 138	128 128 138	129 129 138	129 129 138
45500	129 129 139	130 130 139	130 130 139	131 131 139
46500	131 131 141	132 132 141	132 132 141	133 133 141
47500	133 133 142	133 133 142	134 134 142	134 134 142
48500	134 134 144	135 135 144	135 135 144	136 136 144

FINAL SEGMENT SPEED (V_{FS})

WEIGHT (lb)	V_{FS} (KIAS)	WEIGHT (lb)	V_{FS} (KIAS)	WEIGHT (lb)	V_{FS} (KIAS)
26000	132	35000	152	44000	169
27000	134	36000	154	45000	171
28000	136	37000	156	46000	172
29000	139	38000	158	47000	173
30000	141	39000	160	48000	175
31000	142	40000	162	49000	177
32000	145	41000	164	50000	179
33000	147	42000	165		
34000	150	43000	167		

TAKEOFF SPEEDS (Balanced Field Length)			
V1/VR/V2 SPEEDS ALT T/O-1 MODE - NORMAL V_2 - FLAPS 9°			
PRESSURE ALTITUDE (ft)	STATIC AIR TEMPERATURE (°C)		
SL →	-	-	-
1000 →	-	-	-
2000 →	-	-	-
3000 →	-	-	-
4000 →	42 to 42	-	-
5000 →	38 to 40	-	-
6000 →	33 to 36	37 to 38	-
7000 →	29 to 32	33 to 35	36 to 36
8000 →	25 to 27	28 to 31	32 to 34
10000 →	-	22 to 25	26 to 32
	↓	↓	↓
WEIGHT (lb)	V1 VR V2	V1 VR V2	V1 VR V2
26500	94 96 112	92 94 109	91 92 107
27500	93 96 111	92 94 109	94 95 109
28500	94 96 111	95 97 111	96 98 111
29500	97 99 112	98 99 112	99 100 112
30500	99 101 114	100 102 114	101 102 114
31500	102 103 116	103 104 116	104 105 116
32500	104 106 118	105 106 118	106 107 118
33500	107 108 120	108 109 120	109 109 120
34500	109 110 122	110 111 122	111 111 122
35500	112 112 123	112 113 123	113 113 123
36500	114 114 125	115 115 125	115 115 125
37500	116 116 127	117 117 127	117 117 127
38500	118 118 128	119 119 128	119 119 128
39500	120 120 130	121 121 130	121 121 130
40500	122 122 132	123 123 132	123 123 132
41500	124 124 133	125 125 133	125 125 133
42500	126 126 135	126 126 135	127 127 135
43500	128 128 136	128 128 136	129 129 136
44500	130 130 138	130 130 138	131 131 138
45500	131 131 139	132 132 139	132 132 139
46500	133 133 141	134 134 141	134 134 141
47500	135 135 142	135 135 142	136 136 142
48500	136 136 144	137 137 144	137 137 144

TAKEOFF SPEEDS (Unbalanced Field Length)

VR/V2 SPEEDS
T/O-1 MODE - NORMAL V_2 - FLAPS 9°

PRESSURE ALTITUDE (ft)	STATIC AIR TEMPERATURE (°C)				
SL →	-40 to 39	40 to 44	45 to 48	49 to 50	-
1000 →	-40 to 35	36 to 40	41 to 44	45 to 48	-
2000 →	-40 to 30	31 to 36	37 to 41	42 to 46	-
3000 →	-	-40 to 32	33 to 37	38 to 42	43 to 44
4000 →	-	-40 to 27	28 to 33	34 to 38	39 to 42
5000 →	-	-	-40 to 28	29 to 34	35 to 40
6000 →	-	-	-40 to 24	25 to 30	31 to 36
7000 →	-	-	-	-40 to 25	26 to 31
8000 →	-	-	-	-40 to 21	22 to 27
10000 →	-	-	-	-	-40 to 26
	↓	↓	↓	↓	↓
WEIGHT (lb)	**VR V2**	**VR V2**	**VR V2**	**VR V2**	**VR V2**
26500	109 128	106 124	103 120	99 116	96 112
27500	109 127	105 123	102 119	99 115	96 111
28500	108 126	105 122	102 118	99 114	96 111
29500	108 125	105 121	102 117	98 114	98 113
30500	107 124	104 121	101 117	100 114	101 114
31500	107 124	104 120	101 116	102 116	103 116
32500	107 123	104 119	103 118	104 118	105 118
33500	107 123	105 120	106 120	107 120	108 120
34500	106 122	107 122	108 122	109 122	110 122
35500	108 123	109 123	110 123	111 123	112 123
36500	110 125	111 125	112 125	113 125	114 125
37500	112 127	113 127	114 127	115 127	116 127
38500	114 128	115 128	116 128	117 128	118 129
39500	116 130	117 130	118 130	119 130	120 130
40500	118 132	119 132	120 132	121 132	122 132
41500	120 133	121 133	122 133	123 133	124 133
42500	122 135	123 135	124 135	125 135	126 135
43500	124 136	125 136	126 136	127 136	128 136
44500	126 138	127 138	128 138	129 138	129 138
45500	128 139	129 139	129 139	130 139	131 139
46500	130 141	130 141	131 141	132 141	133 141
47500	131 142	132 142	133 142	134 142	135 142
48500	133 144	134 144	135 144	135 144	136 144

NOTE: For determining V_1, enter the appropriate takeoff analysis with the Static Air Temperature and wind and read V_1 for the Maximum Takeoff Weight. Use the lower between this V_1 and the V_R obtained from the above table as the V_1 for the actual Takeoff Weight.

TAKEOFF SPEEDS (Unbalanced Field Length)					
VR/V2 SPEEDS T/O-1 MODE - NORMAL V_2 - FLAPS 18°					
PRESSURE ALTITUDE (ft)	**STATIC AIR TEMPERATURE (°C)**				
SL →	-40 to 41	42 to 48	49 to 50	-	-
1000 →	-40 to 38	39 to 45	46 to 48	-	-
2000 →	-40 to 33	34 to 41	42 to 46	-	-
3000 →	-40 to 29	30 to 37	38 to 44	-	-
4000 →	-	-40 to 33	34 to 41	42 to 42	-
5000 →	-	-40 to 29	30 to 37	38 to 40	-
6000 →	-	-40 to 24	25 to 33	34 to 38	-
7000 →	-	-	-40 to 28	29 to 36	-
8000 →	-	-	-40 to 24	25 to 32	33 to 34
10000 →	-	-	-	-40 to 25	26 to 33
	↓	↓	↓	↓	↓
WEIGHT (lb)	**VR V2**	**VR V2**	**VR V2**	**VR V2**	**VR V2**
26500	109 123	104 118	100 112	99 110	98 108
27500	108 122	104 117	101 113	100 111	98 109
28500	108 122	104 116	102 114	100 112	99 109
29500	108 121	104 117	103 115	102 113	101 110
30500	108 120	106 118	105 116	104 114	103 111
31500	108 121	107 119	106 117	105 115	104 112
32500	109 121	108 119	107 117	106 116	106 114
33500	110 122	109 120	108 118	108 116	107 115
34500	111 123	110 121	109 119	109 117	108 116
35500	112 124	111 122	110 120	110 118	110 117
36500	113 124	112 122	112 121	111 119	111 118
37500	114 125	113 123	113 121	112 120	112 119
38500	115 126	114 124	114 122	113 121	114 120
39500	116 126	115 125	115 123	115 122	115 122
40500	117 127	116 125	116 124	116 123	117 123
41500	118 128	117 126	117 125	118 125	118 124
42500	119 128	118 127	118 126	119 126	120 126
43500	120 129	119 128	120 127	121 127	122 127
44500	121 130	120 128	121 128	122 128	123 128
45500	121 130	122 130	123 130	124 130	125 130
46500	122 131	123 131	124 131	125 131	126 131
47500	124 132	125 132	126 132	127 132	128 132
48500	125 133	126 133	127 133	128 133	130 134

NOTE: For determining V_1, enter the appropriate takeoff analysis with the Static Air Temperature and wind and read V_1 for the Maximum Takeoff Weight. Use the lower between this V_1 and the V_R obtained from the above table as the V_1 for the actual Takeoff Weight.

TAKEOFF SPEEDS (Unbalanced Field Length)				
VR/V2 SPEEDS T/O-1 MODE - NORMAL V_2 - FLAPS 22°				
PRESSURE ALTITUDE (ft)	**STATIC AIR TEMPERATURE (°C)**			
SL →	-40 to 46	47 to 50	-	-
1000 →	-40 to 43	44 to 48	-	-
2000 →	-40 to 39	40 to 46	-	-
3000 →	-40 to 35	36 to 42	43 to 44	-
4000 →	-40 to 31	32 to 38	39 to 42	-
5000 →	-40 to 26	27 to 34	35 to 40	-
6000 →	-40 to 22	23 to 30	31 to 38	-
7000 →	-40 to 18	19 to 25	26 to 33	34 to 36
8000 →	-	-40 to 20	21 to 29	30 to 34
10000 →	-	-	-40 to 24	25 to 33
	↓	↓	↓	↓
WEIGHT (lb)	**VR V2**	**VR V2**	**VR V2**	**VR V2**
26500	96 107	94 104	93 102	92 101
27500	96 107	96 105	95 103	94 102
28500	98 108	97 106	96 104	95 103
29500	99 108	98 107	97 105	96 103
30500	100 109	99 108	98 106	98 104
31500	101 110	101 109	100 107	99 105
32500	103 111	102 109	101 108	101 107
33500	104 112	103 110	102 109	103 109
34500	105 113	104 111	104 110	105 110
35500	106 114	105 112	106 112	106 112
36500	107 114	106 113	107 113	108 113
37500	108 115	108 115	109 115	109 115
38500	109 116	109 116	110 116	111 116
39500	110 118	111 118	112 118	112 118
40500	112 119	112 119	113 119	114 119
41500	113 120	114 120	115 120	115 120
42500	115 121	115 121	116 121	117 121
43500	116 122	117 122	117 122	118 122
44500	117 123	118 123	119 123	119 124
45500	119 125	119 125	120 125	121 125
46500	120 126	121 126	121 126	122 126
47500	121 127	122 127	123 127	123 127
48500	122 128	123 128	124 128	124 128

NOTE: For determining V_1, enter the appropriate takeoff analysis with the Static Air Temperature and wind and read V_1 for the Maximum Takeoff Weight. Use the lower between this V_1 and the V_R obtained from the above table as the V_1 for the actual Takeoff Weight.

APPROACH CLIMB SPEED (V_{APPCLB}), LANDING CLIMB & REFERENCE SPEEDS (V_{REF}) and FINAL SEGMENT SPEED (V_{FS})				
WEIGHT (lb)	**Approach Climb Speed (KIAS)**	**Landing Climb & Reference Speeds (KIAS)**		**V_{FS} (KIAS)**
	Flaps 9°	**Flaps 45°**	**Flaps 22°**	
27000	125	104	109	134
28000	128	106	111	136
29000	130	108	113	139
30000	132	109	114	141
31000	134	111	116	142
32000	136	113	118	145
33000	138	115	120	147
34000	140	117	121	150
35000	142	118	123	152
36000	144	120	125	154
37000	146	121	127	156
38000	148	123	128	158
39000	150	124	130	160
40000	152	126	131	162
41000	154	127	133	164
42000	156	128	134	165
43000	157	130	136	167
44000	159	131	137	169
45000	161	132	139	171
46000	163	134	140	172
47000	164	135	142	173
48000	166	136	143	175
49000	167	137	144	177
50000	167	137	144	179

APPROACH SPEED (V_{APP})
$V_{APP} = V_{REF}$ + ½ headwind + full gust

HOLDING - (All Engines)

CRUISE CONFIGURATION, BLEED OPEN
ANTI-ICE: OFF
MINIMUM FUEL CONSUMPTION SPEED, STANDARD ATMOSPHERE

WEIGHT (lb)			ALTITUDE (ft)								
			0	5000	10000	15000	20000	25000	30000	35000	37000
49000	IAS	kt	179	174	171	169	170	172	177	185	184
	N1	%	57.2	61.0	65.3	69.1	73.2	77.9	82.0	86.6	89.6
	FF	lb/h/Eng	1072	1028	997	979	974	974	984	1011	1029
48000	IAS	kt	178	173	170	168	168	170	174	182	184
	N1	%	56.7	60.4	64.7	68.6	72.6	77.4	81.5	85.8	88.9
	FF	lb/h/Eng	1054	1009	977	959	952	952	961	985	1003
46000	IAS	kt	175	170	167	165	164	166	170	177	180
	N1	%	55.6	59.3	63.5	67.7	71.4	76.2	80.3	84.5	87.4
	FF	lb/h/Eng	1018	972	939	918	909	908	914	935	952
44000	IAS	kt	173	168	164	161	161	162	165	171	175
	N1	%	54.5	58.1	62.3	66.6	70.2	74.9	79.2	83.3	85.7
	FF	lb/h/Eng	983	935	900	878	867	866	869	885	900
42000	IAS	kt	170	165	161	158	157	158	161	166	169
	N1	%	53.3	56.8	61.0	65.3	69.1	73.6	78.1	82.1	84.2
	FF	lb/h/Eng	948	898	862	838	825	823	824	837	850
40000	IAS	kt	168	162	158	155	153	154	156	161	164
	N1	%	52.2	55.6	59.6	64.0	67.9	72.2	76.9	81.1	82.9
	FF	lb/h/Eng	914	863	825	799	784	780	780	790	801
38000	IAS	kt	166	160	155	152	150	150	151	155	158
	N1	%	50.9	54.3	58.2	62.6	66.8	70.7	75.6	79.7	81.7
	FF	lb/h/Eng	880	827	788	761	744	738	737	744	753
36000	IAS	kt	163	157	152	148	146	145	147	150	152
	N1	%	49.7	53.0	56.8	61.1	65.5	69.3	74.1	78.3	80.5
	FF	lb/h/Eng	847	793	752	723	705	696	695	699	707
34000	IAS	kt	161	154	149	145	142	141	142	145	147
	N1	%	48.5	51.6	55.4	59.6	64.0	67.9	72.5	76.9	79.0
	FF	lb/h/Eng	814	759	716	686	666	655	654	655	661
32000	IAS	kt	158	152	146	142	139	137	138	140	141
	N1	%	47.2	50.2	53.8	57.9	62.4	66.4	70.8	75.5	77.4
	FF	lb/h/Eng	782	725	682	650	628	615	611	612	616
30000	IAS	kt	156	149	143	139	135	133	133	135	136
	N1	%	45.9	48.7	52.3	56.2	60.7	65.0	69.0	73.9	75.9
	FF	lb/h/Eng	751	693	647	614	590	576	570	569	573
28000	IAS	kt	154	146	140	135	132	129	128	129	130
	N1	%	44.6	47.3	50.6	54.5	58.8	63.3	67.3	72.1	74.2
	FF	lb/h/Eng	721	661	614	579	554	537	529	528	530
26000	IAS	kt	151	144	137	132	128	125	124	124	125
	N1	%	43.2	45.7	48.9	52.7	56.8	61.4	65.5	70.0	72.4
	FF	lb/h/Eng	692	630	581	544	518	500	490	487	489

HOLDING - (All Engines)

CRUISE CONFIGURATION, BLEED OPEN
AIRSPEED: 1.3 V_S OR 200 KIAS WHICHEVER IS HIGHER
ANTI-ICE: ON (NO ICE ACCRETION)
STANDARD ATMOSPHERE

WEIGHT (lb)			ALTITUDE (ft)								
			0	5000	10000	15000	20000	25000	30000	35000	37000
49000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	58.4	62.2	66.4	69.9	74.0	78.8	83.0	88.1	91.4
	FF	lb/h/Eng	1227	1171	1128	1099	1087	1083	1092	1120	1141
48000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	57.9	61.7	65.9	69.5	73.5	78.4	82.5	87.2	90.8
	FF	lb/h/Eng	1212	1155	1112	1082	1069	1065	1072	1098	1117
46000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	57.1	60.8	65.0	68.8	72.6	77.4	81.5	85.7	89.3
	FF	lb/h/Eng	1183	1126	1081	1050	1034	1030	1033	1055	1073
44000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	56.3	59.9	64.1	68.1	71.6	76.5	80.6	84.7	87.6
	FF	lb/h/Eng	1155	1097	1051	1019	1000	996	997	1014	1030
42000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	55.5	59.1	63.3	67.3	70.8	75.4	79.7	83.8	85.9
	FF	lb/h/Eng	1129	1070	1023	989	968	963	962	975	989
40000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	54.6	58.3	62.4	66.5	69.9	74.4	78.9	82.9	84.8
	FF	lb/h/Eng	1103	1043	996	960	937	930	929	939	951
38000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	53.8	57.4	61.5	65.6	69.1	73.3	78.2	82.1	84.0
	FF	lb/h/Eng	1079	1018	970	933	908	899	899	905	916
36000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	53.1	56.6	60.6	64.8	68.4	72.4	77.3	81.1	83.1
	FF	lb/h/Eng	1055	995	945	907	881	869	870	874	882
34000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	52.3	55.8	59.7	63.9	67.7	71.4	76.4	80.2	82.3
	FF	lb/h/Eng	1033	972	922	883	855	841	842	844	851
32000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	51.6	55.0	58.9	63.1	67.0	70.6	75.3	79.4	81.4
	FF	lb/h/Eng	1012	951	900	859	830	814	813	817	823
30000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	50.9	54.3	58.1	62.3	66.3	69.7	74.3	78.7	80.5
	FF	lb/h/Eng	993	930	879	838	807	789	785	791	796
28000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	50.2	53.6	57.4	61.5	65.5	68.9	73.3	78.0	79.8
	FF	lb/h/Eng	974	911	859	817	785	765	759	768	772
26000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	49.5	52.8	56.6	60.7	64.8	68.2	72.3	77.3	79.1
	FF	lb/h/Eng	957	894	841	798	765	743	735	742	750

HOLDING - (All Engines)

CRUISE CONFIGURATION, BLEED OPEN
AIRSPEED: 1.3 V_S OR 200 KIAS WHICHEVER IS HIGHER
ANTI-ICE ON (WITH ICE ACCRETION)
STANDARD ATMOSPHERE

WEIGHT (lb)			ALTITUDE (ft)								
			0	5000	10000	15000	20000	25000	30000	35000	37000
49000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	58.4	62.2	66.4	69.9	74.0	78.8	83.0	88.1	91.4
	FF	lb/h/Eng	1358	1286	1227	1184	1159	1145	1148	1174	1195
48000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	57.9	61.7	65.9	69.5	73.5	78.4	82.5	87.2	90.8
	FF	lb/h/Eng	1344	1271	1212	1168	1141	1127	1127	1150	1170
46000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	57.1	60.8	65.0	68.8	72.6	77.4	81.5	85.7	89.3
	FF	lb/h/Eng	1316	1241	1181	1135	1106	1092	1088	1106	1123
44000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	56.3	59.9	64.1	68.1	71.6	76.5	80.6	84.7	87.6
	FF	lb/h/Eng	1288	1213	1152	1104	1073	1058	1050	1063	1078
42000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	55.5	59.1	63.3	67.3	70.8	75.4	79.7	83.8	85.9
	FF	lb/h/Eng	1262	1186	1123	1074	1040	1025	1015	1023	1036
40000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	54.6	58.3	62.4	66.5	69.9	74.4	78.9	82.9	84.8
	FF	lb/h/Eng	1236	1160	1097	1046	1010	991	982	986	996
38000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	53.8	57.4	61.5	65.6	69.1	73.3	78.2	82.1	84.0
	FF	lb/h/Eng	1212	1135	1071	1019	981	960	951	951	960
36000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	53.1	56.6	60.6	64.8	68.4	72.4	77.3	81.1	83.1
	FF	lb/h/Eng	1189	1112	1047	994	954	930	922	918	925
34000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	52.3	55.8	59.7	63.9	67.7	71.4	76.4	80.2	82.3
	FF	lb/h/Eng	1168	1090	1024	970	928	902	893	888	893
32000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	51.6	55.0	58.9	63.1	67.0	70.6	75.3	79.4	81.4
	FF	lb/h/Eng	1147	1068	1002	947	904	875	864	860	864
30000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	50.9	54.3	58.1	62.3	66.3	69.7	74.3	78.7	80.5
	FF	lb/h/Eng	1128	1049	981	925	881	850	836	834	836
28000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	50.2	53.6	57.4	61.5	65.5	68.9	73.3	78.0	79.8
	FF	lb/h/Eng	1109	1030	962	905	860	827	809	810	811
26000	IAS	kt	200	200	200	200	200	200	200	200	200
	N1	%	49.5	52.8	56.6	60.7	64.8	68.2	72.3	77.3	79.1
	FF	lb/h/Eng	1092	1012	944	886	840	805	785	784	789

Copyright © by Embraer. Refer to cover page for details.

DRIFTDOWN TABLE

ANTI-ICE OFF

WEIGHT (lb)		INITIAL DRIFTDOWN SPEED (KIAS)	GROSS LEVEL OFF ALTITUDE – FT (NET LEVEL OFF ALTITUDE - FT)		
START DRIFTDOWN	LEVEL OFF		ISA + 10 & BELOW	ISA + 15	ISA + 20
46000	44200	172	20770 (16010)	20640 (15930)	20220 (15690)
44000	42300	169	22030 (17490)	21880 (17390)	21470 (17180)
42000	40500	165	23430 (19400)	23050 (19250)	22710 (18760)
40000	38700	161	24860 (21055)	24520 (20910)	23950 (20410)
38000	36900	157	26440 (22460)	26170 (22260)	25470 (21950)
36000	34900	154	28150 (24110)	27520 (23730)	26920 (23250)
34000	33000	149	29820 (25850)	29220 (25560)	28260 (24890)
32000	31100	145	31870 (27550)	30784 (27000)	29680 (26410)

ANTI-ICE ON

WEIGHT (lb)		INITIAL SPEED (KIAS)	GROSS AND (NET) LEVEL OFF ALTITUDE - FT				
START DRIFTDOWN	LEVEL OFF		ISA – 10 & BELOW	ISA - 5	ISA	ISA + 5	ISA + 10
46000	43600	173	17190 (13220)	17090 (13030)	15980 (12280)	14260 (10670)	12630 (8550)
44000	42000	169	18730 (14620)	18540 (14410)	17480 (13460)	15620 (11890)	13850 (10330)
42000	40100	165	20390 (16150)	20200 (15940)	19200 (14940)	17370 (13260)	15350 (11750)
40000	38200	161	21790 (17840)	21450 (17780)	20600 (16620)	18920 (14830)	16960 (13220)
38000	36300	158	23180 (19780)	22730 (19570)	21840 (18350)	20540 (16590)	18450 (14670)
36000	34400	154	24590 (21390)	24300 (21050)	23150 (20270)	21850 (18450)	20010 (16400)
34000	32600	149	26080 (22990)	25340 (22580)	24420 (21670)	23190 (20370)	21410 (18250)
32000	30700	145	27370 (24460)	26550 (23900)	25660 (23020)	24390 (21700)	22690 (19810)

Copyright © by Embraer. Refer to cover page for details.

UNFACTORED LANDING DISTANCE

UNFACTORED LANDING DISTANCE (FT) – DRY RUNWAY
EMB-145 - FLAPS 45°
ISA CONDITIONS - SLOPE 0%

WEIGHT (lb)	ALTITUDE							
	0 ft				1000 ft			
	WIND				WIND			
	-10 kt	0 kt	10 kt	20 kt	-10 kt	0 kt	10 kt	20 kt
44000	3298	2830	2683	2540	3371	2896	2747	2603
42000	3190	2732	2588	2449	3260	2795	2650	2509
40000	3085	2637	2497	2360	3149	2695	2552	2415
38000	2984	2546	2408	2275	3046	2601	2462	2326
36000	2883	2454	2320	2189	2942	2507	2371	2239
34000	2778	2359	2227	2100	2834	2410	2277	2147
32000	2668	2259	2131	2006	2721	2307	2178	2051
30000	2555	2157	2032	1911	2605	2202	2076	1953
28000	2441	2054	1932	1813	2488	2097	1974	1854

WEIGHT (lb)	ALTITUDE							
	2000 ft				3000 ft			
	WIND				WIND			
	-10 kt	0 kt	10 kt	20 kt	-10 kt	0 kt	10 kt	20 kt
44000	3448	2965	2814	2668	3527	3037	2884	2735
42000	3333	2861	2714	2570	3409	2930	2780	2635
40000	3218	2758	2614	2473	3291	2823	2677	2534
38000	3110	2659	2518	2381	3176	2720	2576	2437
36000	3002	2562	2424	2290	3065	2620	2479	2344
34000	2892	2462	2327	2196	2952	2516	2380	2247
32000	2776	2358	2226	2098	2834	2409	2276	2147
30000	2657	2250	2122	1998	2712	2299	2169	2043
28000	2537	2141	2017	1896	2588	2188	2062	1939

WEIGHT (lb)	ALTITUDE							
	4000 ft				5000 ft			
	WIND				WIND			
	-10 kt	0 kt	10 kt	20 kt	-10 kt	0 kt	10 kt	20 kt
44000	3611	3113	2957	2806	3696	3191	3032	2879
42000	3488	3002	2849	2702	3570	3076	2921	2771
40000	3367	2891	2743	2599	3445	2962	2812	2665
38000	3247	2783	2638	2497	3319	2849	2701	2558
36000	3132	2679	2537	2400	3200	2741	2597	2458
34000	3015	2573	2436	2301	3080	2633	2493	2356
32000	2894	2464	2329	2197	2956	2519	2383	2250
30000	2768	2350	2220	2092	2827	2404	2271	2142
28000	2642	2236	2109	1984	2698	2286	2157	2032

NOTE: Landing distance in feet.

Copyright © by Embraer. Refer to cover page for details.

UNFACTORED LANDING DISTANCE

UNFACTORED LANDING DISTANCE (FT) – DRY RUNWAY
EMB-145 - FLAPS 22°
ISA CONDITIONS - SLOPE 0%

WEIGHT (lb)	ALTITUDE							
	0 ft				1000 ft			
	WIND				WIND			
	-10 kt	0 kt	10 kt	20 kt	-10 kt	0 kt	10 kt	20 kt
44000	4281	3709	3529	3353	4381	3800	3617	3438
42000	4123	3566	3390	3218	4218	3652	3473	3298
40000	3964	3421	3249	3081	4053	3502	3327	3158
38000	3810	3281	3114	2951	3895	3358	3188	3022
36000	3660	3145	2982	2823	3740	3217	3052	2890
34000	3507	3005	2846	2692	3581	3073	2912	2755
32000	3356	2869	2714	2563	3427	2932	2776	2623
30000	3212	2737	2587	2440	3278	2798	2645	2497
28000	3067	2606	2459	2317	3129	2662	2514	2369

WEIGHT (lb)	ALTITUDE							
	2000 ft				3000 ft			
	WIND				WIND			
	-10 kt	0 kt	10 kt	20 kt	-10 kt	0 kt	10 kt	20 kt
44000	4485	3895	3708	3527	4594	3994	3804	3620
42000	4316	3741	3559	3382	4419	3835	3650	3470
40000	4145	3586	3409	3236	4242	3674	3494	3319
38000	3982	3437	3265	3097	4073	3520	3346	3175
36000	3822	3292	3124	2961	3908	3370	3200	3034
34000	3659	3143	2980	2821	3740	3217	3052	2891
32000	3499	2998	2840	2685	3576	3068	2907	2750
30000	3346	2859	2705	2555	3418	2925	2768	2616
28000	3193	2720	2570	2424	3260	2781	2629	2481

WEIGHT (lb)	ALTITUDE							
	4000 ft				5000 ft			
	WIND				WIND			
	-10 kt	0 kt	10 kt	20 kt	-10 kt	0 kt	10 kt	20 kt
44000	4707	4097	3904	3716	4827	4205	4009	3818
42000	4526	3932	3744	3562	4639	4034	3844	3658
40000	4343	3766	3583	3406	4450	3863	3677	3496
38000	4169	3607	3429	3256	4269	3698	3518	3342
36000	3998	3452	3279	3111	4092	3538	3362	3191
34000	3824	3294	3126	2962	3913	3374	3204	3038
32000	3655	3140	2976	2818	3738	3215	3050	2889
30000	3492	2992	2834	2679	3570	3063	2902	2746
28000	3330	2844	2690	2540	3402	2910	2754	2602

NOTE: Landing distance in feet.

Copyright © by Embraer. Refer to cover page for details.

ADVISORY INFORMATION
EMB-145 UNFACTORED LANDING DISTANCES - CONTAMINATED RUNWAYS (ft)
ALL ENGINES TYPES – FAA CERTIFICATION
STANDING WATER 0.125 in/SLUSH 0.15 in
WET SNOW 0.25 in/DRY SNOW 0.625 in

WEIGHT (lb)	FLAP 22°	FLAP 45°
29000	6033	4822
30000	6194	4946
31000	6357	5071
32000	6532	5201
33000	6707	5330
34000	6883	5459
35000	7058	5589
36000	7245	5721
37000	7436	5852
38000	7627	5983
39000	7819	6114
40000	8010	6246
41000	8197	6377
42000	8383	6508
43000	8570	6639
44000	8757	6768
45000	8971	6914
46000	9188	7062
47000	9404	7210
48000	9621	7357

CORRECTIONS	
ALTITUDE:	LANDING DISTANCE + 3% per 1000 ft above sea level.
WIND:	LANDING DISTANCE + 11% per 5 kt tailwind.
OVERSPEED:	LANDING DISTANCE + 9% per 5 kt above V_{REF}.

Copyright © by Embraer. Refer to cover page for details.

ADVISORY INFORMATION
EMB-145 UNFACTORED LANDING DISTANCES - CONTAMINATED RUNWAYS (ft)
ALL ENGINES TYPES – FAA CERTIFICATION
STANDING WATER 0.25 in/SLUSH 0.29 in
WET SNOW 0.50 in/DRY SNOW 1.25 in

WEIGHT (lb)	FLAP 22°	FLAP 45°
29000	5276	4357
30000	5411	4464
31000	5546	4571
32000	5693	4683
33000	5839	4794
34000	5986	4905
35000	6132	5016
36000	6289	5129
37000	6448	5243
38000	6608	5356
39000	6768	5470
40000	6928	5583
41000	7086	5697
42000	7243	5810
43000	7401	5923
44000	7559	6036
45000	7739	6163
46000	7922	6292
47000	8105	6421
48000	8288	6549

CORRECTIONS	
ALTITUDE:	LANDING DISTANCE + 3% per 1000 ft above sea level.
WIND:	LANDING DISTANCE + 11% per 5 kt tailwind.
OVERSPEED:	LANDING DISTANCE + 8% per 5 kt above V_{REF}.

Copyright © by Embraer. Refer to cover page for details.

ADVISORY INFORMATION
EMB-145 UNFACTORED LANDING DISTANCES - CONTAMINATED RUNWAYS (ft)
ALL ENGINES TYPES – FAA CERTIFICATION
STANDING WATER 0.50 in/SLUSH 0.59 in
WET SNOW 1.00 in/DRY SNOW 2.50 in

WEIGHT (lb)	FLAP 22°	FLAP 45°
29000	4411	3856
30000	4515	3943
31000	4619	4031
32000	4732	4122
33000	4845	4213
34000	4957	4304
35000	5070	4395
36000	5190	4488
37000	5313	4581
38000	5436	4674
39000	5559	4766
40000	5682	4859
41000	5804	4952
42000	5926	5044
43000	6048	5137
44000	6170	5229
45000	6310	5332
46000	6451	5437
47000	6593	5542
48000	6734	5647

CORRECTIONS	
ALTITUDE:	LANDING DISTANCE + 3% per 1000 ft above sea level.
WIND:	LANDING DISTANCE + 10% per 5 kt tailwind.
OVERSPEED:	LANDING DISTANCE + 8% per 5 kt above V_{REF}.

Copyright © by Embraer. Refer to cover page for details.

ADVISORY INFORMATION
EMB-145 UNFACTORED LANDING DISTANCES - CONTAMINATED RUNWAYS (ft)
ALL ENGINES TYPES – FAA CERTIFICATION
COMPACTED SNOW

WEIGHT (lb)	FLAP 22°	FLAP 45°
29000	3944	3645
30000	4036	3734
31000	4129	3823
32000	4222	3909
33000	4315	3996
34000	4408	4082
35000	4501	4169
36000	4593	4250
37000	4685	4329
38000	4777	4409
39000	4868	4488
40000	4960	4567
41000	5052	4646
42000	5143	4726
43000	5235	4805
44000	5327	4884
45000	5417	4960
46000	5507	5035
47000	5597	5111
48000	5688	5186

CORRECTIONS	
ALTITUDE:	LANDING DISTANCE + 3% per 1000 ft above sea level.
WIND:	LANDING DISTANCE + 11% per 5 kt tailwind.
OVERSPEED:	LANDING DISTANCE + 7% per 5 kt above V_{REF}.

Copyright © by Embraer. Refer to cover page for details.

ADVISORY INFORMATION
EMB-145 UNFACTORED LANDING DISTANCES - CONTAMINATED RUNWAYS (ft)
ALL ENGINES TYPES – FAA CERTIFICATION
ICE

WEIGHT (lb)	FLAP 22°	FLAP 45°
29000	11551	9411
30000	11556	9406
31000	11572	9412
32000	11656	9487
33000	11740	9561
34000	11824	9635
35000	11909	9709
36000	12032	9810
37000	12170	9921
38000	12308	10032
39000	12446	10143
40000	12584	10254
41000	12721	10365
42000	12859	10476
43000	12997	10587
44000	13135	10698
45000	13303	10826
46000	13473	10955
47000	13644	11084
48000	13814	11213

CORRECTIONS	
ALTITUDE:	LANDING DISTANCE + 3% per 1000 ft above sea level.
WIND:	LANDING DISTANCE + 24% per 5 kt tailwind.
OVERSPEED:	LANDING DISTANCE + 5% per 5 kt above V_{REF}.

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

Copyright © by Embraer. Refer to cover page for details.

INTENTIONALLY BLANK

EMERGENCY EVACUATION

Parking Brake.....APPLY
Cabin.....DEPRESSURIZE
Fire Extinguishing Handles.....PULL
APU Fuel Shutoff Valve .....PUSH IN
Engines and APU Fire Extinguishing Bottles (if necessary) .....DISCHARGE
Ventral Tank Transfer Knob (if applicable).....OFF
Fuel Pumps Pwr 1 and 2.....OFF
Hydraulic Elec Pumps 1 and 2 .....OFF
Cabin Crew.....NOTIFY
Emerg Lts .....ON
Evacuation.....INITIATE
ATC .....NOTIFY
Before leaving the airplane:
Batteries 1 and 2.....OFF

END